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6	INTERACTIONAL AERODYNAMICS OF THE SINGLE ROTOR HELICOPTER CONFIGURATION.
1	VOLUME IV-F. One-Third Octave Band Spectrograms of Wake Split-Film Data, Air Ejectors With Hubcaps; Wings.
4	Philip F./Sheridan
4	Boeing Vertol Company
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# APPLIED TECHNOLOGY LABORATORY POSITION STATEMENT

In 1975 a wind tunnel test program was conducted in the Boeing-Vertol 20-foot V/STOL Wind Tunnel on a 1/5th-scale UTTAS model to investigate and find solutions for several aerodynamic problems encountered during the UTTAS flight-testing. Specifically, these tests focused upon (a) the structure of the hub/rotor wake in the vicinity of the empennage, (b) the formulation of the ground vortex and its relation to hub loads and fuselage loads during transition, and (c) the occurrence of vibratory air pressures from the blade passing over the fuselage. Only portions of the above-mentioned wind tunnel test data were reduced and analyzed in addressing the flight-test problems of the UTTAS aircraft.

Under Contract DAAJ02-77-C-0020, Boeing-Vertol completed analyses on the data to understand more completely the aerodynamic interactions that are involved and to formulate instructions for the guidance of designers in these respects. The results of these studies are applicable to all existing and future single-rotor/tail rotor helicopters. The data have been segregated according to aerodynamic interactions and associated phenomena/problem areas. From this body of knowledge, a generalized set of design guidelines meaningful to the single-rotor helicopter design concept formulation were developed and are included in these reports.

Mr. Robert P. Smith of the Aeronautical Technology Division, Aeromechanics Technical Area, served as project engineer for this effort.

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This is the sixth of the seven sub-volumes of Volume IV containing one-third octave band spectrographs of the model helicopter hub/rotor wake as it was modified by various aerodynamic devices. This sub-volume deals with the effects of air ejector systems in configurations already possessing hub caps and also of several wing configurations mounted variously to alter the wake.

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# **PREFACE**

The entire report describing the investigation of INTERACTIONAL AERODYNAMICS OF THE SINGLE-ROTOR HELICOPTER CONFIGURATION comprises eight numbered volumes bound as 33 separate documents. The complete list of these documents is as follows:

# Volume I, Final Report

Volume II, Harmonic Analyses of Airframe Surface Pressure Data

- A Runs 7-14, Forward Section
- B Runs 7-14, Mid Section
- C Runs 7-14, Aft Section
- D Runs 15-22, Forward Section
- E Runs 15-22, Mid Section
- F Runs 15-22, Aft Section
- G Runs 23-33, Forward Section
- H Runs 23-33, Mid Section
- I Runs 23-33, Aft Section

Volume III, Flow Angle and Velocity Wake Profiles in Low-Frequency Band

- A Basic Investigations and Hubcap Variations
- B Air Ejector Systems and Other Devices

Volume IV, One-Third Octave Band Spectrograms of Wake Split-Film Data

- A Buildup to Baseline
- B Basic Configuration Wake Explorations
- C Solid Hubcaps
- D Open Hubcaps
- E Air Ejectors
- F Air Ejectors With Hubcaps; Wings
- G Fairings and Surface Devices

Volume V, Harmonic Analyses of Hub Wake

Volume VI, One-Third Octave Band Spectrograms of Wake Single Film Data

- A Buildup to Baseline
- B Basic Configuration Wake Exploration
- C Hubcaps and Air Ejectors

Volume VII, Frequency Analyses of Wake Split-Film Data

- A Buildup to Baseline
- B Basic Configuration Wake Explorations
- C Solid Hubcaps



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D - Open Hubcaps

E - Air Ejectors

F - Air Ejectors With Hubcaps; Wings

G - Fairings and Surface Devices

Volume VIII, Frequency Analyses of Wake Single Film Data

A - Buildup to Baseline

B - Basic Configuration Wake Exploration

C - Hubcaps and Air Ejectors

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# INTRODUCTION

Volume IV presents spectrograms of the flow angles and velocity components for each run and its test points. Specifically, these machine plots show the root mean square value of each wake parameter over discrete frequency intervals one-third of an octave band in width. The octave arrangement is selected to provide 19 spectral increments from 3.9 to 250.0 Hz centerband frequency. A special computer program is employed to derive wake parameters within these bands consistent with corresponding basic spectral functions depicted in Volume VII.

The graphs showing the one-third octave band values are sequenced in the same order as the Outline of Wake Investigations (Table 1). These graphs are distributed among Volumes IV-A through IV-G by the major categories of Table I in the following arrangement:

Volume	IV-A	Build-up to Baseline
Volume	IV-B	Basic Configuration
Volume	IV-C	Effect of Hub Caps Section 1 & 2
Volume	IV-D	Effect of Hub Caps Section 3 & 4
Volume	IV-E	Effect of Hub Caps Section 5 and
		Effect of Air Ejectors
Volume	IV-F	Air Ejectors with Open Hub Caps and
		Effect of Wings and Misc. Section
Volume	IV-G	Effect of Wings and Misc. Sections
		2 and 3

The Table I outline-and other material is included for reference and as a context to the work of each sub-volume. Table 2, the List of Test Runs, arranges the runs in numerical order and gives pertinent text parameters.

The Index of Rake Positions, Table 3, lists the hot film transducer rake positions in the model coordinate system for each run and its test points. The main feature of Table 3 is the indexing of the test point number to the model waterline station and butt line as it varied from run to run. The table groups the runs as they shared the indexing correspondence of point with position. It is emphasized that the runs in a group do not necessarily all share the same number of test points but they do have same correspondence within their respective ranges of test points.

The orientation of the rake is shown pictorially in Figures 1 through 6 for the various test runs. Figure 7 presents a scaled drawing of the model with reference to the three-axis coordinate system. Table 4 lists the center frequency and the upper and lower band limits for each of the numbered one-third octave bands.

	TABLE 1			
	OUTLINE OF WAKE IN	VESTIGATIONS		
- 64-64 - 64-64	Description	Configuration Code		Base- line
Build-	up to Baseline			gr de
1. Na	celles removed	K <sub>13</sub> +H <sub>1</sub> -N	149	150
2. Bl	ades off, rotating hub	K <sub>13</sub> -M+H <sub>1.0</sub>	160	156
3.	" , non-rotating hub	K <sub>13</sub> -M+H <sub>1.0</sub>	158	156
4.	" , hub off	K <sub>13</sub> -M-H <sub>1.0</sub>	159	156
Basic	Configuration			
1. Wake	Explorations near Empennage			
(a)	15" Long. + traverse at T/R C.L.	K <sub>11</sub>	1111	
	9" Vert. + " above T/R "		112	
(c)		•	113	
(a)	8" " (continue 112)	•	1114	
(e)	13" " behind stab.	•	115	
(f)	Lateral traverse, left stab.	Barrell Track Control	116	
	(One T.P. only)		1227	
	Same continued		117	
	Same continued (One T.P. only)		1118	
	Lateral traverse right stab.  T/R effect on wake	K <sub>11</sub> +T <sub>2</sub> <sup>0</sup>	121	115
2. Clim	b/Descent Studies			
(=)	Climb 900 FPM	K1 1	135	
	Descent 800 FPM	"	136	
Effect	Of Hub Caps			
1.	Solid Caps on Canister			
	(a) 7.6" diam. 2.17" ht. soft Pitch Arms	K <sub>11</sub> -H <sub>1.0</sub> +H <sub>1.2</sub>	137	136
	(b) 7.6" diam. 2.17" ht. stiff	K <sub>13</sub> +H <sub>1.2</sub>	153	156
	Pitch Arms (b) 7.6" diam. 2.45" ht. flt.	K <sub>13</sub> +H <sub>1-2.1</sub> +I <sub>r</sub>	207	188
	test config.	+E1.0	STATE OF THE STATE	10 年 10 10 10 10 10 10 10 10 10 10 10 10 10

				TABL	E 1 (CONT	INGED)			
		(	OUTLI	NE OF	WAKE INV	ESTIGATIONS	rateo II.		
311	1	esci	ripti	on		Configura Code		Run No.	Base- line
ffect	of Hu	b Car	os (Co	ntinue	<u>d</u> )		ulio esti.		OTHER.
2.	Soli	d Car	s Rai	sed Ab	ove Caniste	r	er care		and it
					ht. 70"	H <sub>1,2,2</sub> +I <sub>1</sub>	+E1.0	208	188
	(b)	10.0	1, .55 diam 1, .50	. 3.25	" ht. 1.55"	H <sub>1.8.1</sub> +I <sub>1</sub>	+E1_0	L89	188
		10.0	diam	. 4.12	5" ht. 5" gap	H <sub>1.8.2</sub> +I <sub>1</sub>	+E1.0	L90	188
	(d)		t of		. 1.5		"	210	188
3.	Open	Caps	With	out Un	derbody				
	(a) (b)	10.0	diam	. 1.25	gap,blade gap, no blades	H <sub>1.0.1</sub> +I <sub>1</sub>	+E1.0	L93 L66	188/16 158
	(c) (d)	"		2.05	gap, blade gap, no	H1.14.1+I	1+E1.0	211	188 158
	(e) (f) (g)	16" ć	liam.	2.00"	blades gap,blades gap,blades gap, no		]	191 168 167	188 156/16 158
	(h)		•		blade gap,blades	11 7 1		169	156
4.	Open	Caps	with	Under	body	9.50	2 487 33-1		2 7
	(a) (b) (c)	7.6"	diam.	1.25"	gap " center post	H <sub>1.11.1</sub> +I <sub>2</sub> H <sub>1.11.1</sub> +I <sub>3</sub> H <sub>1.11.2</sub> +I <sub>3</sub>	+E4 0 1	194	188 188 194
	(d)	10.0	diam	5"	gap, no blades	H1.5.1-M		164	158
	(e)	"	. "	1.25"	gap, no blades	H <sub>1.5.2</sub> -M	1	.61	158
	(f)		•	2.0"	gap, no blades	H1.5.4-M	1	.63	158
	(g)	•	•	4.0"	gap, no blades	H1.5.3-M	antra ri	.62	158
	(h)			1.25"	dab	H1.5.2		154	156/16

### TABLE 1 (CONTINUED) OUTLINE OF WAKE INVESTIGATIONS Configuration Run Base-Description Code\* No. line Miscellaneous Hub Covers (a) Hub fairing 16" diam. H1.3 151 150 (b) Wham-O-Frisbee 10" diam. H1.9.0+E1.2 182 181 (c) Fab. glass Frisbee 16" diam. H1.9.1+E1.2 183 181 Effect of Air Ejectors Basic system no blowing H1.0+E1.0 172 156 2. 40 psi 173 156/172 3. 150 psi 174 156/172 4. Wide chord shroud 40 psi H1.0+E2.5.1 175 156/173 " 150 psi 176 156/174 HI-0+E3-5-2 H1-0+E3-5-4 H1-0+E3-5-4 6. W/C shroud w. lip 40 psi 184 156/173 7. Same Contoured Parallel 150 psi 187 156/174 Bifurcated duct 0 psi 203 156 8. 40 psi 204 156/203 10. 205 156/203 150 psi Air Ejectors with Open Hub Caps with Underbodies 1. 7.6" diam. 1.25" gap, 0 psi H<sub>1.11.1</sub>+1<sub>2</sub>+E<sub>1.0</sub> 194 188/172 2. 20 psi 195 188 \*\* 10 196 188/173 3. 40 psi \* 4. 150 psi 197 188/174 5. H1.11.1+I2+E4.0 198 188/194 0 psi 40 psi 150 psi 199 188/196 6. 188/196 200 8. Same with center post H1.11.2+I2+E4.6 201 188/200 9. 10.0" diam. 2.0" gap wide ch'd H1.5.4+E2.5.1 177 156/176 shroud (150 psi) Effect of Wings and Misc. Wings 178 ISI (a) Nacelle-mounted stub wing H1.0+W1.0+E1.1 H1.0+W3.0+E1.0 180 181 (b) Single slotted flapped wing 179 181 H1.0+W2.0+E1.0 (c) Dougle slotted flapped wing 186 (d) Boom-mounted stub wing H1.0+W4.0 156 \*Basic Code is Kl3.

TABLE 1 (CONTINUED)

# OUTLINE OF WAKE INVESTIGATIONS

	Description	Configuration Code*	Run No.	Base- line
2.	Crown Fairings  (a) Flat top behind shaft  (b) Round top behind shaft  (c) Extended flat top fairing  (d) Flat top + 16" cap, 4" gap  (e) Forward fairing/nacelle fairing	K <sub>11</sub> +D <sub>1</sub> K <sub>11</sub> +D <sub>2</sub> H <sub>1</sub> +D <sub>4</sub> H <sub>1·7·2</sub> +D <sub>4</sub> P <sub>1·0</sub>	-140 141 170 171 152	138 138 156 170 156
3.	Surface Devices (a) Vortex generators (b) Guidevane between nacelles (c) Longitudinal strakes (d) 14% porosity spoiler	K <sub>11</sub> +VG <sub>2.1</sub> K <sub>11</sub> +FV <sub>1</sub> H <sub>1·5.3</sub> +S <sub>4</sub> K <sub>11</sub> +X <sub>1</sub>	139 142 155 143	138 138 156 138
		30.34		
		200 (200 (200 (200 (200 (200 (200 (200		
				-6
		State that "T. S. a Cran that "T. S. a Cran C		
		athe care contraction organization objects as an important description became caste against	Garage Garage Access	

\*Basic Code is Kl3 unless noted otherwise.

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		HIID
	RUNS	THE
2	TEST	NG OF
TABLE 2	OF	ATTC
I	LIST OF TEST	BASIC INVESTIGATIONS OF THE HIR MAKE
		BASTC

					,	-				,				
	TAIL		Off	=	=	=	=	=	=	=	=	on	Off	=
MR	1 2	n/d	8	=	=	-	=	=	-	=	=	=	=	=
MODEL	9	) <del> </del>	-2.0	=	=	=	z	=	=	=	=	=	-4.5	-2.0
MODEL	•	8	0.9	=	=	=	=	=	=	=	=	=	0.9-	6.0 -2.0
DICK	LDG.	pst	8	=	=	=	=	=	=	=	=	=	=	=
	RPM MR/TR		1433/0	=	=		=	=	=		=	1433/	=	=
	KNOTS		80	=	=	=	=	=	=		=	=	=	
	CONFIGURATION/CONDITION		$_{11}/15$ " Long. wake traverse at TR center line	" /9" Vert. wake traverse above TR center line	" /2" Vert traverse through MR vortex	" /8" Vert. traverse below TR center line	" /13" Vert. traverse behind stabilizer	" /Lateral traverse - left stabilizer	" /116 continued	" /116 continued	" /Lateral traverse - right stabilizer	K <sub>11</sub> +T <sub>2</sub> /Effect of tail rotor flow on wake	K <sub>11</sub> /Wake in 900 fpm climb	" /Wake in 800 fpm descent
MIG	NO.		111	112	113	114	115	116	117	118	119	121	135	136

# TABLE 2(CONTINUED)

		~												-
	TAIL	ROTOR	OÉÉ		•		•		•		•	=		
	MR HT.	p/q	8		=			=				=		
	MODEL	•	-3.8		=			=	•	•				=
	MODEL	o g	9					:				<b>E</b>		
CES	DISK	pst.	8	=	=			=		4.5			8	=
IS DEVI	RPM	MR/TR	1433/0		=		•	=		1075/0			1433/0	
OF TEST RUNS WAKE-ALTERIN	VTUN	KNOTS	80	•	=					09	•		80	
LIST OF TEST RUNS EVALUATION OF WAKE-ALTERING DEVICES	CONFIGURATION/CONDITION		Kll-Hl.0+Hl.2/Effect of 7.6 inch diam. solid hub cap	K <sub>11</sub> /Repeat of base run	Kll+VG2.1/Effect of vortex gener- ators on aft crown	K <sub>11</sub> +D <sub>1</sub> /Flat-topped "doghouse" fair- ing on aft crown	K <sub>11</sub> +D <sub>2</sub> /Rounded-top fairing	K <sub>11</sub> +FV <sub>1</sub> /Deflection vane on crown between nacelles	K <sub>11</sub> +X <sub>1</sub> /Variable porosity spoiler	K <sub>13</sub> +H <sub>1</sub> -N <sub>1</sub> /Effect of nacelles off also add stiff pitch arms (K <sub>13</sub> )	K <sub>13</sub> +H <sub>1</sub> /60 knot baseline	K <sub>13</sub> +H <sub>1.3</sub> /16 inch diam. helmet fair- ing	K <sub>13</sub> +P <sub>1.0</sub> /Pylon and intake fairings	$ m K_{13}^{+H}_{1.2}/Repeat$ 137 with $ m K_{13}$ pitch arms
	RUN	NO.	137	138	139	140	141	142	143	149	150	151	152	153

8	TABLE 2 (CONTINUED) LIST OF TEST RUNS	CONTINUED TEST RUP	18					
	EVALUATION OF WAKE-ALTERING DEVICES	-ALTERI	NG DEVIC	ES				
RUN	CONSTGHBATTON/CONDITITON	Vrun	RPM	DISK	MODEL	MODEL	MR HT.	TAIL
NO.	NOTITION (NOTITION TO A STATE OF THE STATE O	KNOTS	MR/TR	LDG.	e g	•	p/q	ROTOR
154	K13+H1.5.2/10" open hub cap, 7" underbody, 1.25"gap	80	1433/0	8	9	-3.8	8	Off
155	K <sub>13</sub> +H <sub>1.5.2</sub> +S <sub>4</sub> /Same as 154 except strakes on aft crown		=	•		=		=
156	K <sub>13</sub> +H <sub>1.0</sub> /Baseline with K <sub>13</sub> ,i.e., stiff pitch arms			•	=		=	
158	K <sub>13</sub> -M+H <sub>1.0</sub> /Wake studies with blades off, hub not rotating		0/0	•	=	•	=	
159	K <sub>13</sub> -M-H <sub>1.0</sub> /Wake studies with hub off				=	=	=	
160	K <sub>13</sub> -M+H <sub>1.0</sub> /Same as 158 except hub is rotating		1433/0	•	=	-	=	-
191	K <sub>13</sub> -M+H <sub>1.5.2</sub> /Repeat of 154 without blades		0/0	-	•	=	=	
162	K <sub>13</sub> -M+H <sub>1.5.3</sub> /Same as 161 except 4"		•		•	•		
163	K <sub>13</sub> -M+H <sub>1.5.4</sub> /Same as 161 except 2"							
164	K <sub>13</sub> -M+H <sub>1.5.1</sub> /Same as 161 except 0.5" gap					:		
165	K <sub>13</sub> -M+H <sub>1.0.1</sub> /10" open hub cap,no underbody,same cap vert.position as Run 154					=		=
166	K <sub>13</sub> -M+H <sub>1.0.2</sub> /Same as 165 with cap lowered by 0.5"						=	=

# TABLE 2 (CONTINUED) LIST OF TEST RUNS EVALUATION OF WAKE-ALTERING DEVICES

11	OR	.41											
TAIL	ROTOR	Off	•	•	=	•	2	•	•	•		•	=
MR HT.	p/q	8		=		=		:		=		=	:
MODEL	•	-3.8		=	=	2	•	•	•			=	
MODEL	° 5	9		= \		=		=				=	
DISK	rbg.	8	=	a	=		=		u	•			
RPM	MR/TR	0/0	1433/0			=	=				•		
VTUN	KNOTS	80	<b>e</b> -		•								
CONFIGURATION/CONDITION		K <sub>13</sub> -M+H <sub>1.7.1</sub> /16" open cap, no underbody, 2" gap	Kl3+Hl.7.1/Blades on, same cap config. as 167	Kl3+Hl.7.2/16" open cap, no under- body, 4" gap	K <sub>13</sub> +H <sub>1.0</sub> +D <sub>4.0</sub> /Extended flat top fairing on aft crown	K <sub>13</sub> +H <sub>1,7,2</sub> +D <sub>4,0</sub> /Same fairing as 170, same cap as 169	Kl3+H1.0+E1.0(Opsi)/Basic air ejector zero blowing baseline	K <sub>13</sub> +H <sub>1.0</sub> +E <sub>1.0</sub> (40 psi)/Same as 172 with 40 psi supply	K <sub>13</sub> +H <sub>1.0</sub> +E <sub>1.0</sub> (150 psi)/Same as 172 with 150 psi supply	Kl3+H1 0+E2,5,1 (40 psi)/Ejector with wide chord shroud at 40 psi	K <sub>13</sub> +H <sub>1.0</sub> +E <sub>2.5.1</sub> (150 psi)/Same as 174 with 150 psi supply	K <sub>13</sub> +H <sub>1</sub> .5 <sub>1</sub> 4+E <sub>2</sub> 5,1(150 psi)/Same as	K13+H1.0+W1.0
RUN	NO.	167	168	169	170	171	172	173	174	175	176	177	178

	TABLE 2(C	TABLE 2(CONTINUED)						
	LIST OF TEST RUNS	TEST RUI	SN					
	EVALUATION OF WAKE-ALTERING DEVICES	ALTERIN	3 DEVICE	60				
RUN	CONFIGURATION	Vrun	RPM	DISK	MODEL	MODEL	MR HT.	TAII.
	NOTTINGO (NOTTINGO TINGO	KNOTS	MR/TR	LDG.	° g	0	h/d	ROTOR
179	K13+H1.0+W2.0+E1.0(0 psi)/Double slotted flapped wing	80	1433/0	8	9	-3.8	8	Off
180	K <sub>13</sub> +H <sub>1.0</sub> +W <sub>3.0</sub> +E <sub>1.0</sub> (0 psi)/Single slotted flapped wing	=	2	•		=	=	=
181	Kl3+H1.0+E1.2 (0 psi)/Baseline with ejector tube moved aft	=	=		=	=		-
182	K <sub>13</sub> +H <sub>1.9.0</sub> +E <sub>1.2</sub> (0 psi)/Standard 10" frishee	=			=	2	-	
183	K <sub>13</sub> +H <sub>1.9.1</sub> +E <sub>1.2</sub> (0 psi)/16" fabri- cated frisbee	=			=		=	
184	Kl3+H1.0+E3.5.2 (40 psi)/Wide chord with lip at 40 psi	=						
185	K <sub>13</sub> +H <sub>1.0</sub> +E <sub>3.5.2</sub> (150 psi)/Same as 184 with 150 psi air	=		=	=	=	•	
186	K13+H1.0+W4.0/Boom mounted stub wing			•	=	=	=	-
187	Kl3+Hl.0+E3.5.4 (150 psi)/Like 185 with modified shroud		29°0-16					-
188	Kl3+H1.0+I1+E1.0(0 psi)/Baseline with I, instr. ring	•			=	=	=	
189	0 psi)/Somm. 3.25" }	•			=	=	=	=
190	K <sub>13</sub> +H <sub>1.8.2</sub> +I <sub>1</sub> +E <sub>1.0</sub> (0 psi)/Same as 190 except + 4.12" height	=	=				=	=

	TABLE 2(CONTINUED)  LIST OF TEST RUNS  EVALUATION OF WAKE-ALTERING DEVICES	TEST RUI	NS G DEVICE	v				
RON		Verin	Mag		NA NA	MODEL	HT.	
NO.	CONFIGURATION/CONDITION	KNOTS	MR/TR	LDG.		-	h/d	ROTOR
191	Kl3+H1.0.2+I1+E1.0 (0 psi)/10" cap, no underbody, 1.87" gap	80	1433/0	8	9	-3.8	8	Off
193	E1.0 (0 I		•	•	•			
194	Kl3+H1.11.1+I2+E1.0(0 psi)/7.6" cap, underbody, 1.25" gap	•			•	•	•	
195	K13+H1.11.1+I2+E1.0(20 psi)/Same as 194 with 20 psi air		=	•				
196	K13+H1.11.1+I2+E1.0(40 psi)/Same as 194 with 40 psi air		=	•	•	•	=	
197	K13+H1.11.1+I2+E1.0(150 psi)/Same as 194 with 150 psi air	•	•			•		
198	Kl3 <sup>+H</sup> 1.11.1 <sup>+I</sup> 2 <sup>+E</sup> 4.0 (0 psi)/Same as 194 except blowing tube 2" aft		=	=	•			
199	K13+H1.11.1+I2+E4.0 (40 psi)/Same as 198 with 40 psi air				•			
200	Kl3 <sup>+H</sup> 1,11,1 <sup>+</sup> I2 <sup>+</sup> E4,0 (150 psi)/Same as 198 with 150 psi air	•		=	•	:		
201	Kl3+H1.11.2+I2+E4.0 (150 psi)/Same as 200 except center support cap	•	=		•			
202	K <sub>13</sub> +H <sub>1,11,2</sub> +I <sub>2</sub> /Baseline with I <sub>2</sub> and no blowing tube			=	•	:		
203	K <sub>13</sub> +H <sub>1.0</sub> +E <sub>5.0</sub> (0 psi)/Bifurcated air duct baseline				=	=		

	TAIL	ROTOR	off		•		-					
10000	HT.	p/q	8		:	2		•				
	MODEL	•	-3.8		•		2					
	ANC	9 8	9	2		-	=	•				
ω	DISK	pst.	8		Ξ,		s	=	,			
IS S DEVICE	RPM	MR/TR	1433/0		•		•					
ONTINUED) FEST RUN ALTERING	VTUN	KNOTS	80		= TO	•	•	•				
TABLE 2(CONTINUED)  LIST OF TEST RUNS  EVALUATION OF WAKE-ALTERING DEVICES	CONFIGURATION/CONDITION		Kl3+Hl.0+E5.0 (150 psi)/Bifurcated duct with 150 psi air	K13+H1.0+E5.0	K13+H1.2.1+I1+	K <sub>13</sub> +H <sub>1.2.2</sub> +I <sub>1</sub> 207 exc	Kl3+Hl.15.1+I1+El.0 (0 psi)/Repeat of 189	Kl3+Hl.14.1+I1+El.0 (0 psi)/Like 189 and 210 except cap is open				
	RON	NO.	204	205	207	208	210	211		85		

TABLE 3
INDEX TO RAKE POSITIONS

RUN NUMBER	TEST POINT	WATER LINE	MODEL STATION	BUTT LINE	LOCATION
111	20 21 22	53.5	103.1	-7.25 "	1
	24 26		107.0		
	28 30		111.0	"	
	32 34 36	" "	114.9 116.9 118.9		
112	2 4	48.9	107.3	-7.25	1
	6 8	52.7 54.5	103.3		
	10	56.2 57.2			
113	2 4	51.7 52.3	103.3	-3.25	1
	6 8	52.8 53.3	" "		
	10	53.9			
114	2 4	44.5	103.0	-3.25	1
	6 8	48.2			
	10	51.9			
115	3 4 6 9 10 12	52.9 52.0	124.7	-3.25	1
	6 9	50.0 48.0			
	10	46.0			
	14	42.1 53.0			
	18 20	54.0 55.0		1 :	

TABLE 3 (CONTINUED)

INDEX TO RAKE POSITIONS

RUN NUMBER	TEST POINT	WATER	MODEL STATION	BUTT	LOCATION FIGURE
116	7	36.9	100.5	-17.5	1
117	2	37.6	100.5	-16.0 -14.0	1
	6 8	37.3	99.6	-12.0	
	8			-10.0	
	10	•		- 8.0	
118	2	37.6	100.5	- 6.0	1
119	2 5 8 9	37.3	99.6	+ 6.0	1
	5		and the		
	9	2,001		10	
				14	
	16	1	n	16	
	20 25	51.5 52.3	102.5	17.5	
			101.7	-17.5	
121	3	62.9	129.0	+ 5.7	2
	4	53.5	97.0		
	6 8	46.0			
	10	42.1			
135	2	56.9	106.3	- 5.7	3
	4	54.5	•		
	6 8	52.5 50.5			
	10	48.5			
	12	46.5			
	14	44.5	• 11 - 12	•	
136	2	56.5	104.0	- 8.0	4
	6	54.5		•	
		52.5	:		
	8 10 12	50.6			
	12	46.5		•	
	14	44.5		•	
	17	37.1			
	19	39.0 41.0			

TABLE 3 (CONTINUED)

INDEX TO RAKE POSITIONS

RUN NUMBER	TEST POINT	WATER LINE	MODEL STATION	BUTT	LOCATION FIGURE
137	3 5 7 9 11 13 15 17	38.7 39.9 42.0 44.0 46.0 48.0 50.0 52.0 54.0	98.4 100.5 103.6		5
138-41, 143	2 3 4 5 6 7 8 9	38.8 40.0 42.0 44.0 46.0 48.0 50.0 52.0 54.0	98.4 100.5 103.6	- 8.0	5
142	7 8 9 10 11 12 13 14 15 16 17	37.8 40.2 42.0 44.0 46.0 48.0 50.0 52.0 54.0 56.8	98.4 " 100.5 103.6	- 8.0	5

TABLE 3 (CONTINUED)

INDEX TO RAKE POSITIONS

RUN NUMBER	TEST POINT	WATER LINE	MODEL STATION	BUTT	LOCATION FIGURE
149-151	2 3 4 5 6 7 8 9	38.8 40.0 42.0 44.0 46.0 48.0 50.0 52.0 54.0	98.5 100.6 103.5	- 8.0 "	5
152-6, 158 161-4, 166 167, 169-71 175, 177-9 180,182,184 186-8, 190 191,193,194 196,198,201 204,207,208	2 3 4 5 6 7 8 9	42.9 44.9 46.9 48.9 50.9 52.9 54.9 56.9	97.9 100.6 104.6	0.0	6
159	1 2 3 4 5	54.9 52.9 50.7 48.6 46.7	104.6	0.0	6
160,203	5 6 7 8 9 10	42.9 44.9 46.9 48.9 50.9 52.9 54.9	97.9 100.6 104.6	0.0	6
165	3 4 5 6 7 8	44.9 42.9 46.9 48.9 50.9 52.9	97.9 100.6 104.6	0.0	6

TABLE 3 (CONTINUED)

INDEX TO RAKE POSITIONS

RUN NUMBER	TEST POINT	WATER LINE	MODEL STATION	BUTT LINE	LOCATION FIGURE
168, 183	4	42.9	97.9	0.0	6
	5	44.9	"	"	
	6	46.9	100.6		
	5 6 7 8	48.9	1 "		
	8	50.9	104.6	n	
	9	52.9		п	
	10	54.9	0.008		
172	3 4 6 7 8 9	42.9	97.9	0.0	6
	4	44.9	"		
	6	44.9		,	
	,	46.9	100.6	п	100 100
	8	50.9	104.6	11	
	10	52.9	104.0		
	11	54.9	tt		
		<b></b>	+		<del> </del>
173,174,176	1	42.9	97.9	0.0	6
185,195,197	2	44.9			
199,200,205	3	46.9	100.6	Ü,	
210	4	48.9	104.6		
	6	52.9	104.6		
	1 2 3 4 5 6 7	54.9			
181	2	42.9	97.9	0.0	6
	2 3 4 5 6 7 9	44.9		"	
	4	46.9	100.6		
	5	48.9			The second second
	6	50.9	104.6	•	
	7	52.9	"	"	
	9	54.9			
	10	"			
	11				
	12 13	42.9	97.9		
- 2	13	42.5	37.3		384
		a.cor	3,08		
		0.001	2.02		
			2.52		

TABLE 3 (CONTINUED)

INDEX TO RAKE POSITIONS

	-1				
RUN NUMBER	TEST POINT	WATER	MODEL	LINE	LOCATION FIGURE
189	29 30 31 32 33 34 35 36 37 38 39	42.9 44.9 46.9 48.9 50.9 50.9 52.9 54.9	97.9 100.6 " 104.6 " 100.6 104.6	0.0	6
202	3 4 5 6 7	43.4 44.9 46.9 48.9 50.9	97.9 100.6 104.6	0.0	6

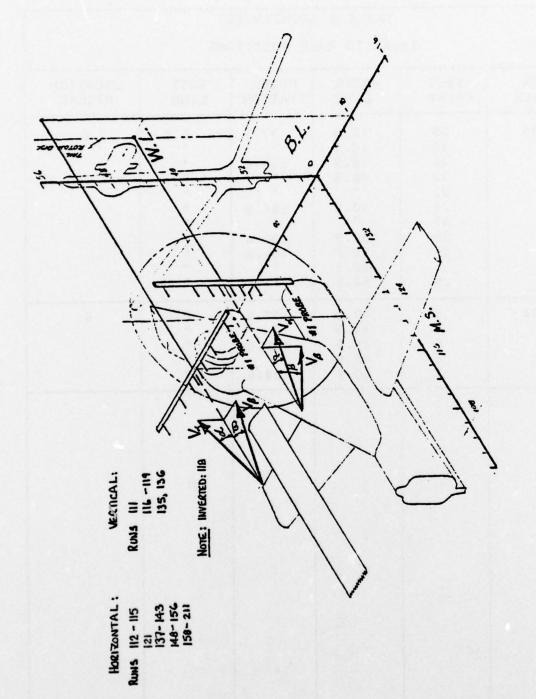
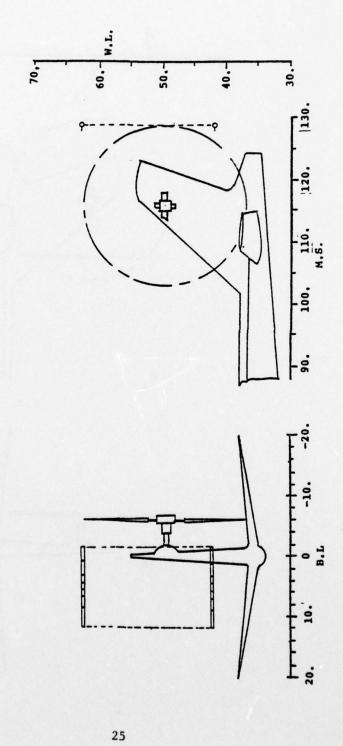


FIGURE 1 - RAKE ORIENTATION DIAGRAM



-HOT FILM RAKE LOCATIONS FIGURE 2



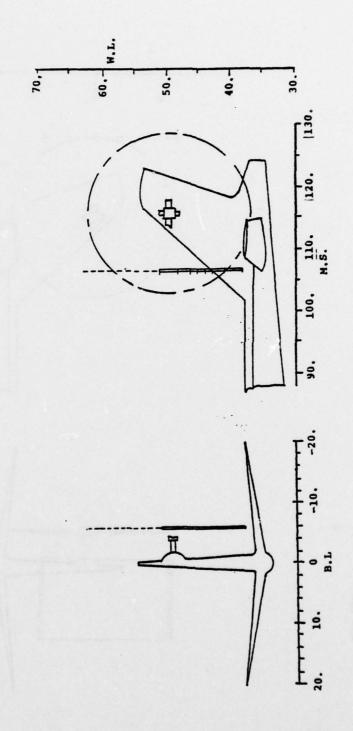


FIGURE 3 -HOT FILM RAKE LOCATIONS



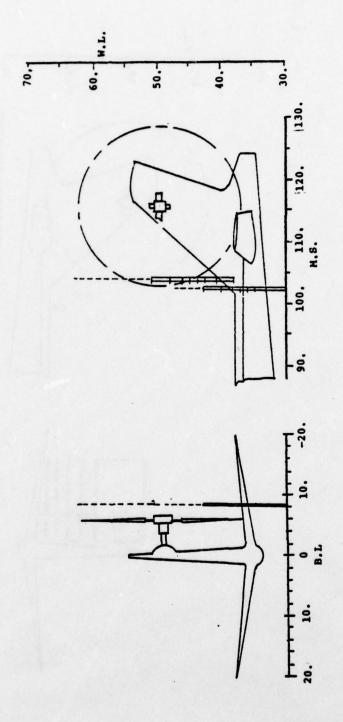


FIGURE 4 -HOT FILM RAKE LOCATIONS

RUN 137, 138, 139, 140, 141, 142, 143, 143, 143, 149, 150, 151

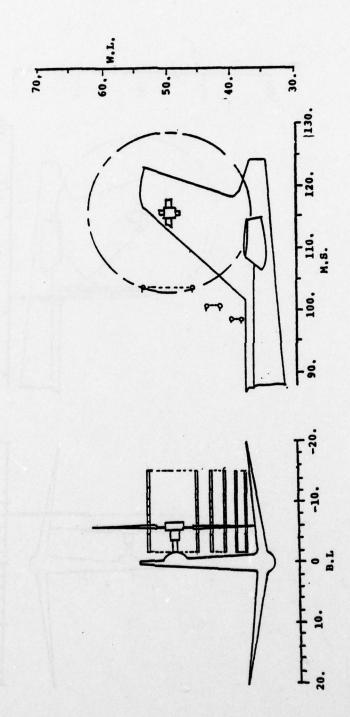


FIGURE 5 -HOT FILM RAKE LOCATIONS

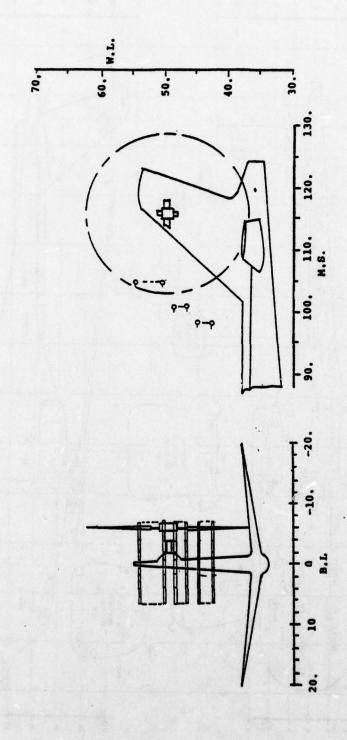


FIGURE 6 -HOT FILM RAKE LOCATIONS

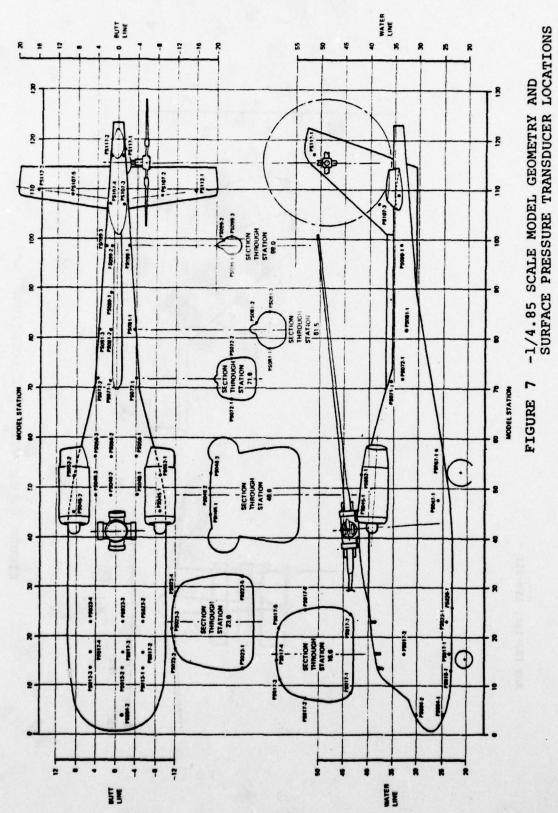
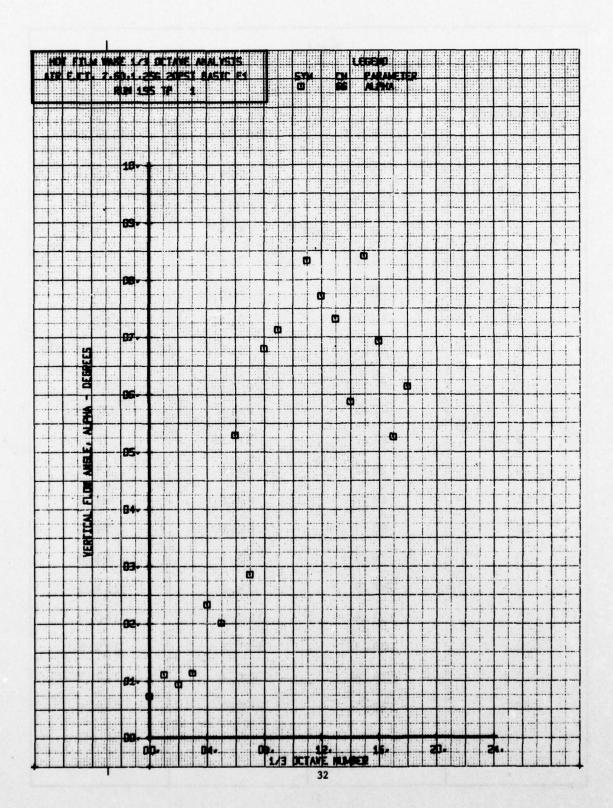
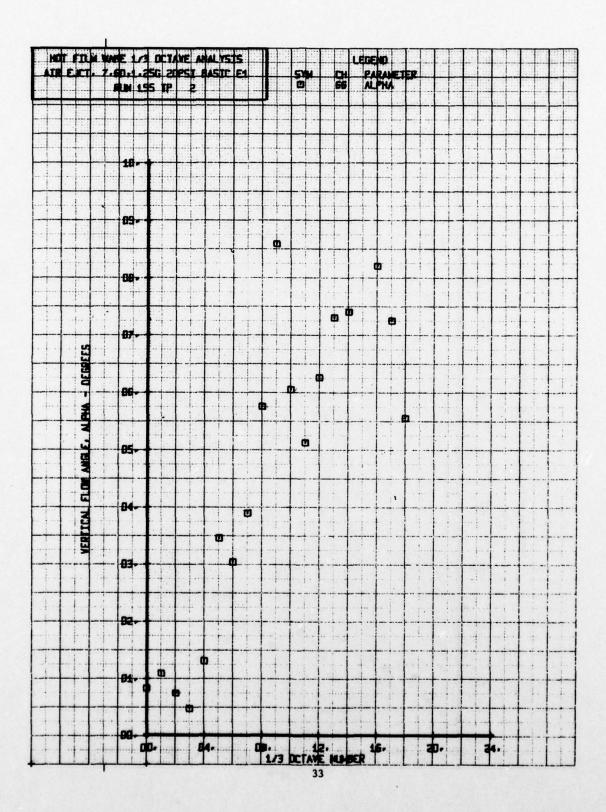
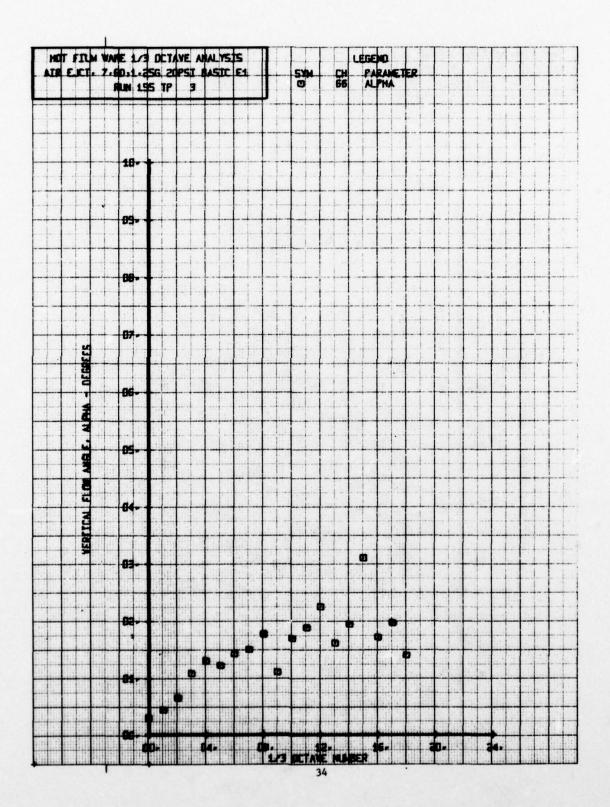


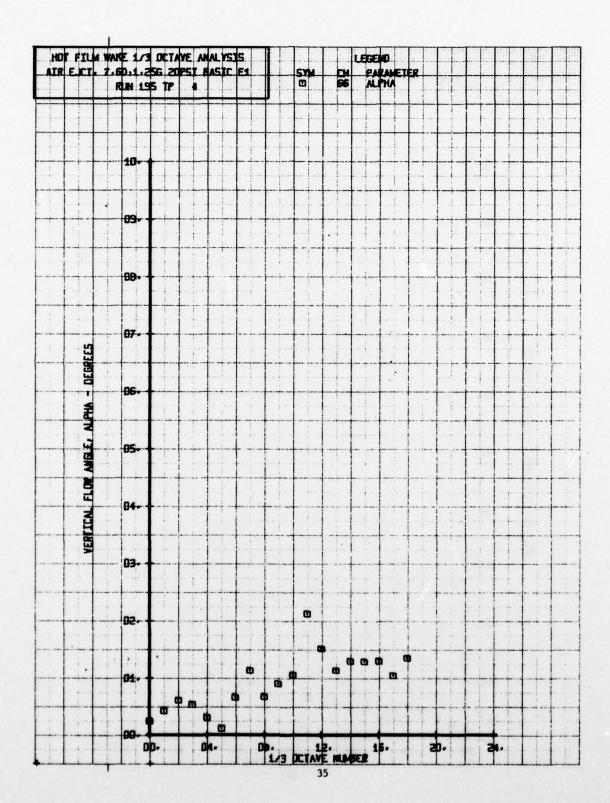
TABLE 4
1/3 OCTAVE BAND IDENTIFICATION

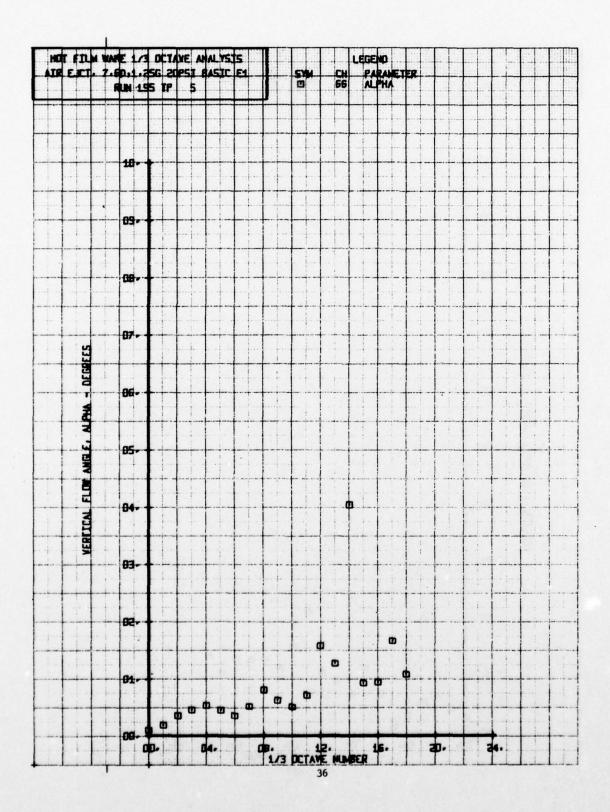
BAND NUMBER		BAND WIDTH - Hz	
BAND NOMBER	MINIMUM	CENTER	MAXIMUM
0	3.5	3.4	4.4
1	4.4	4.9	5.5
2	5.5	6,2	7.0
3	7.0	7.8	8.7
4	8.7	9.8	11.0
5	11.0	12,4	13.9
6	13.4	15.6	17.5
7	17.5	19.7	22.1
8	22.1	24.8	27.8
9	27.8	31.25	35.1
10	35.1	39.4	44.2
11	44.2	49.6	55.7
12	55.7	62.5	70.2
13	70.2	78.7	88.9
14	88.9	99.2	111.4
15	111.4	125.0	140.3
16	140.3	157.5	176.8
17	176.8	198.4	222.7
18	222.7	250.0	280.6

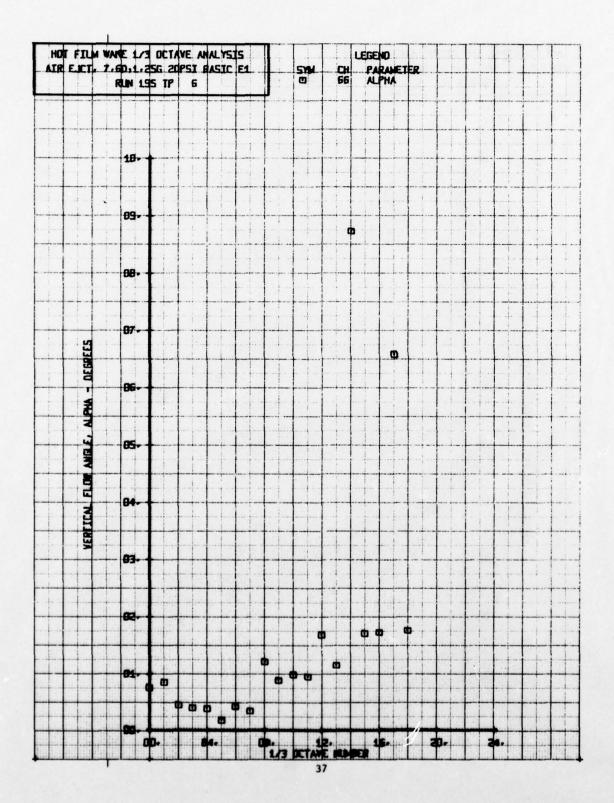


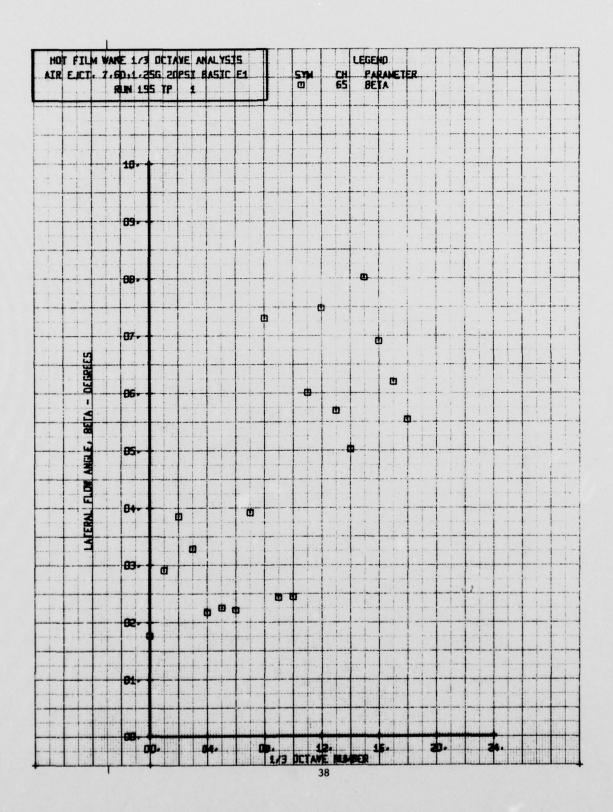


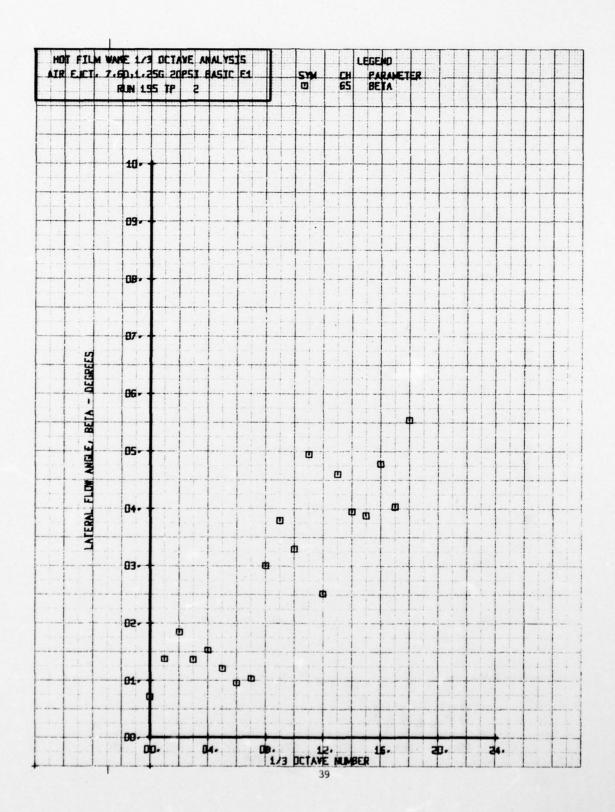


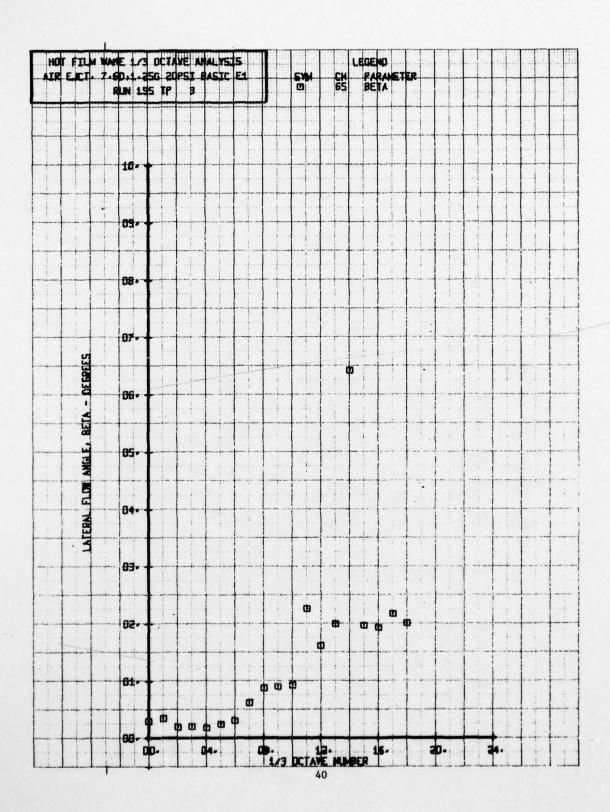


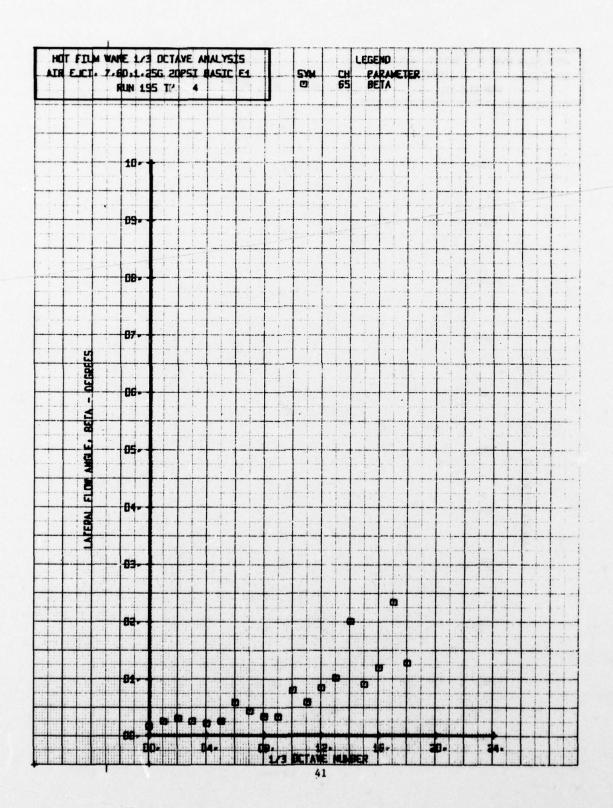


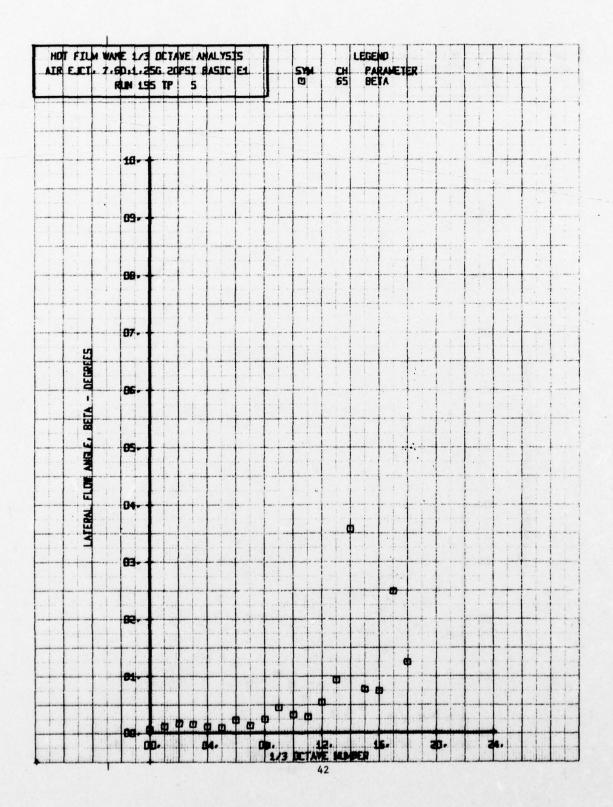


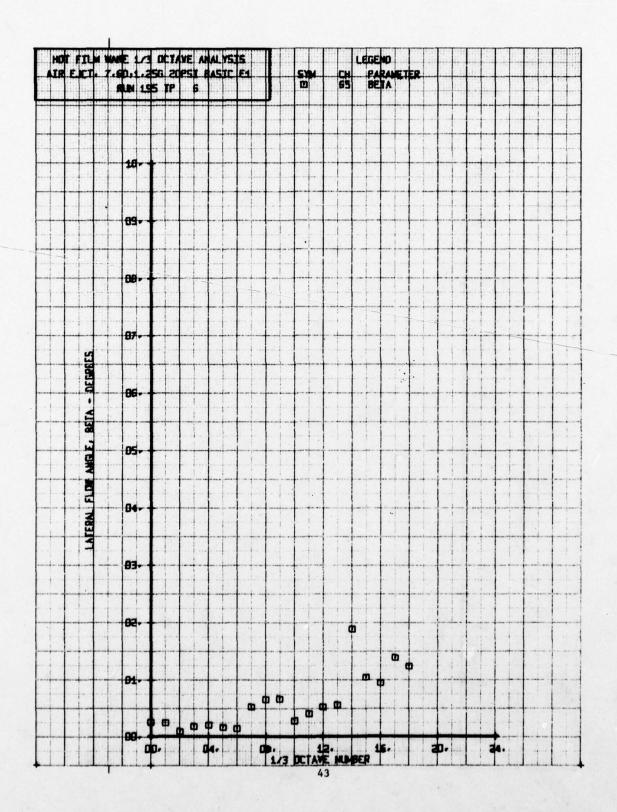


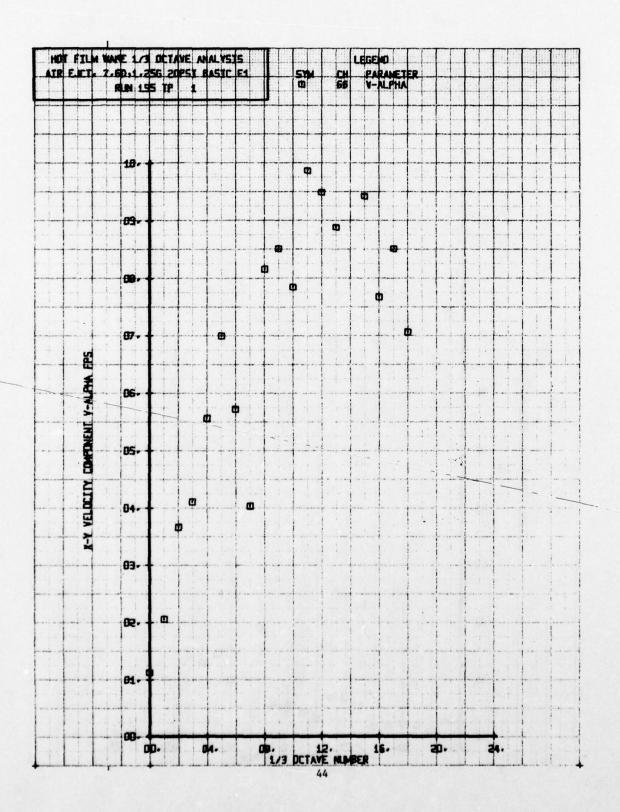


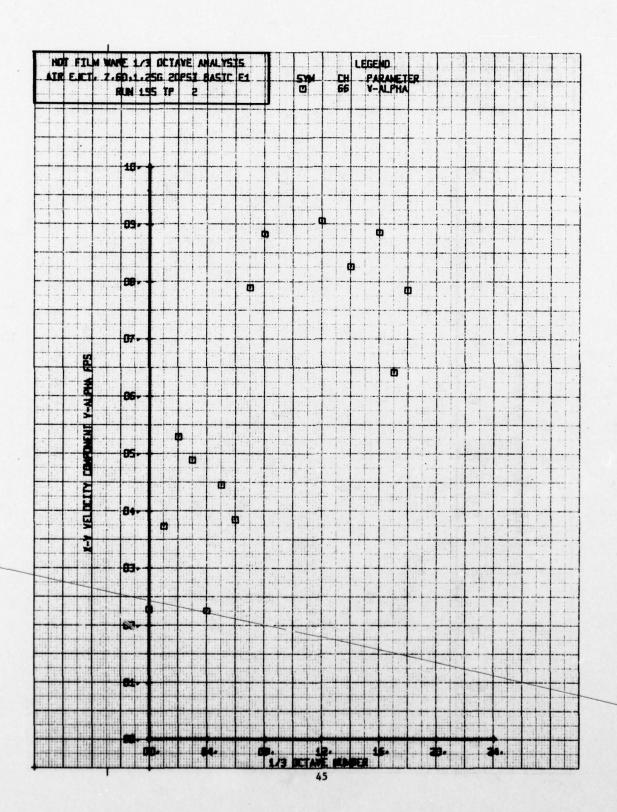


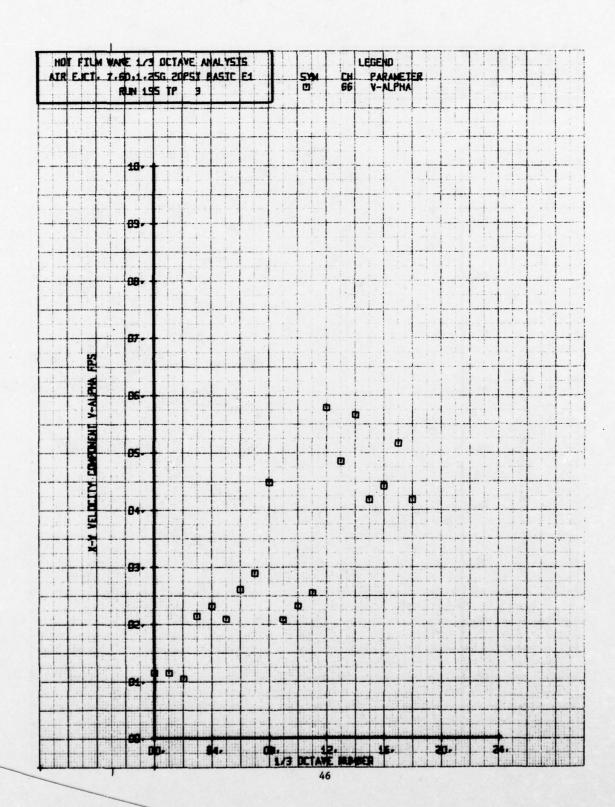


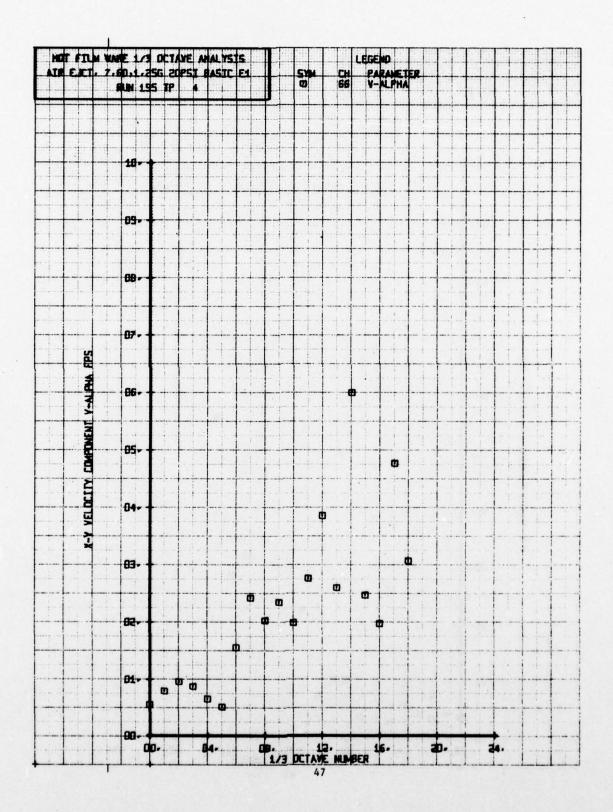


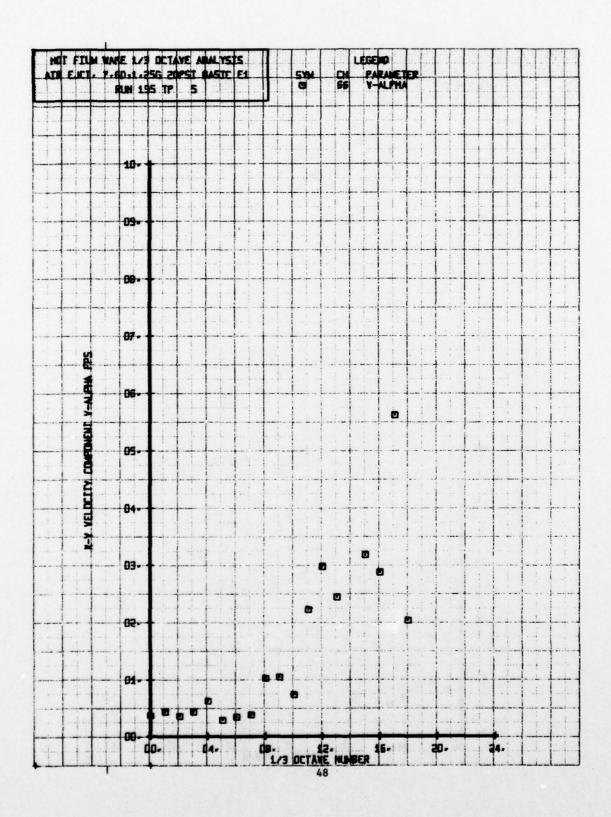




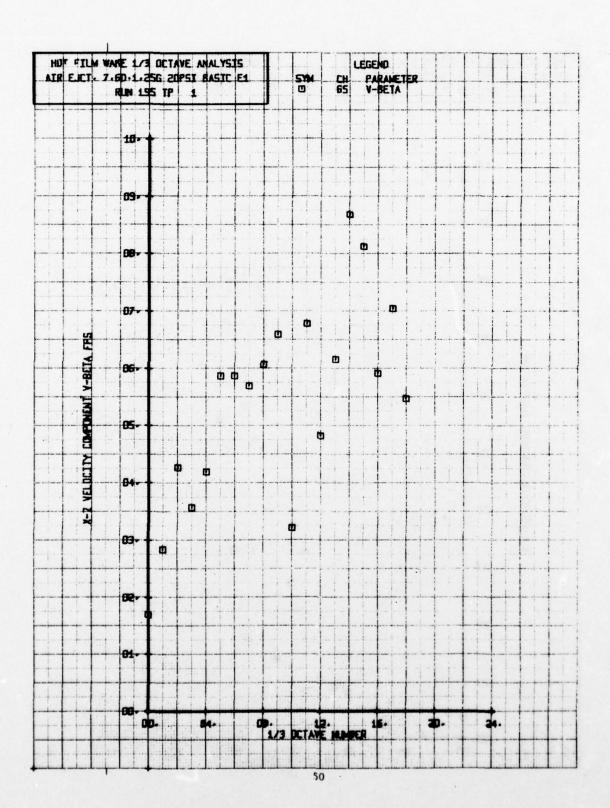


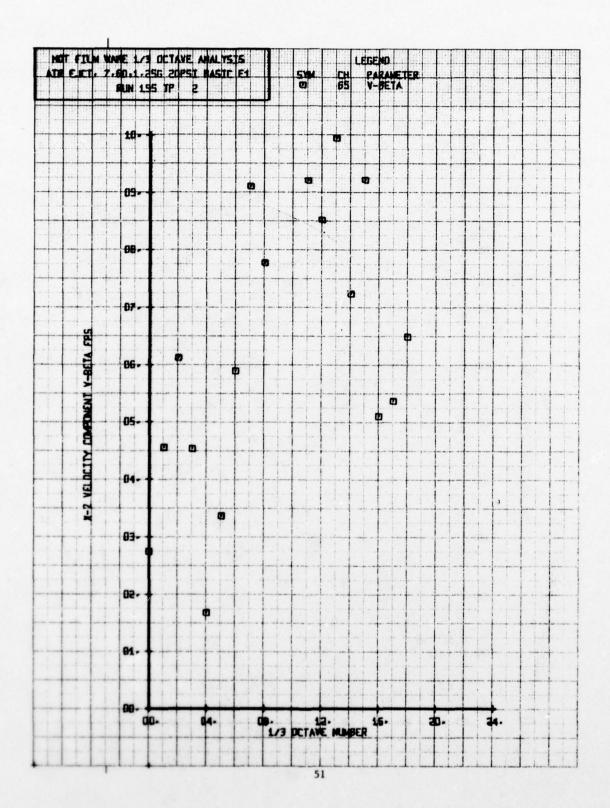


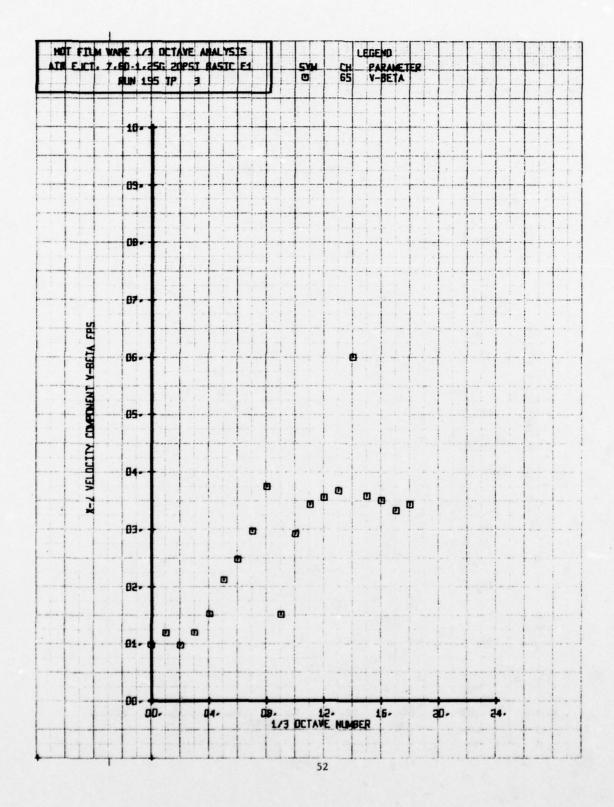


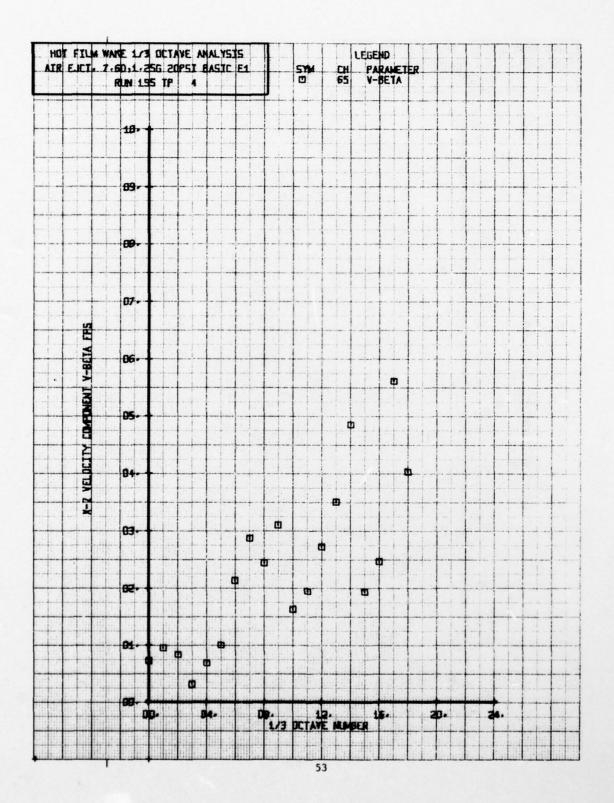


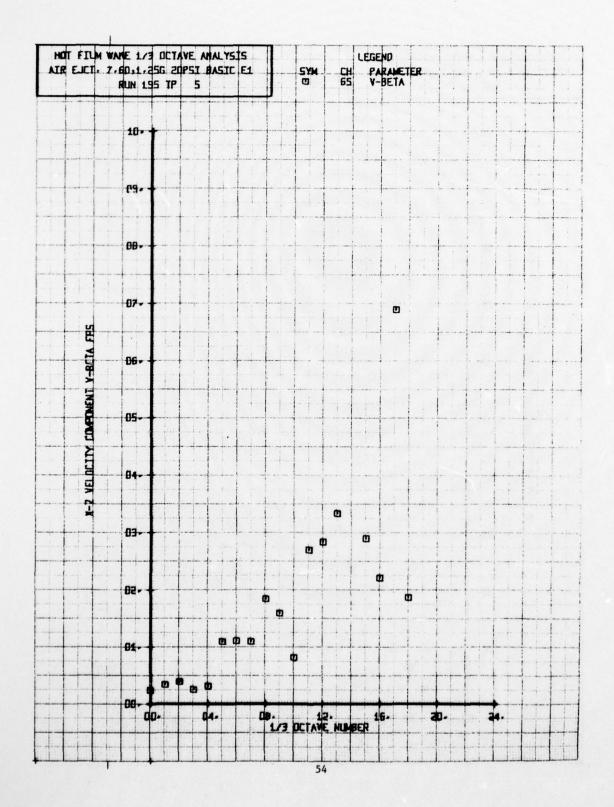
HOT	FIL	NW N	E 1/3	OCT	VE A	MALYS	S		-		LEGE	ND RAMET ALPHA						
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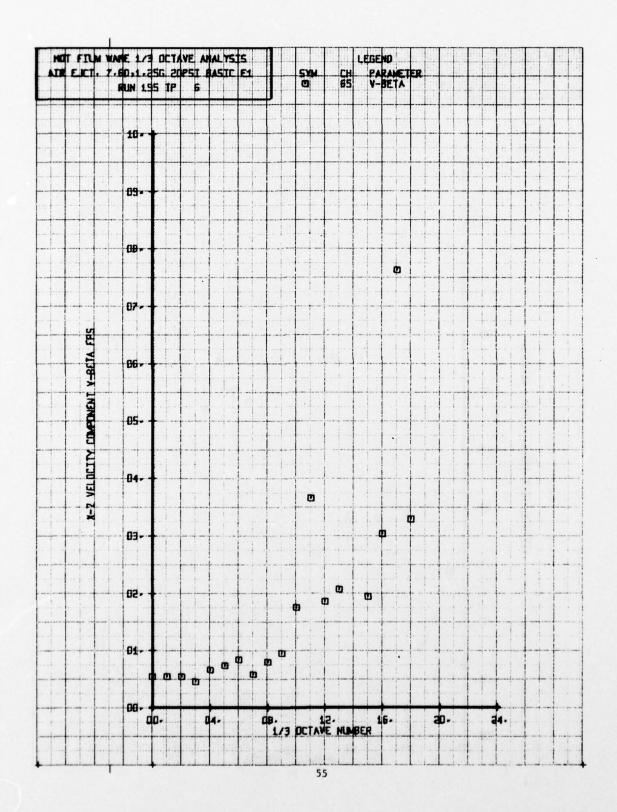


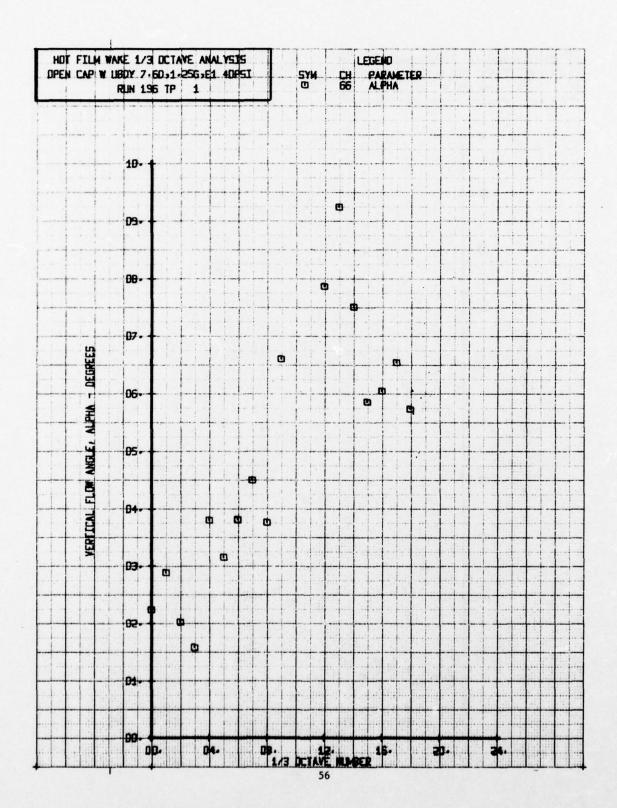


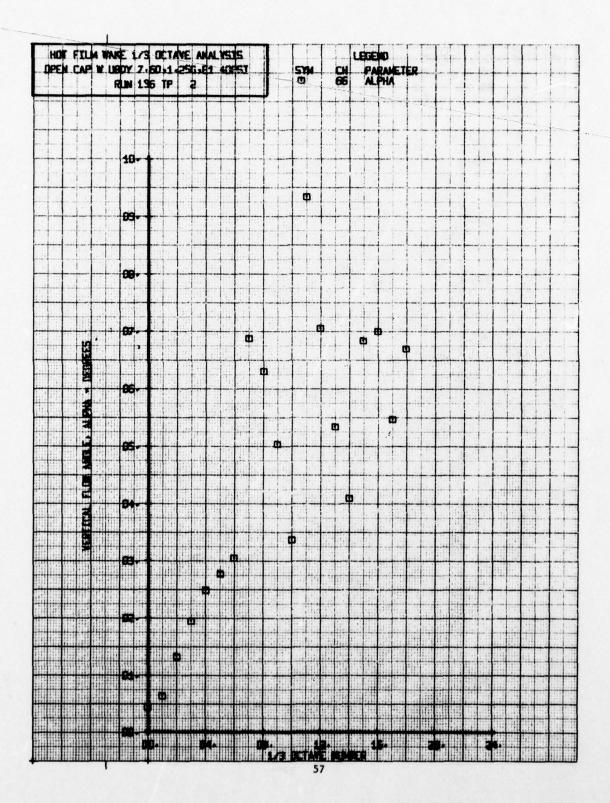


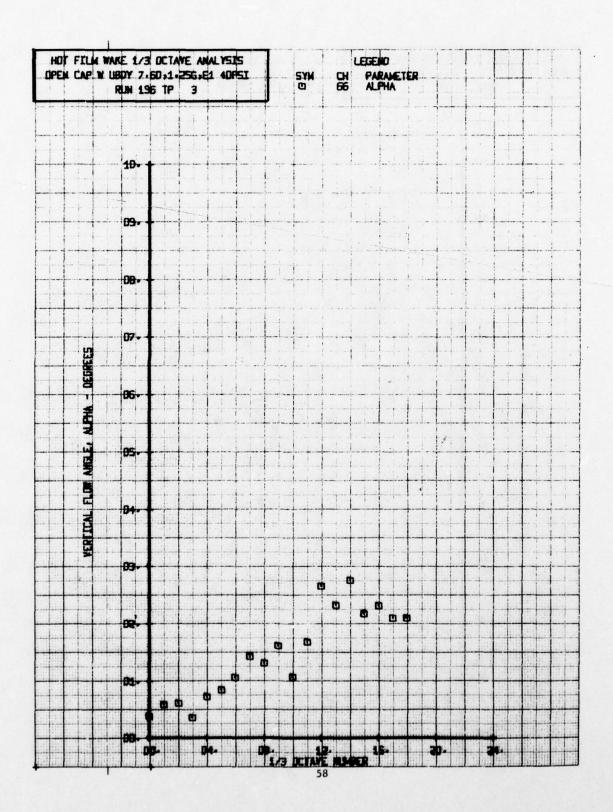


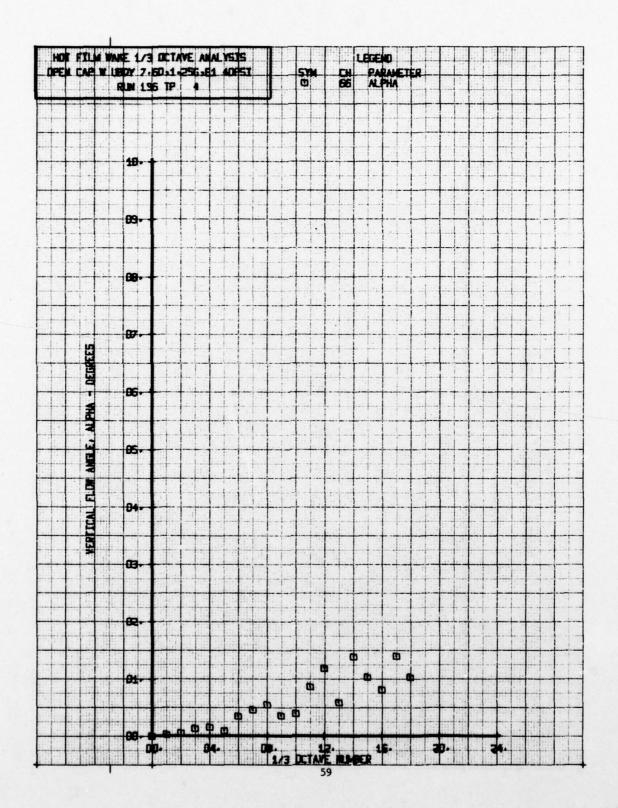


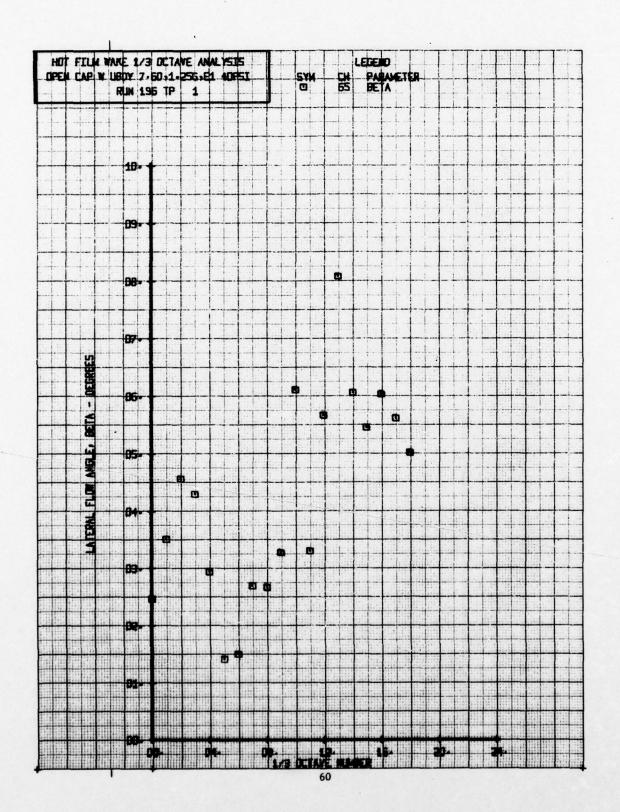


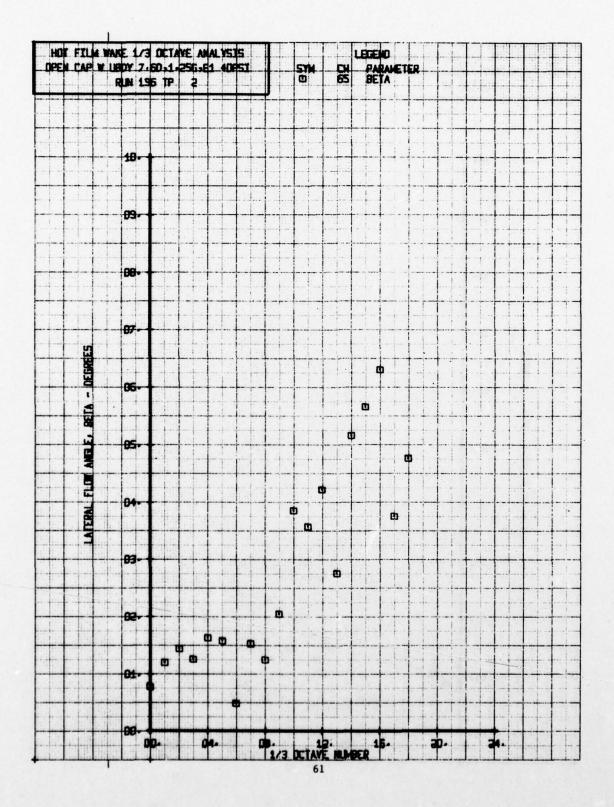


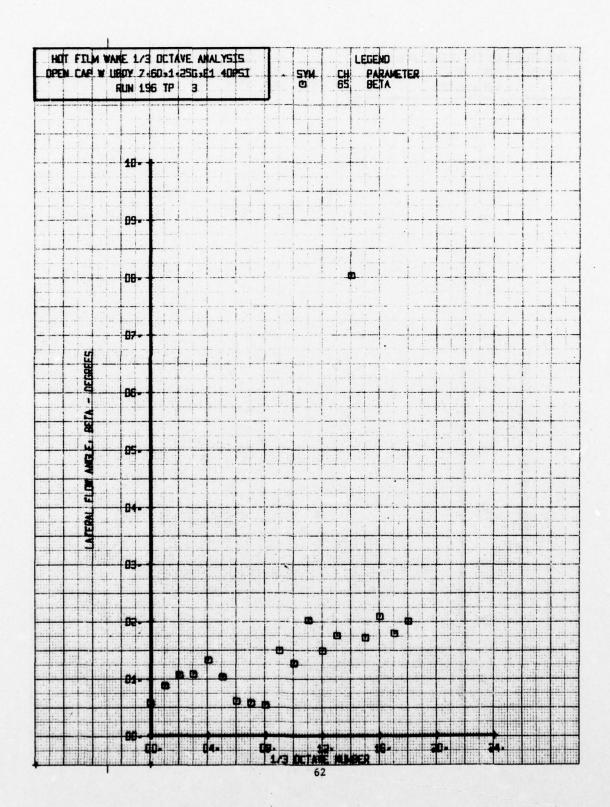


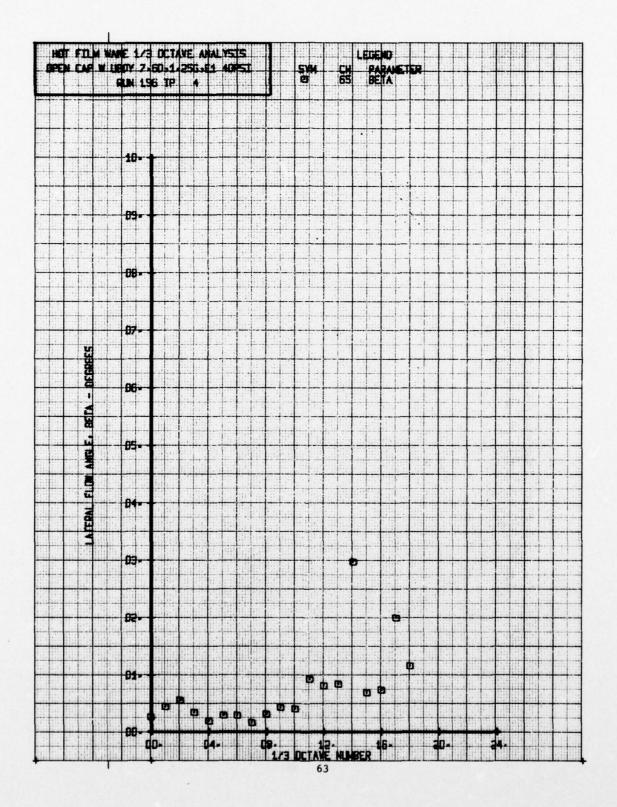


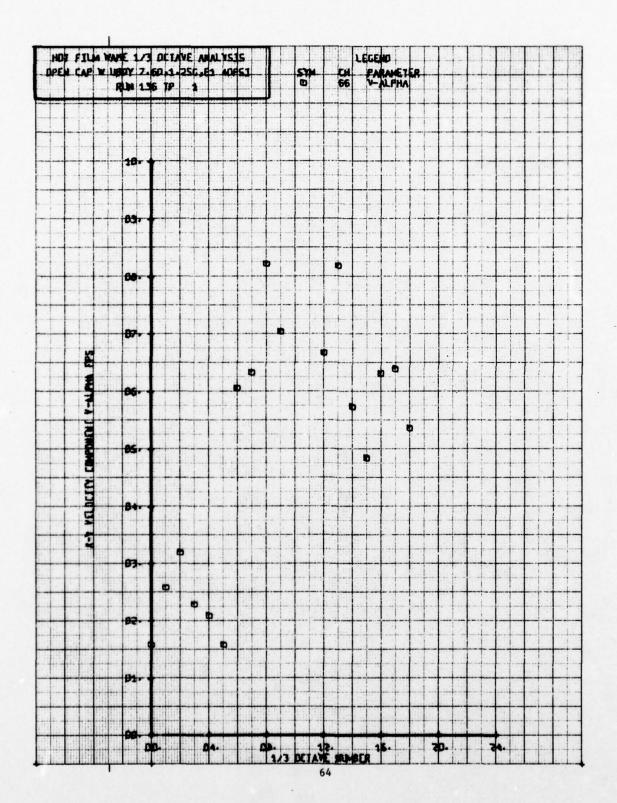


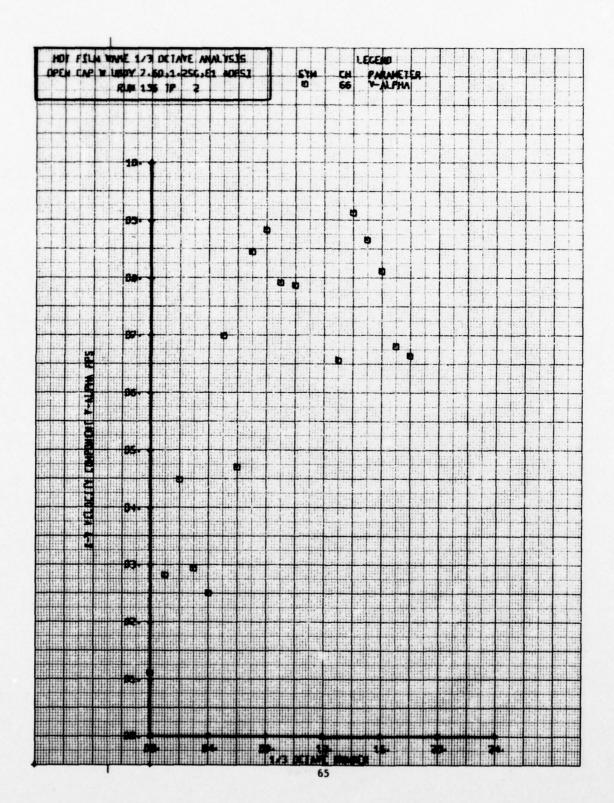


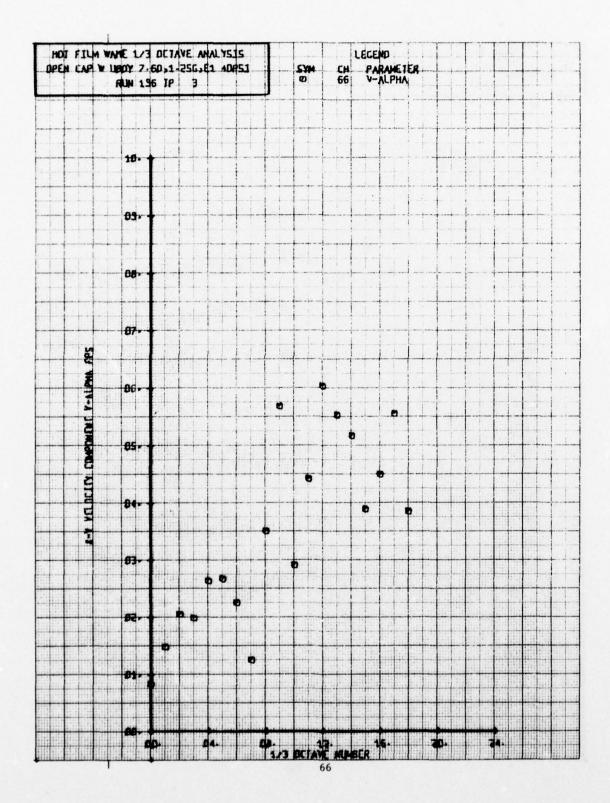


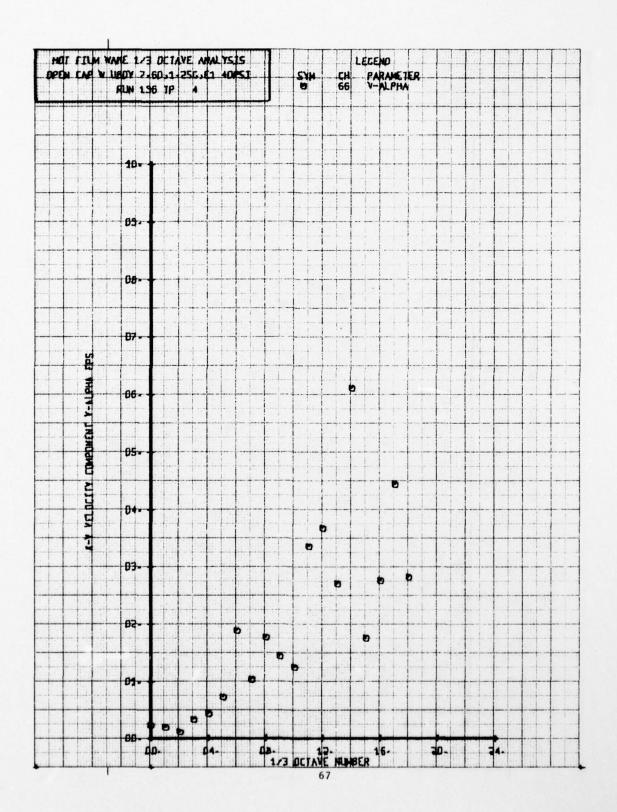


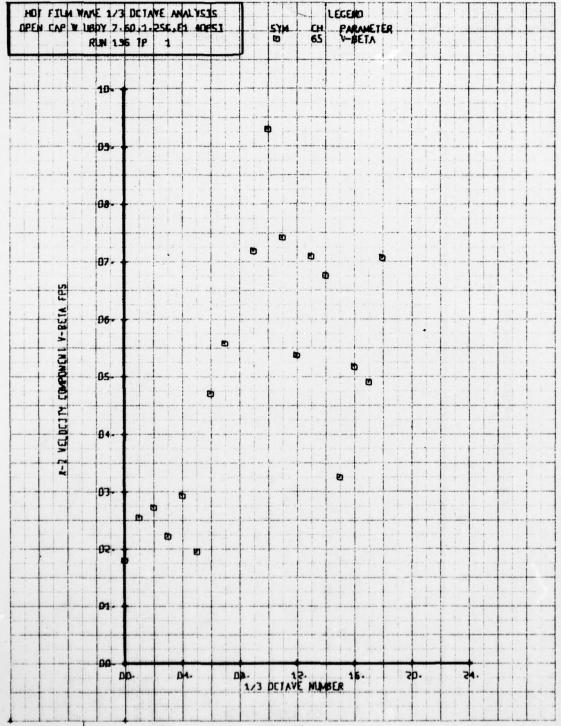


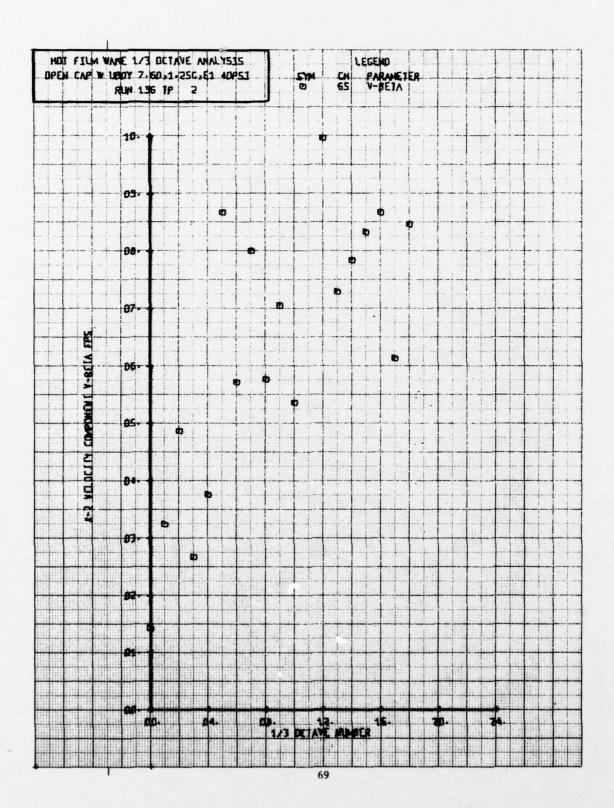


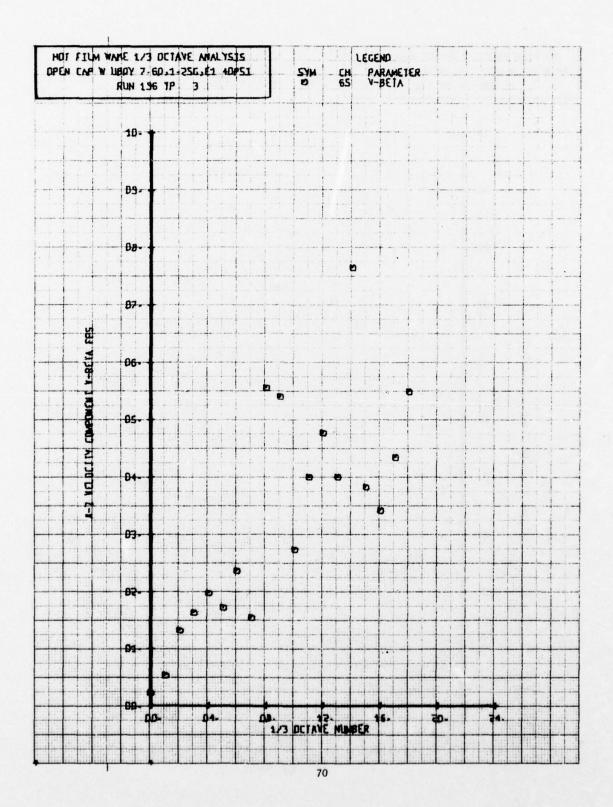


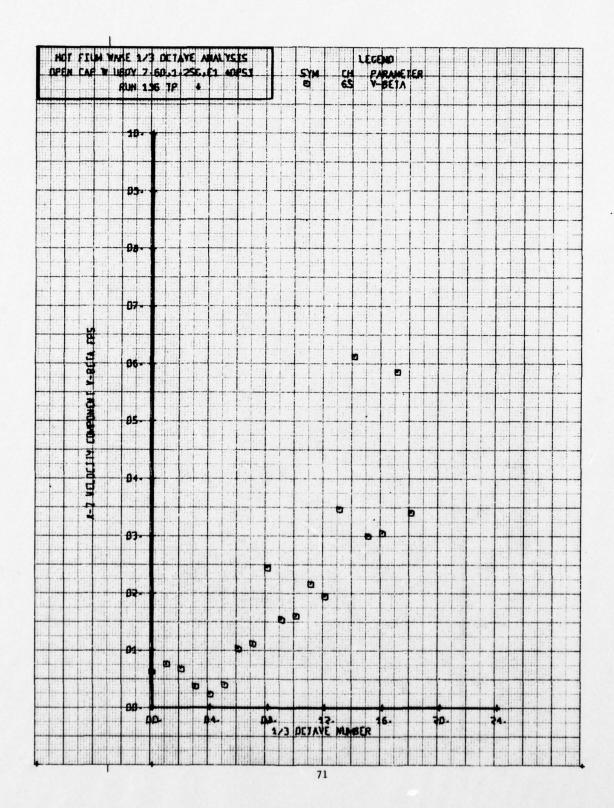


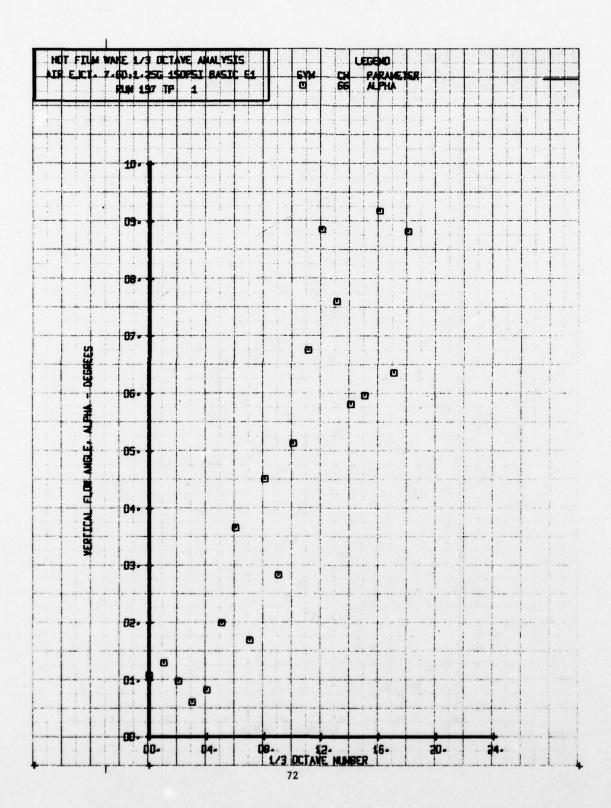




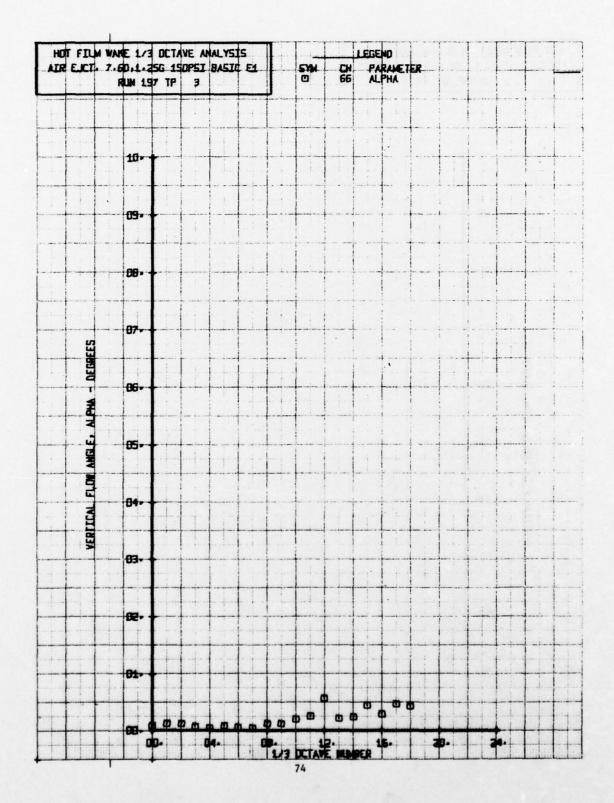


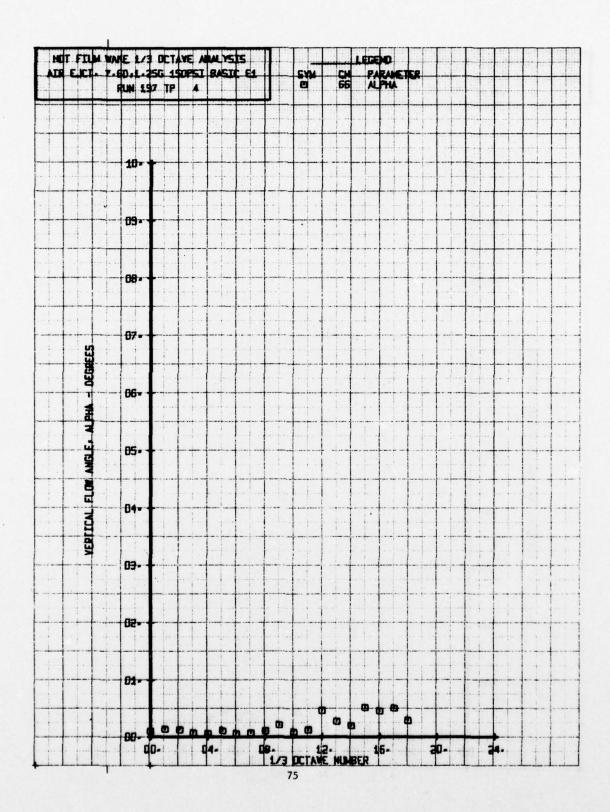


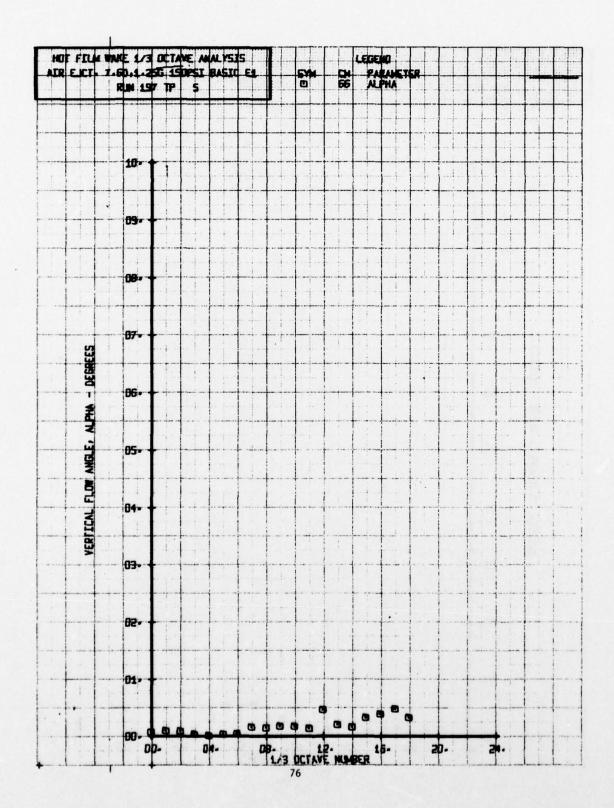


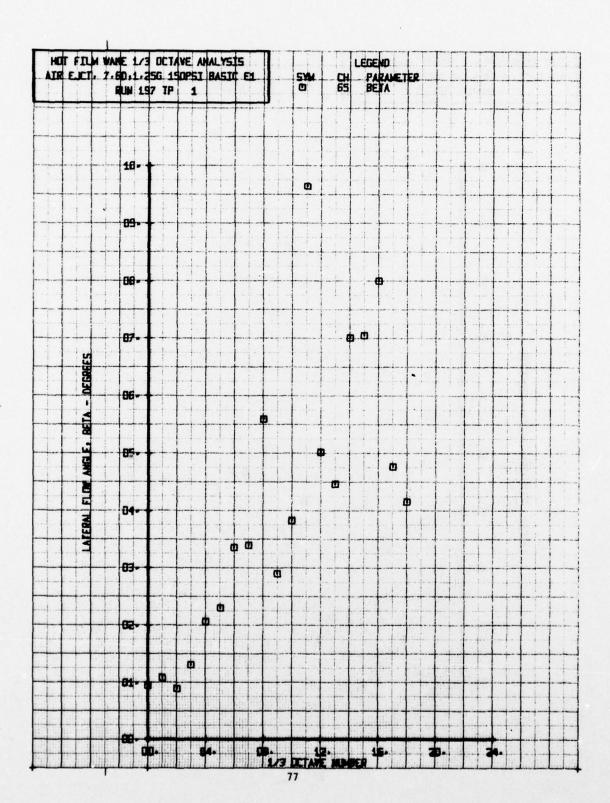


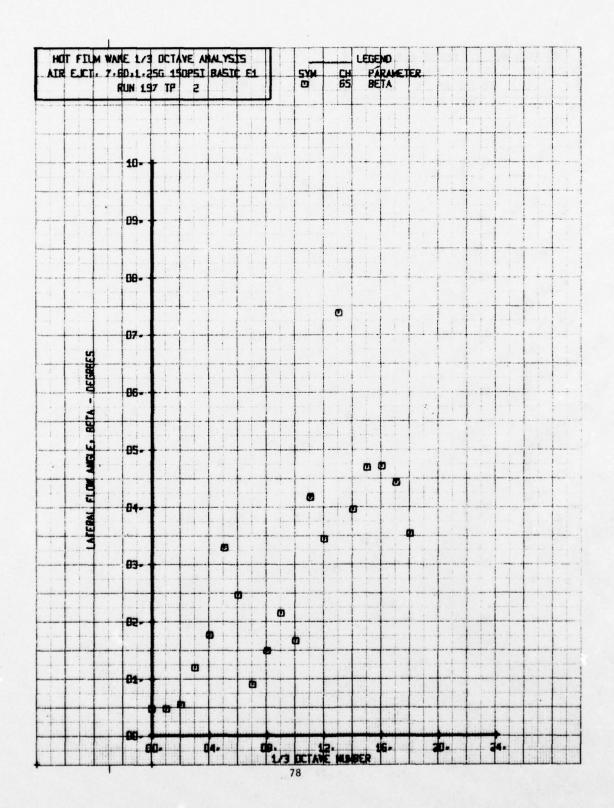
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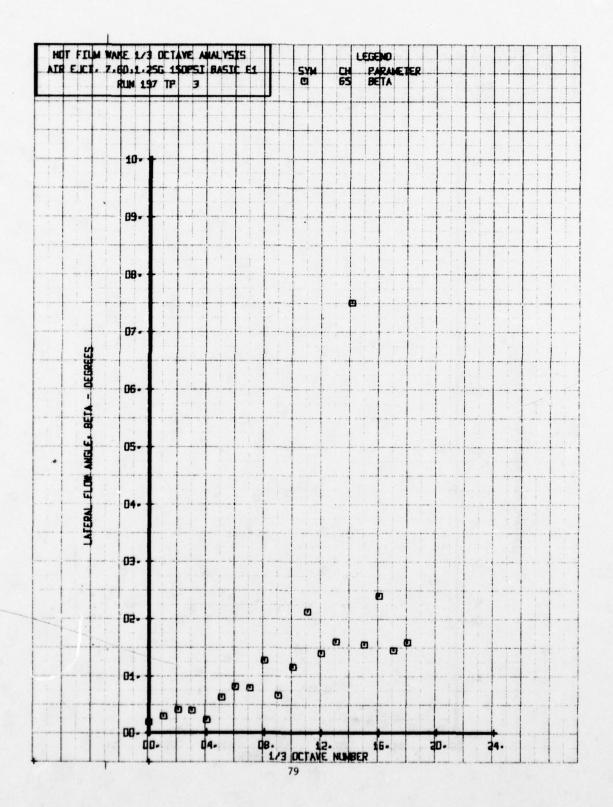


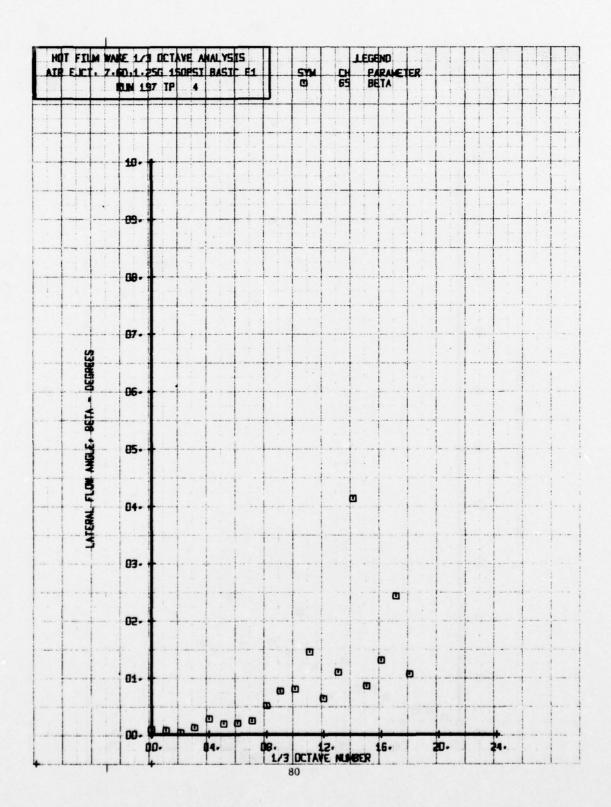


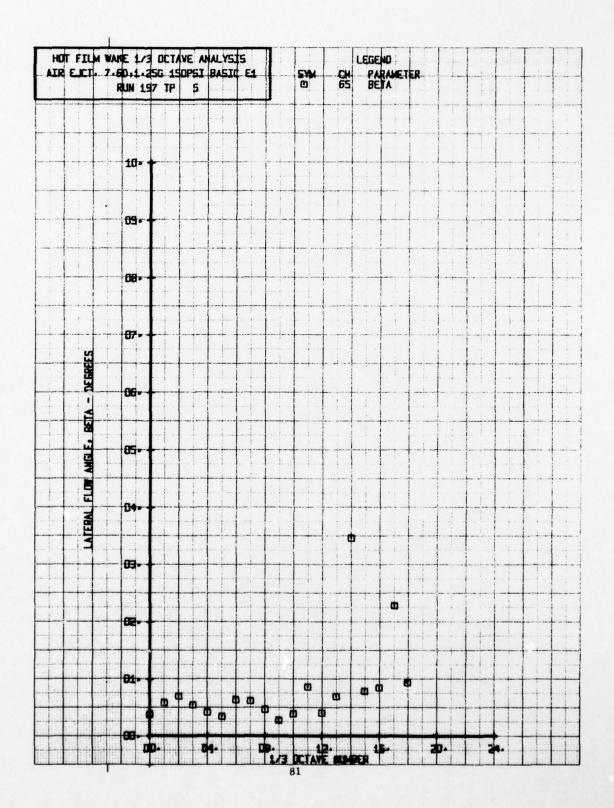


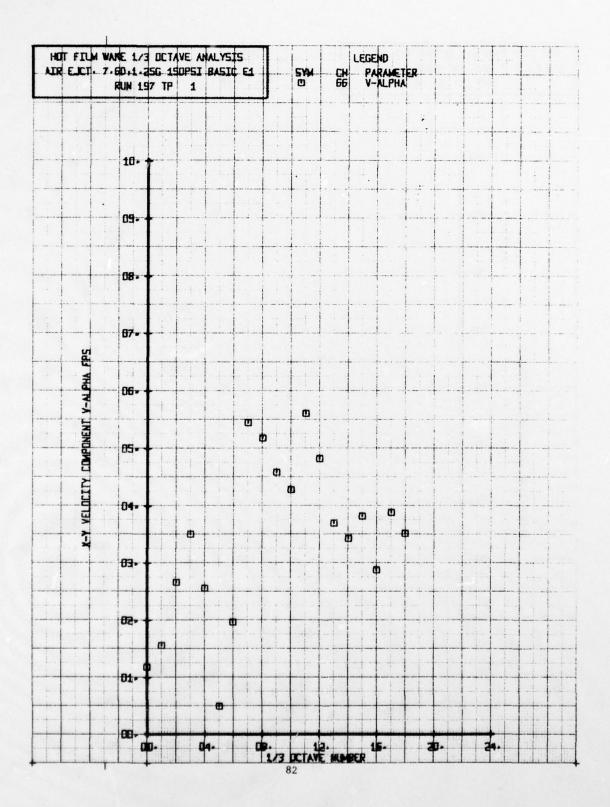


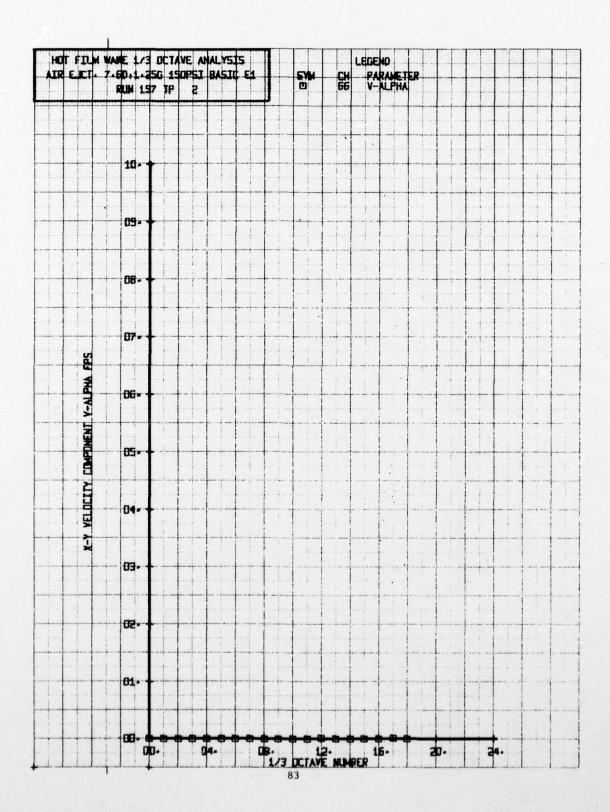


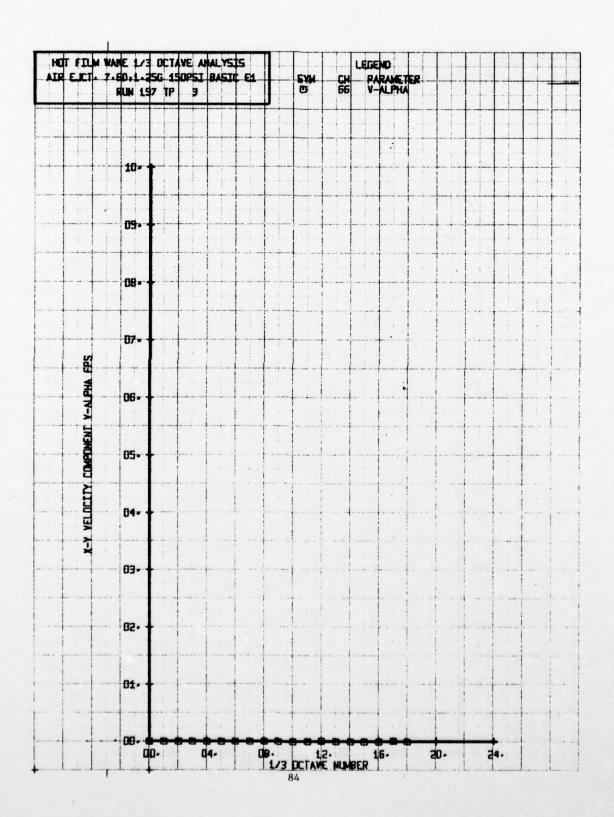


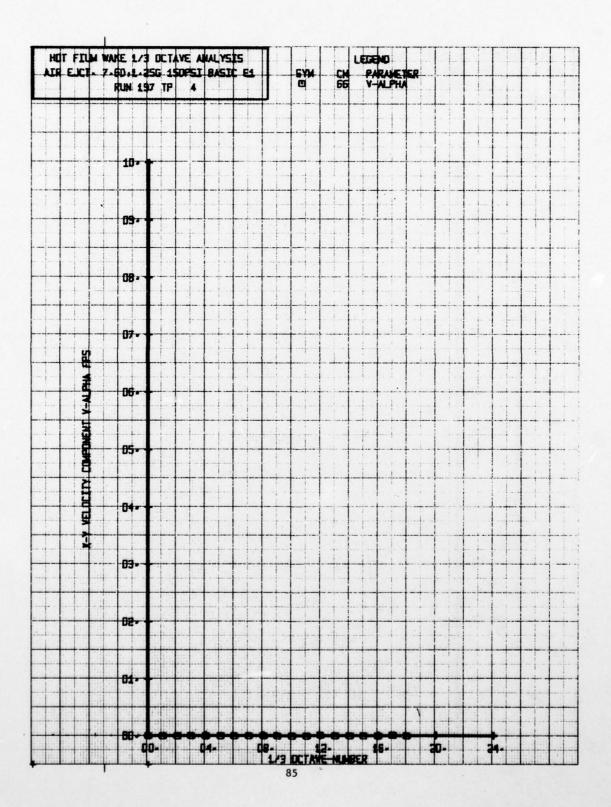


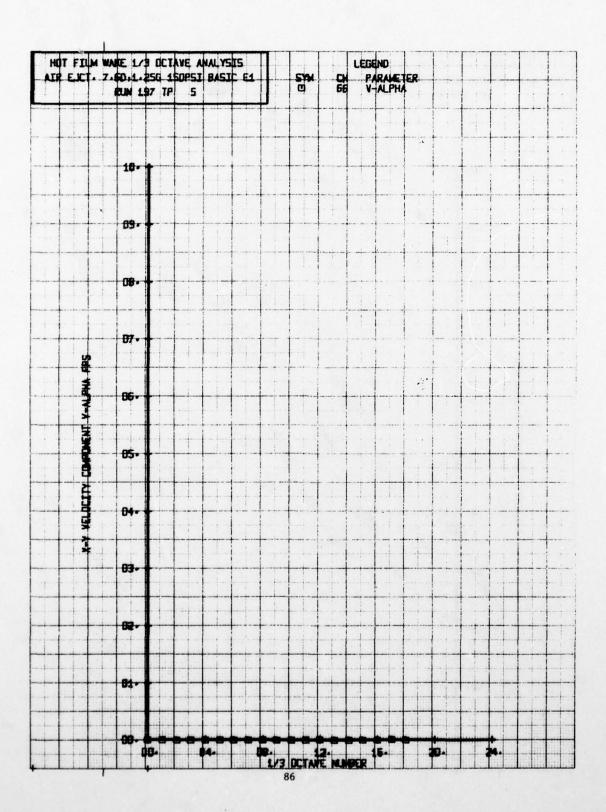


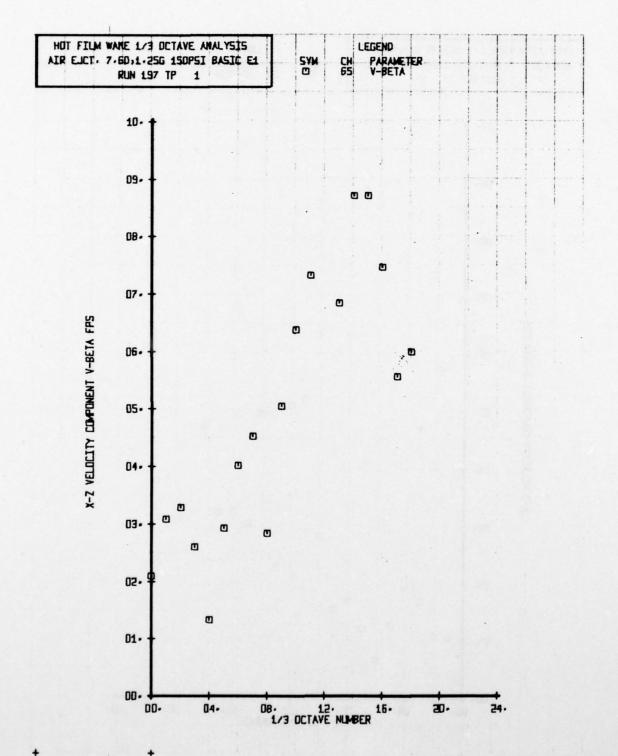


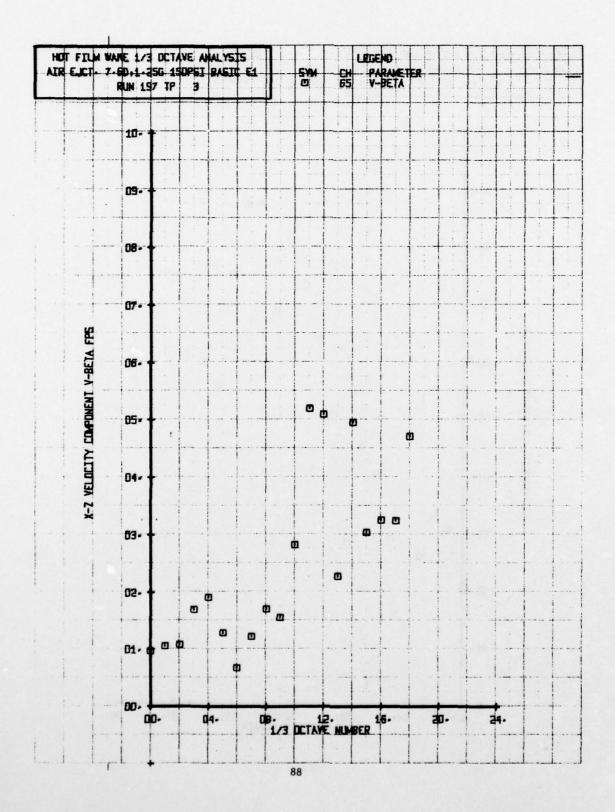












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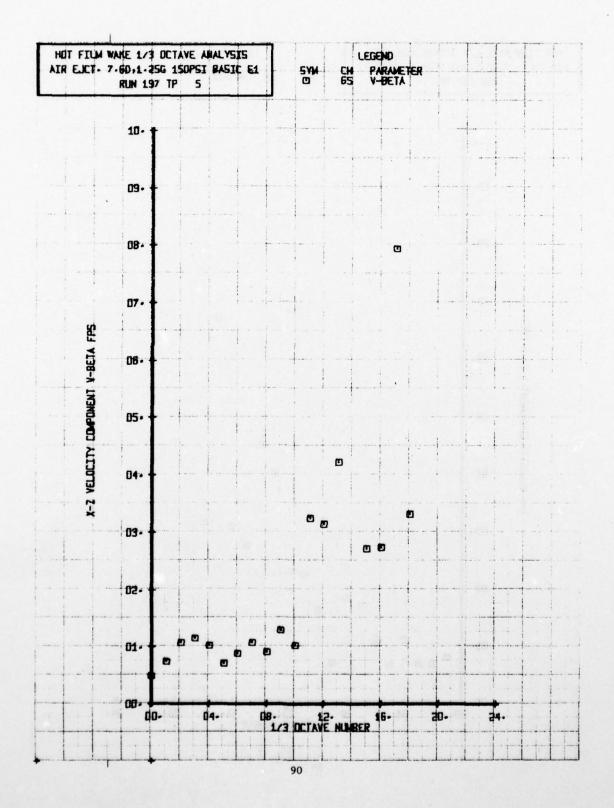
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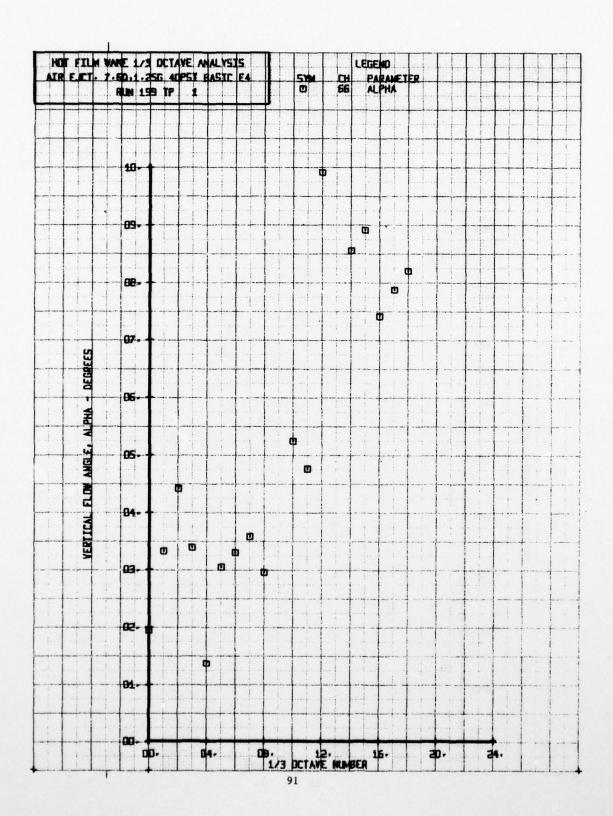
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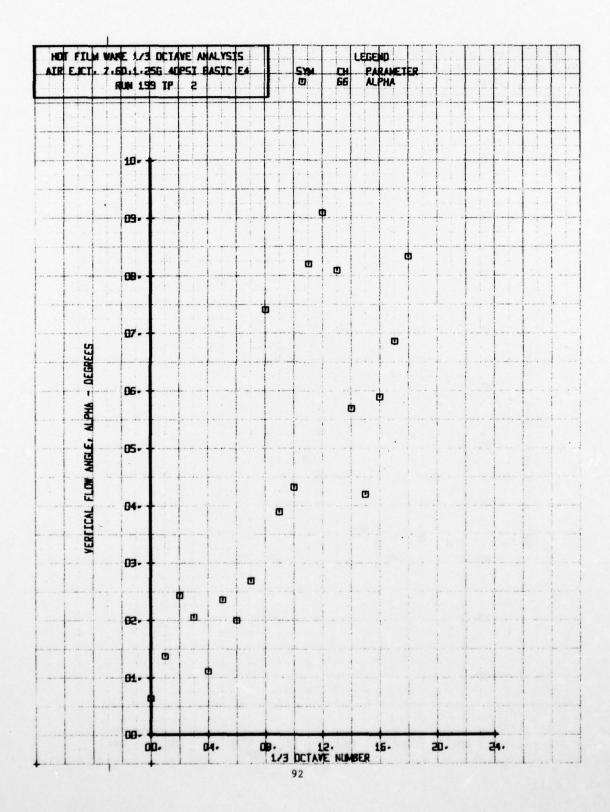
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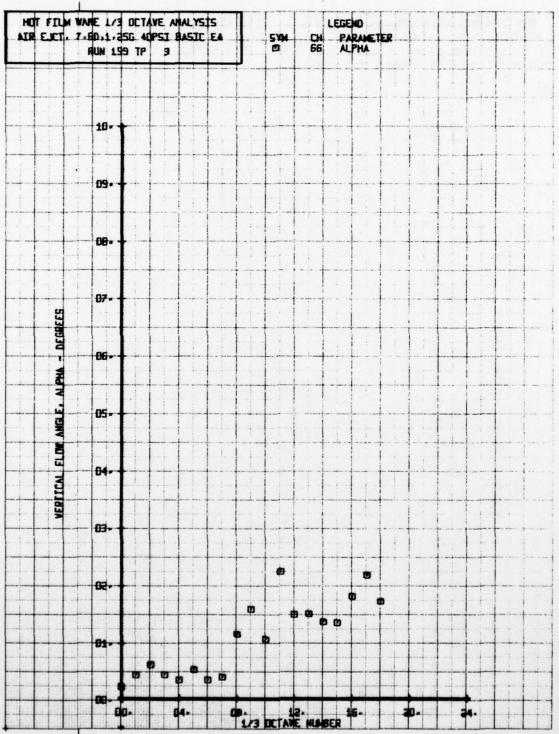
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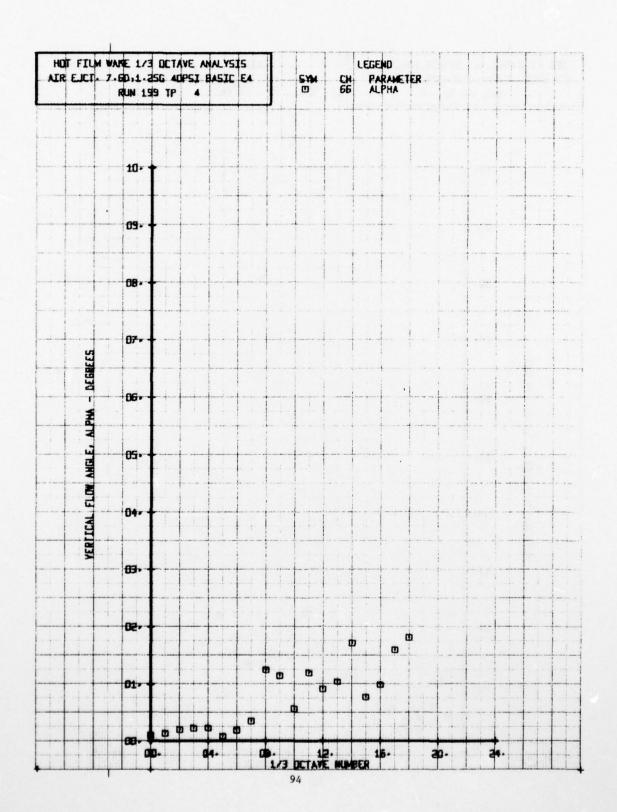
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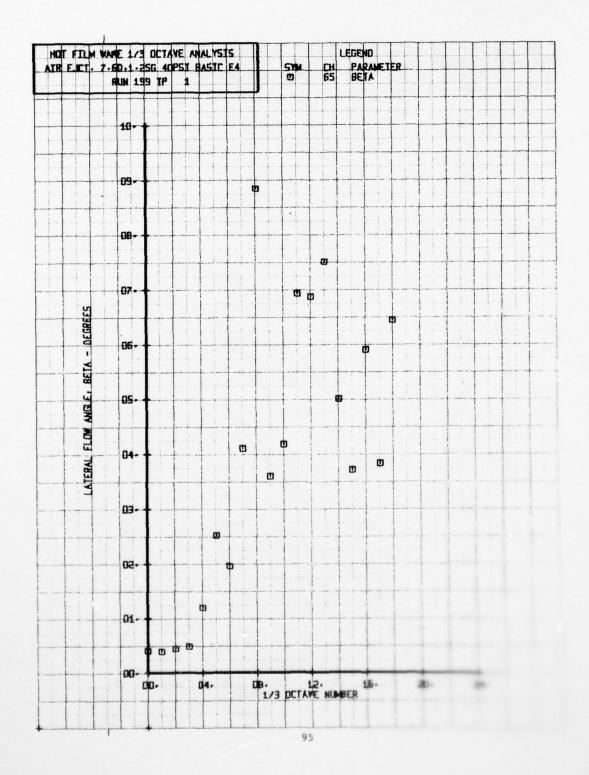










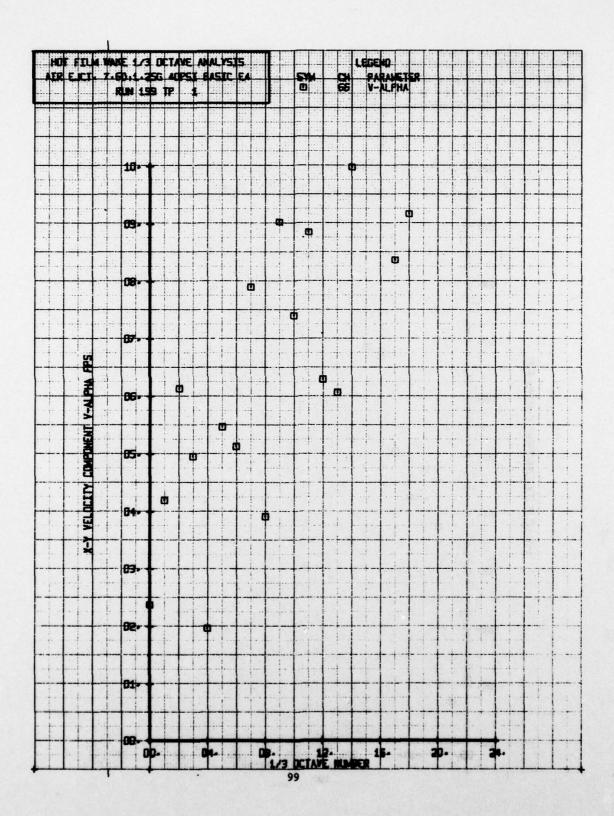


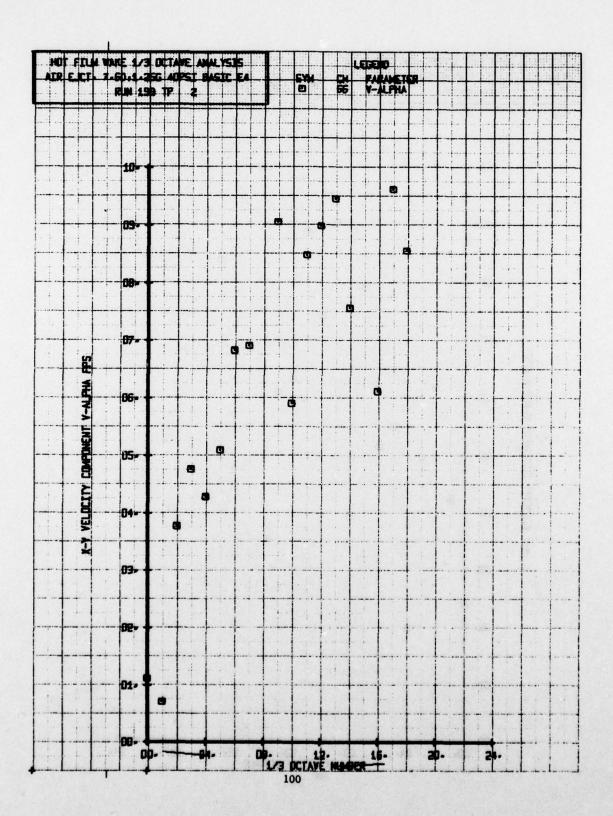
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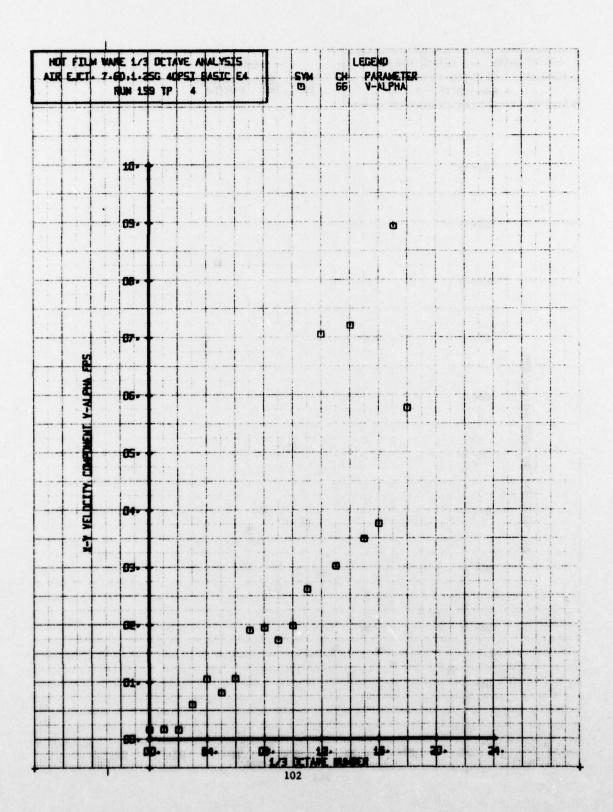
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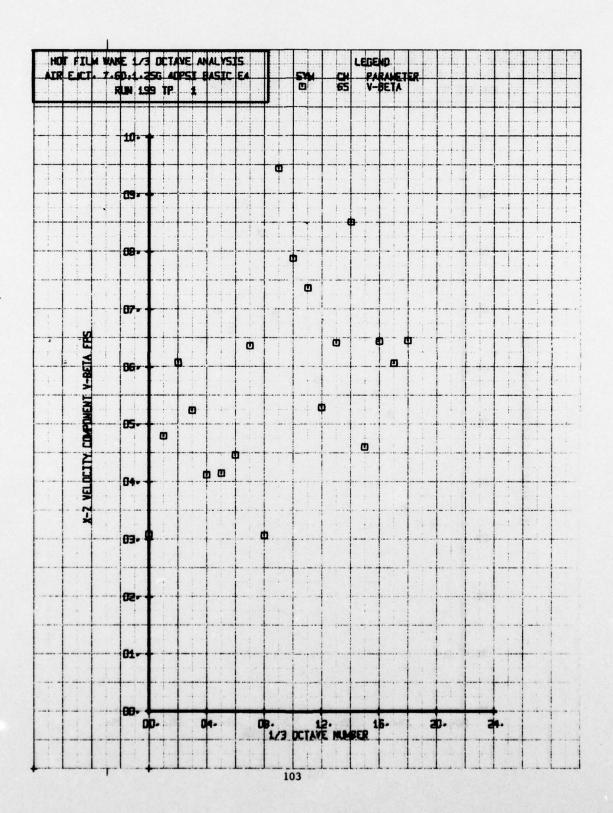
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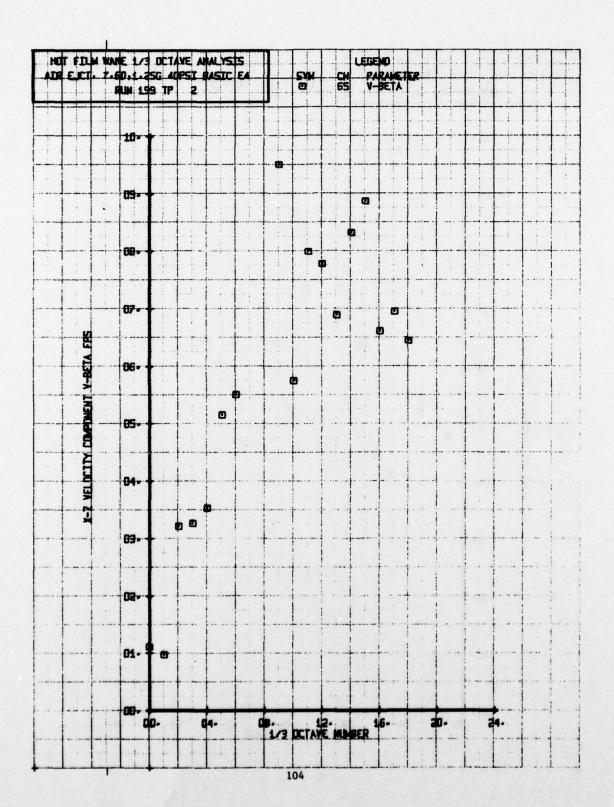
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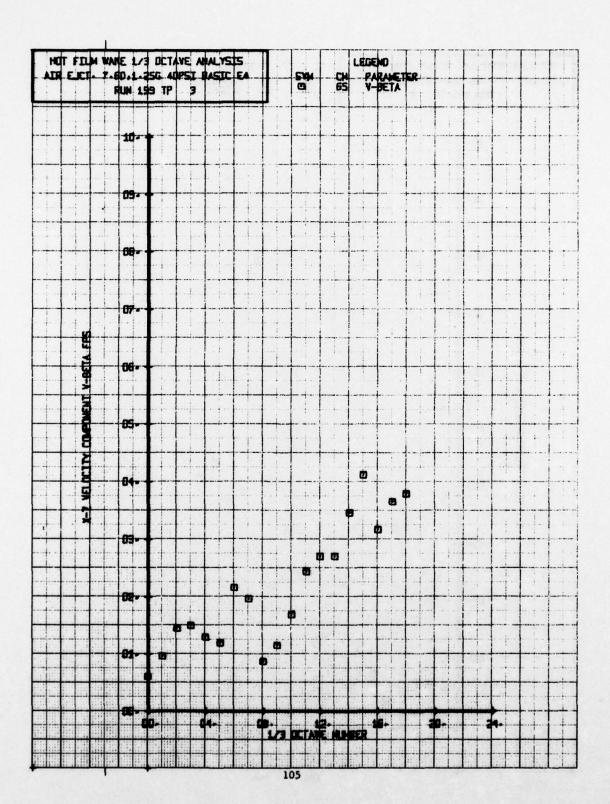


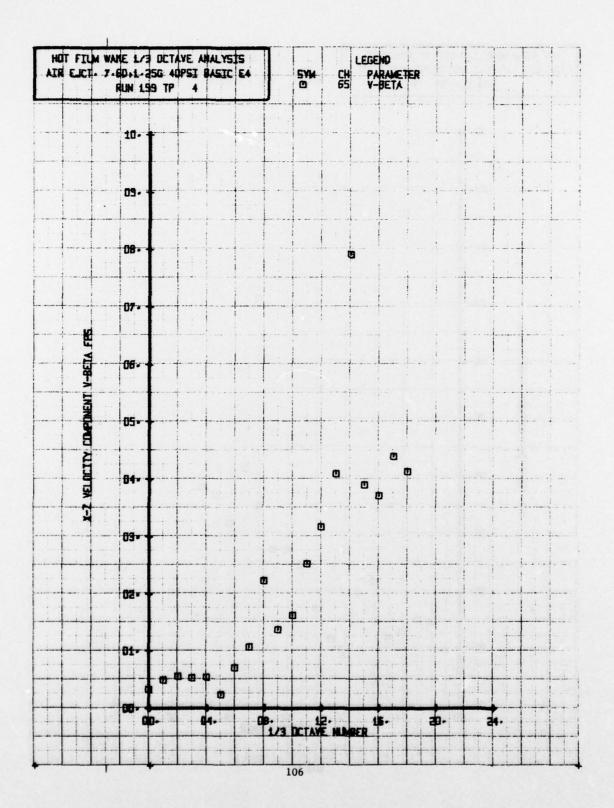


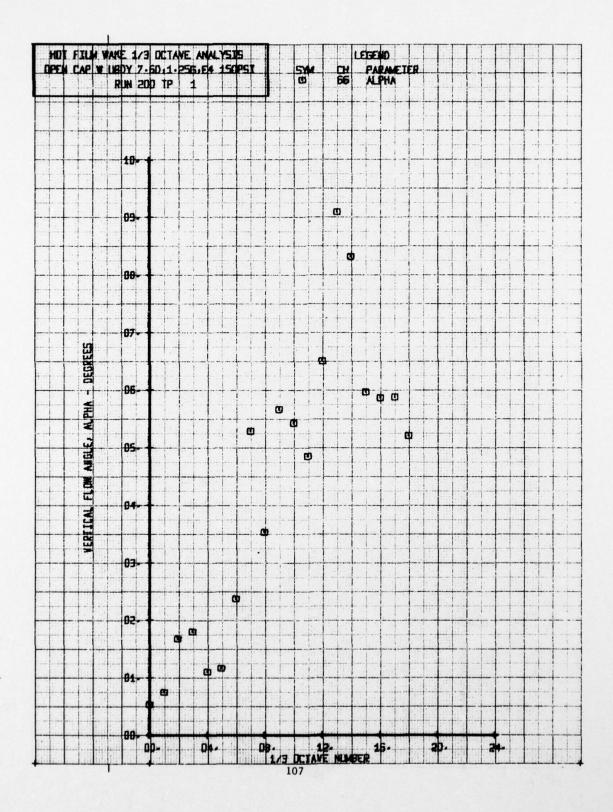


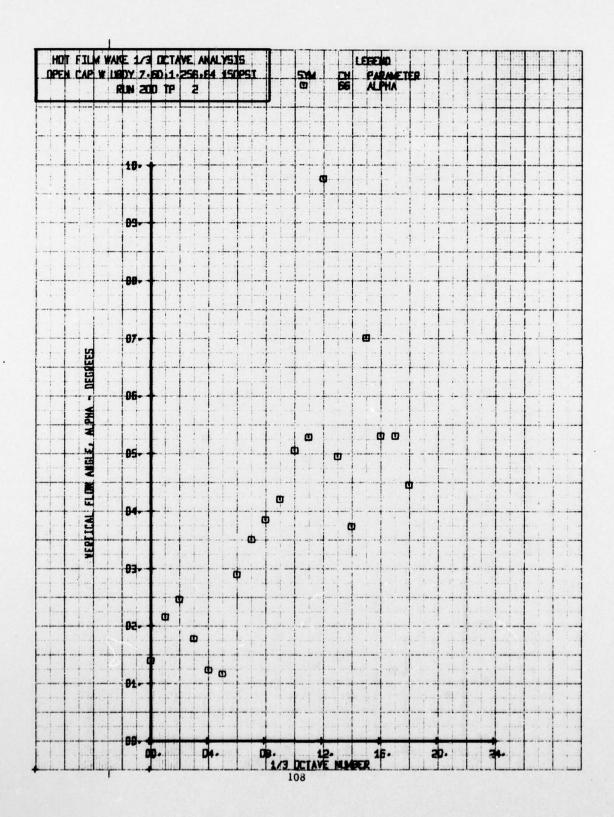


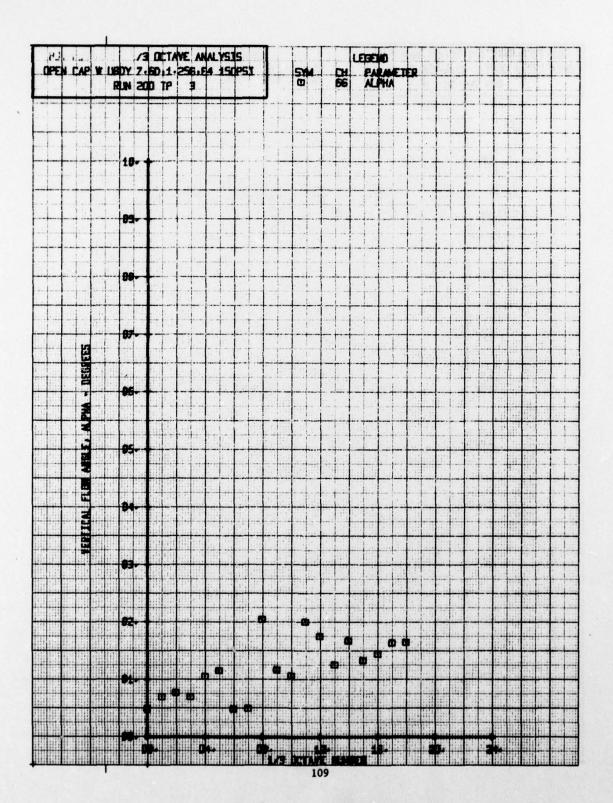


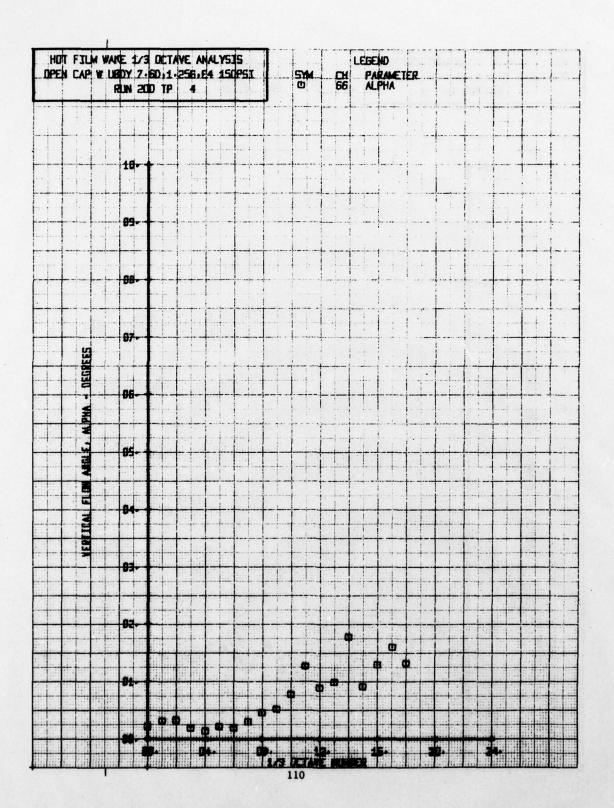


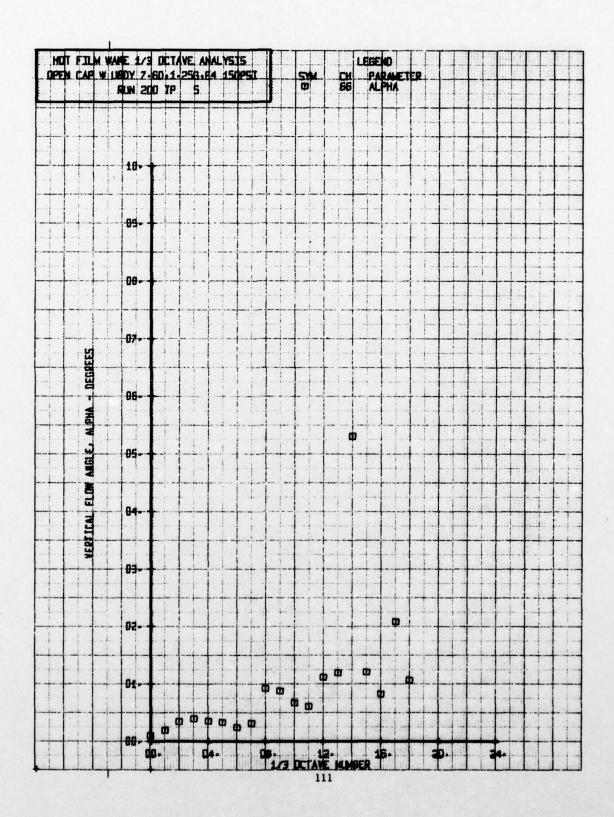


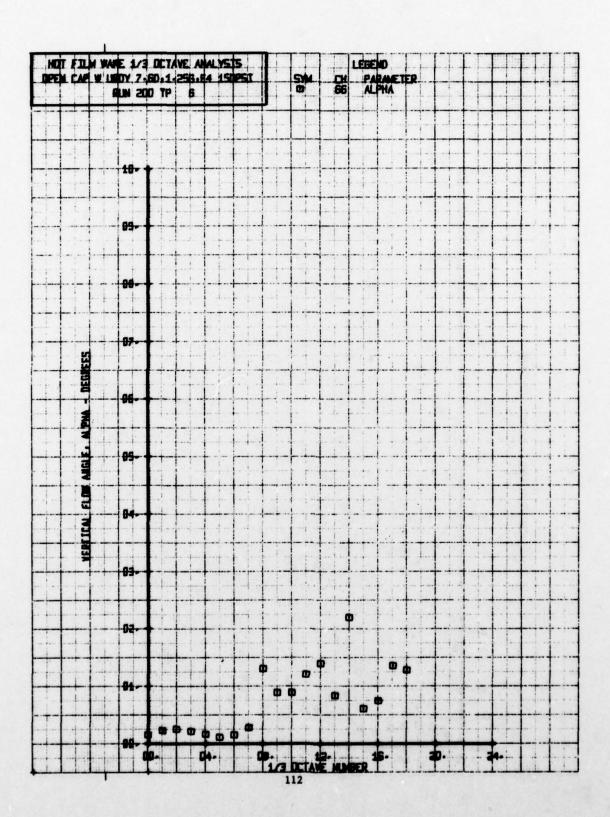


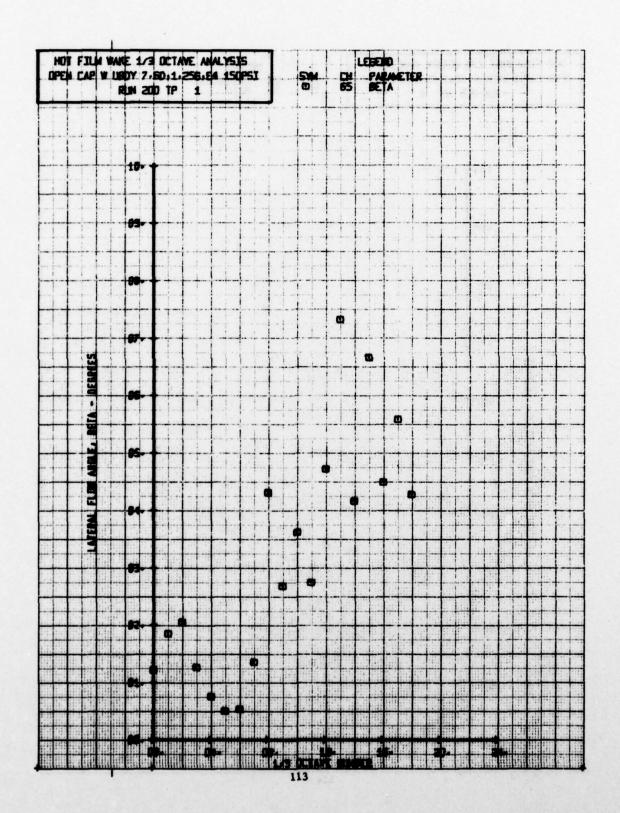


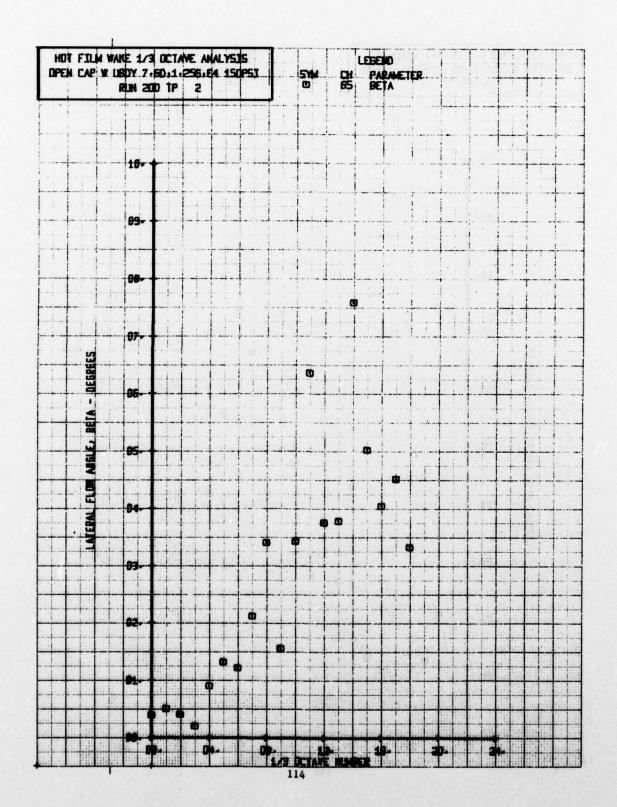


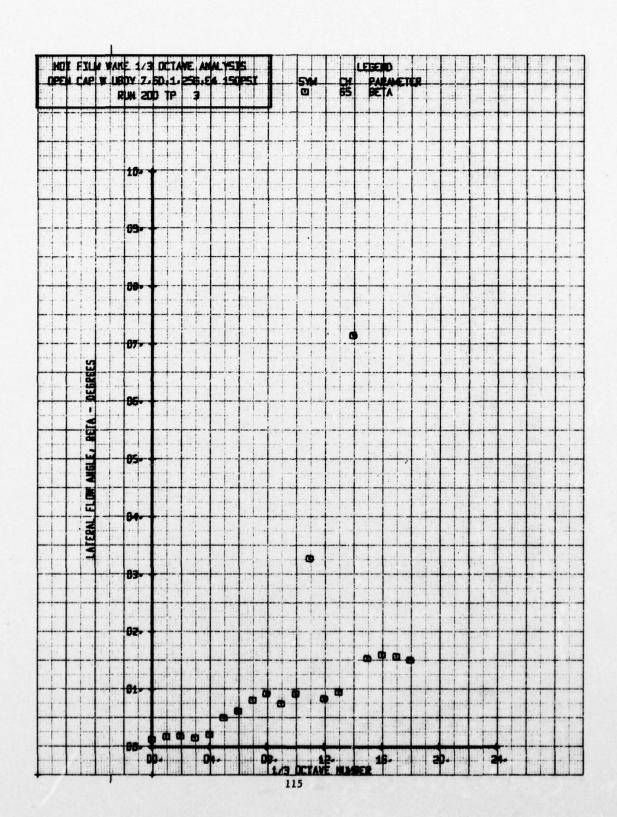


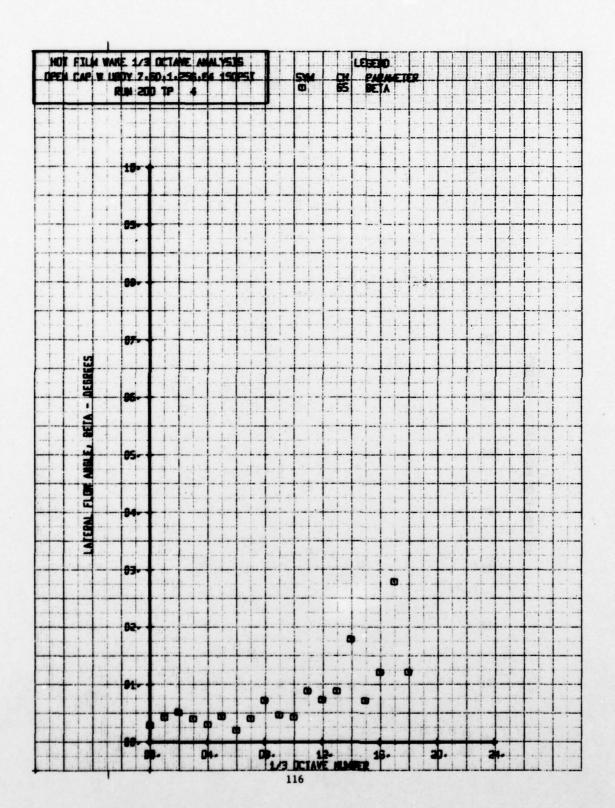


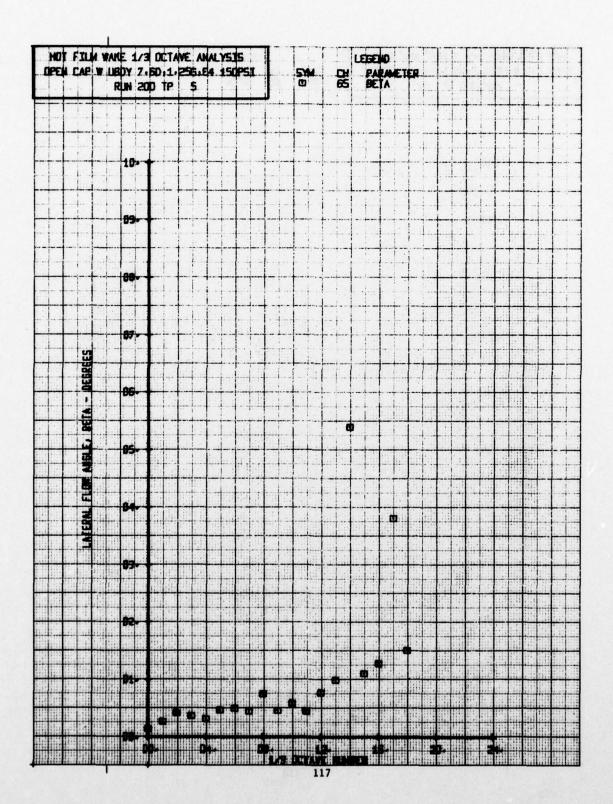


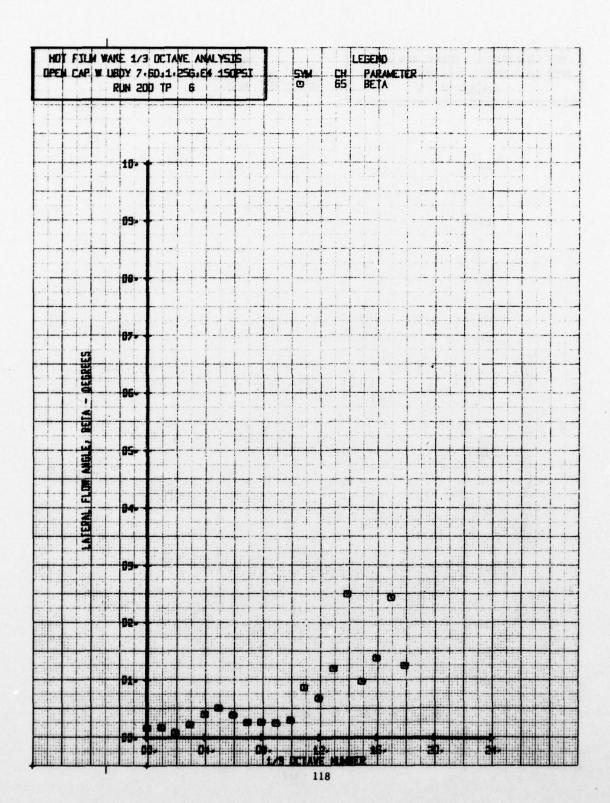


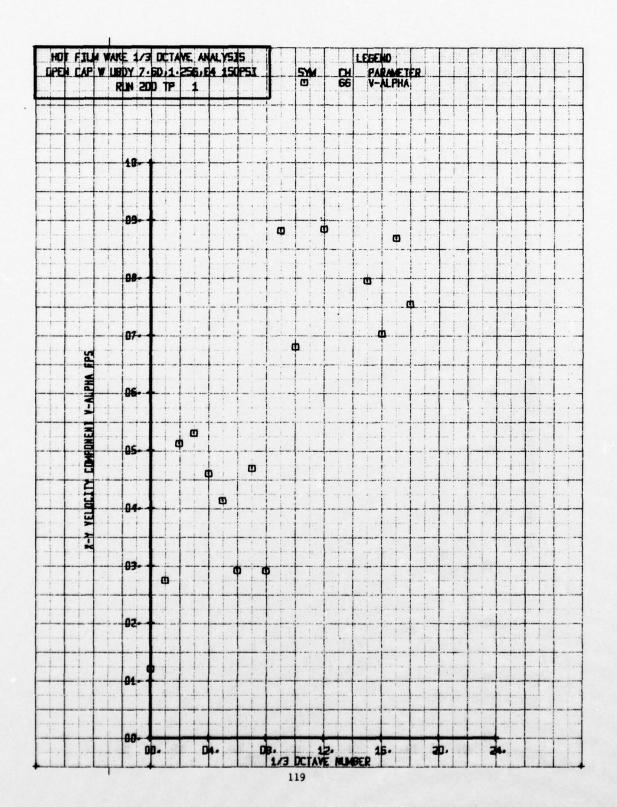


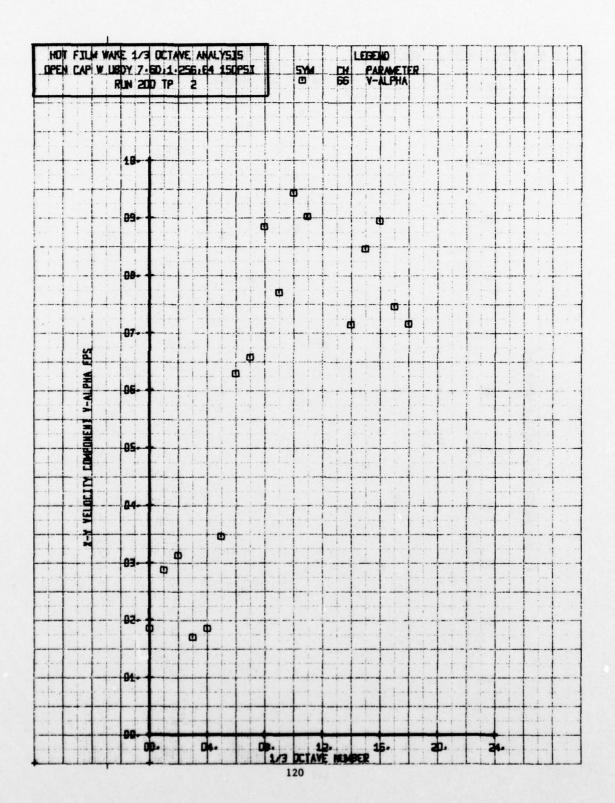


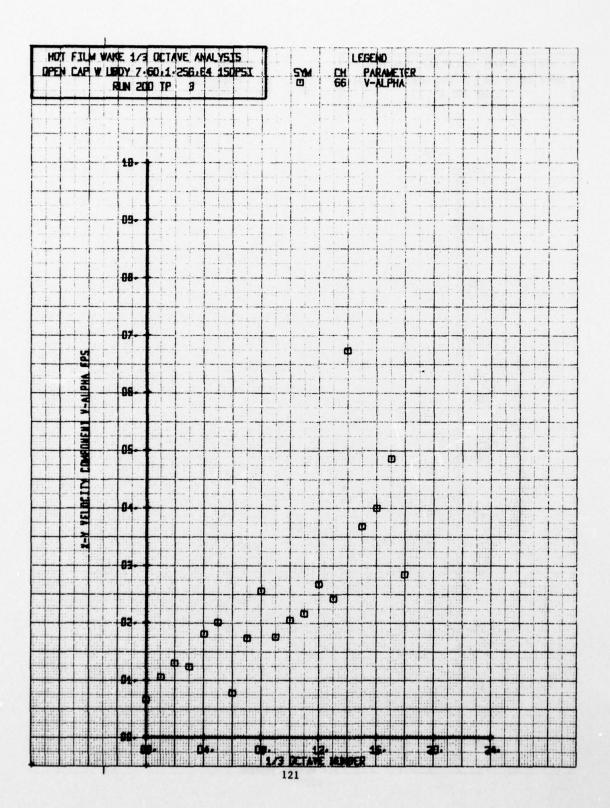


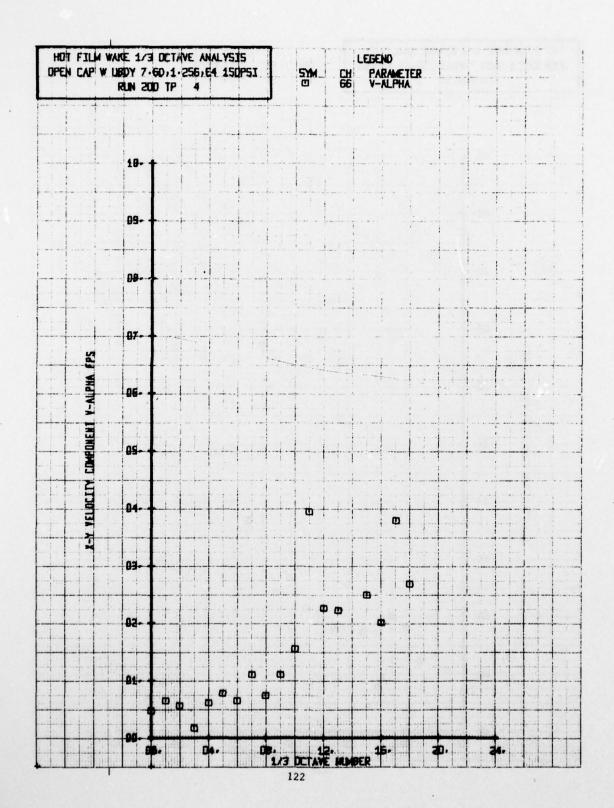


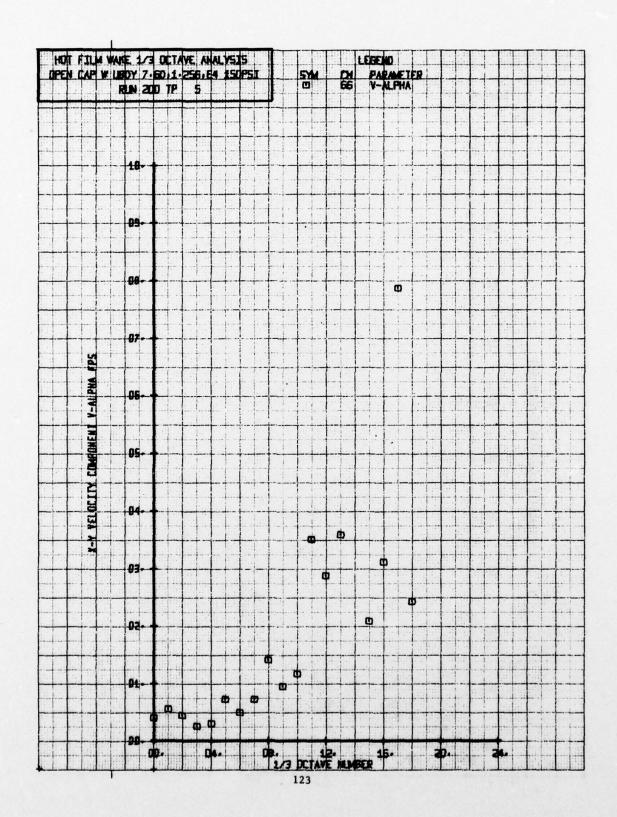






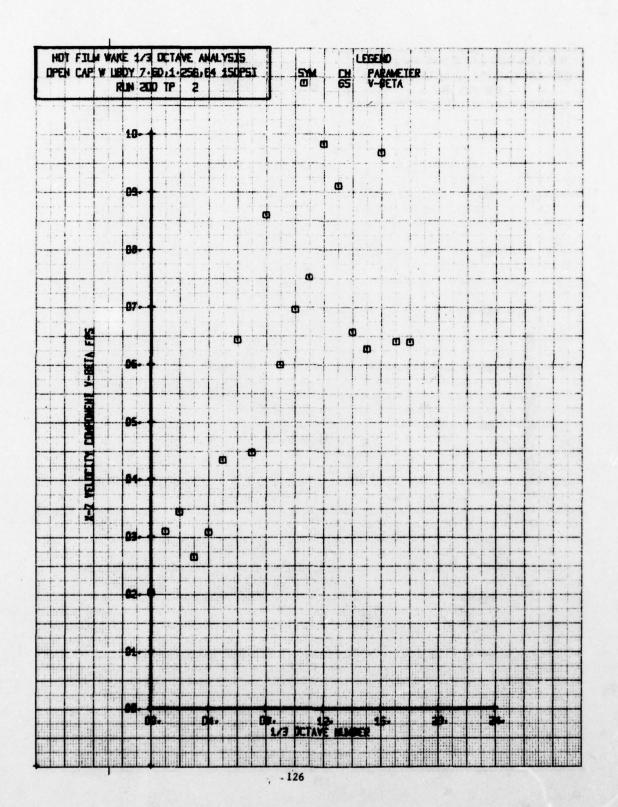


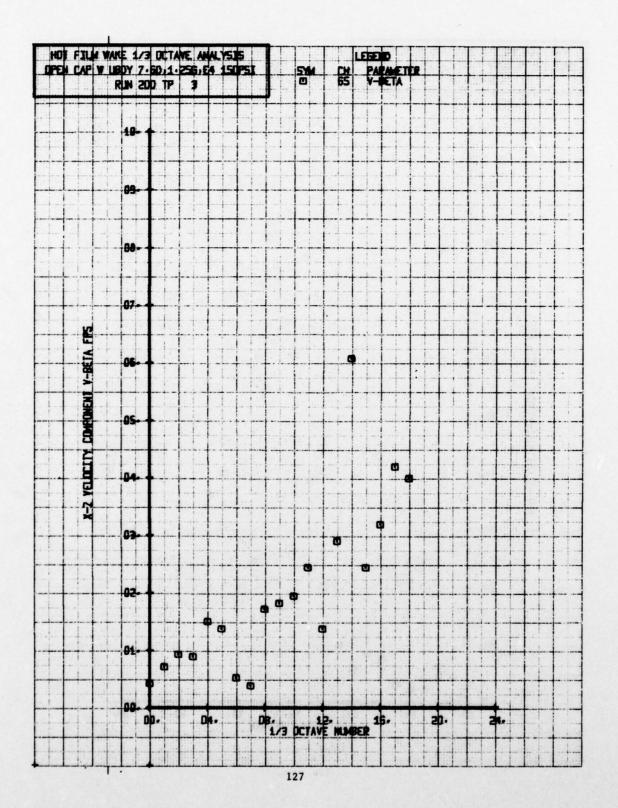




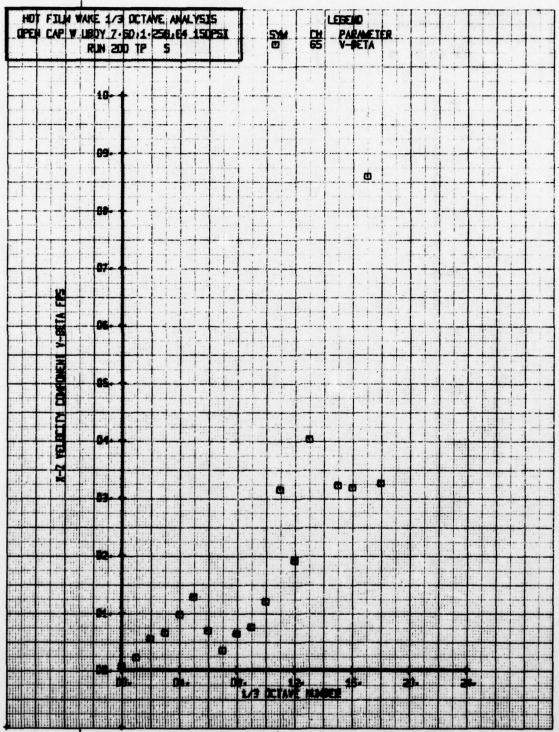
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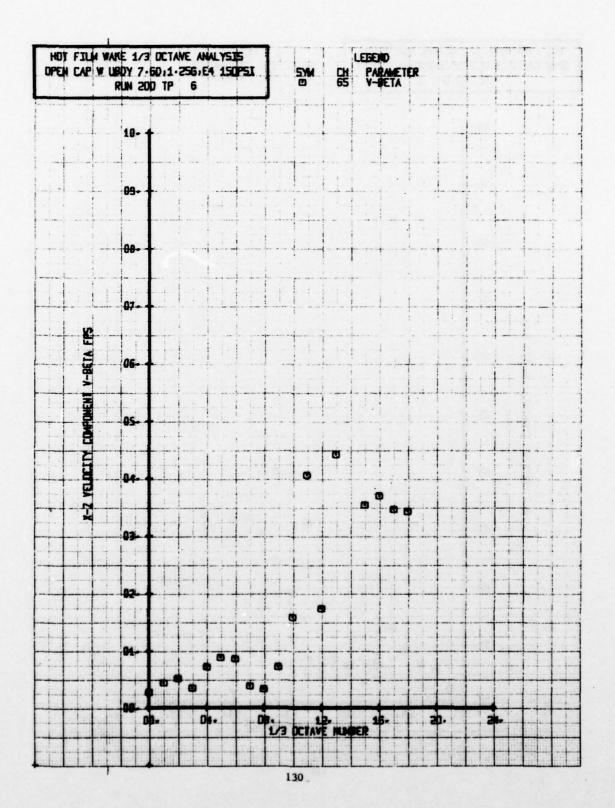
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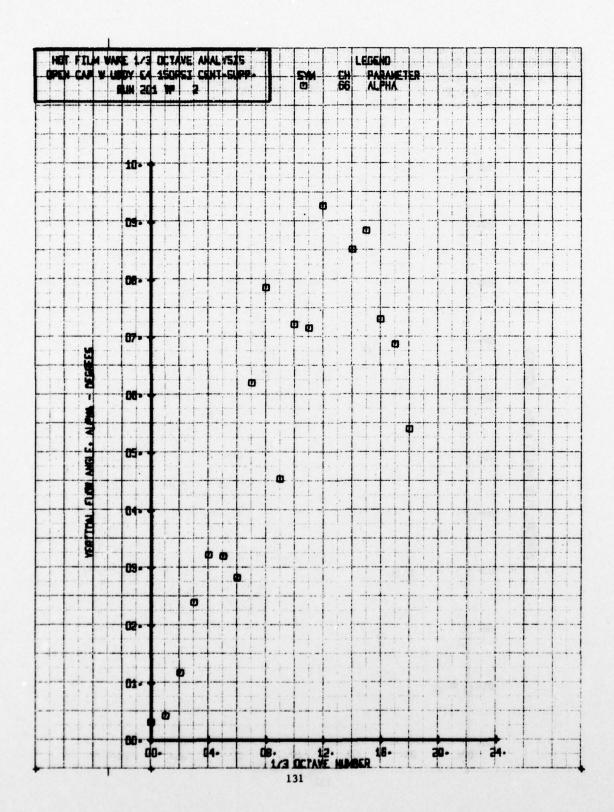


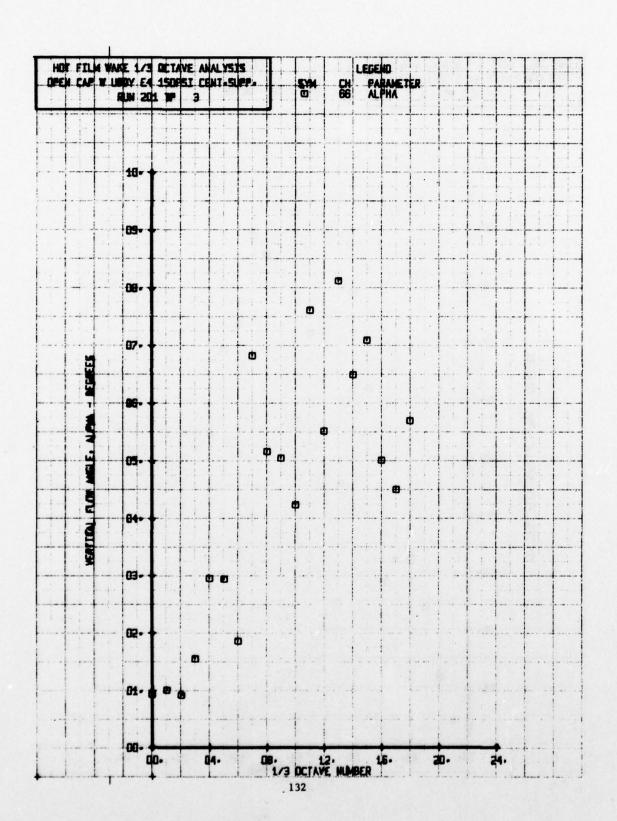


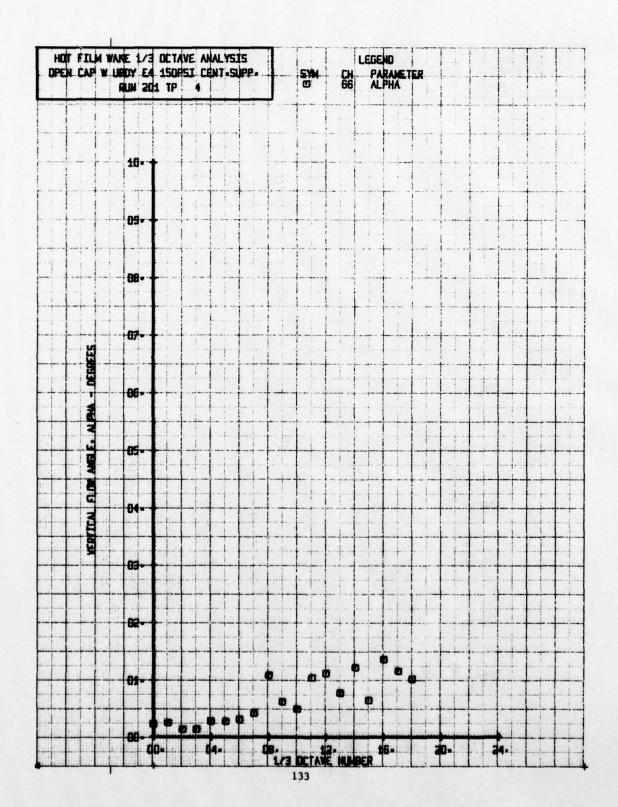
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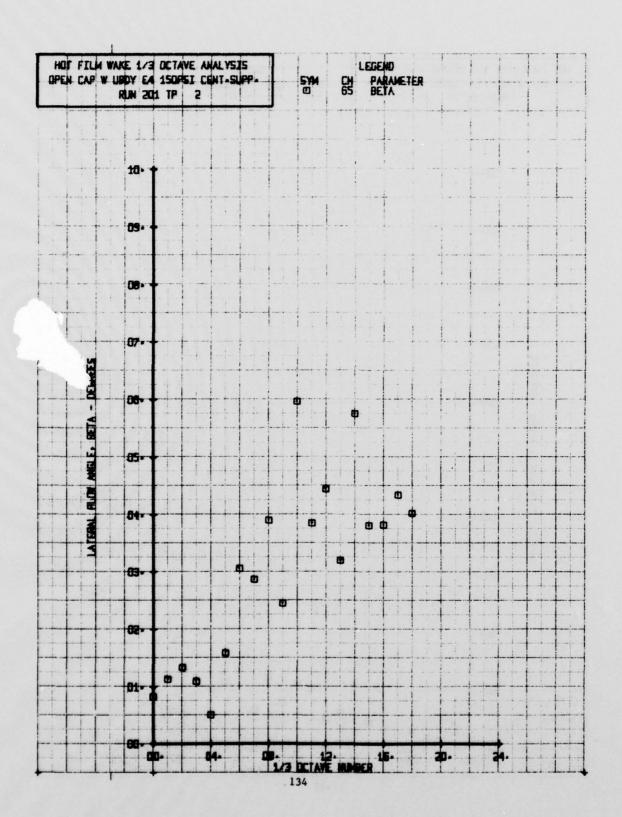


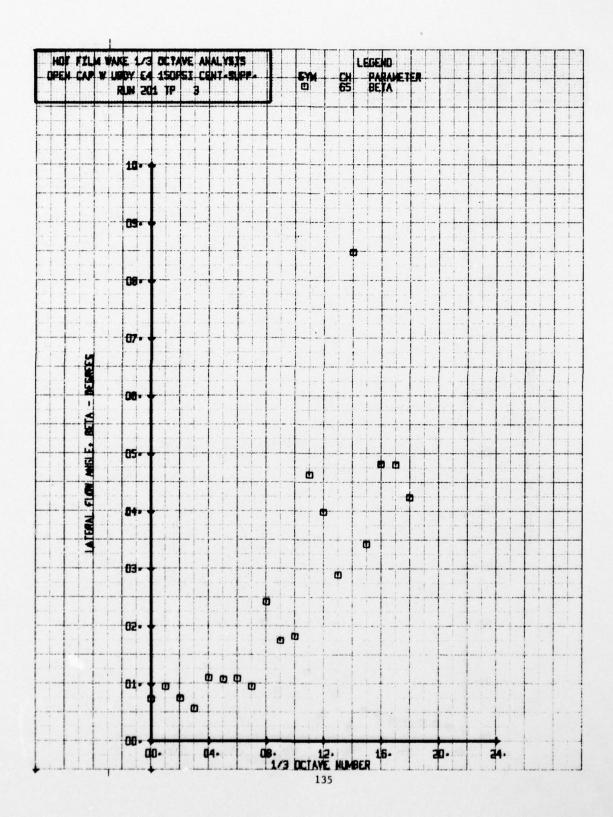


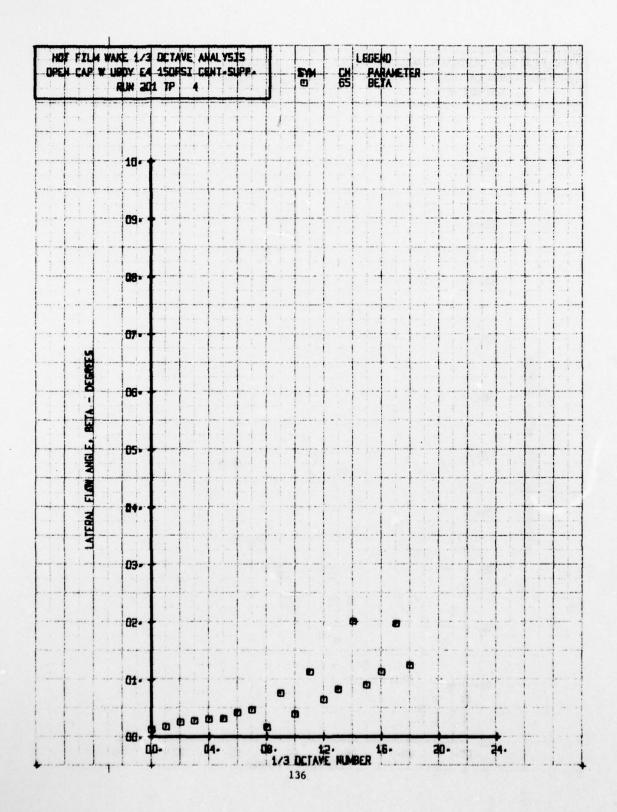


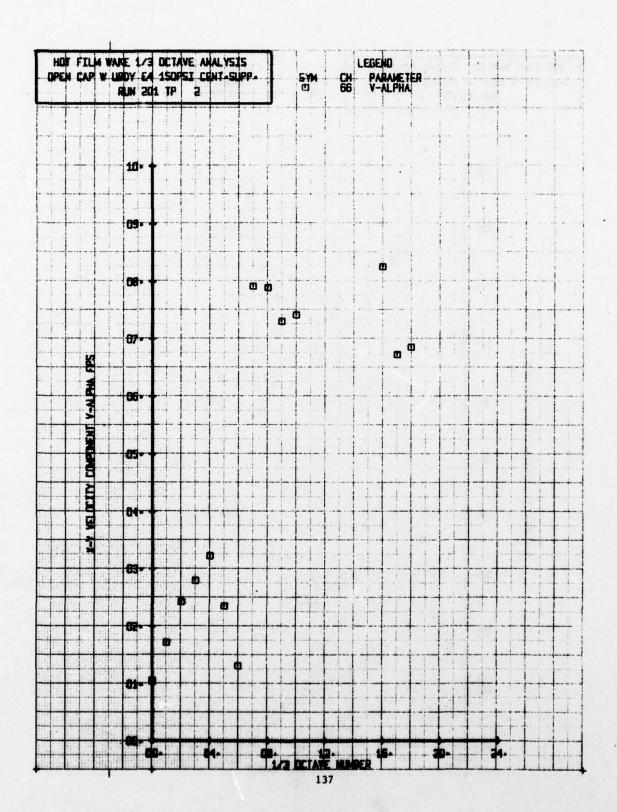


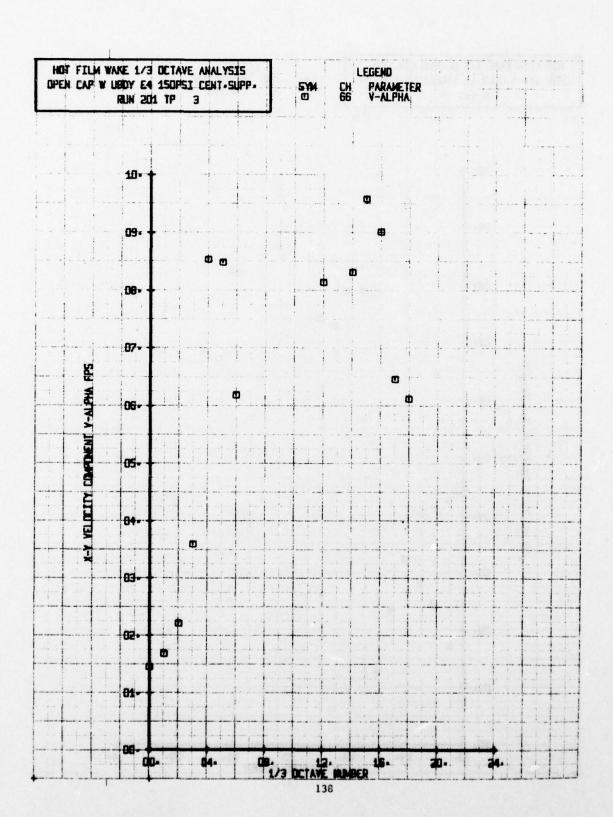


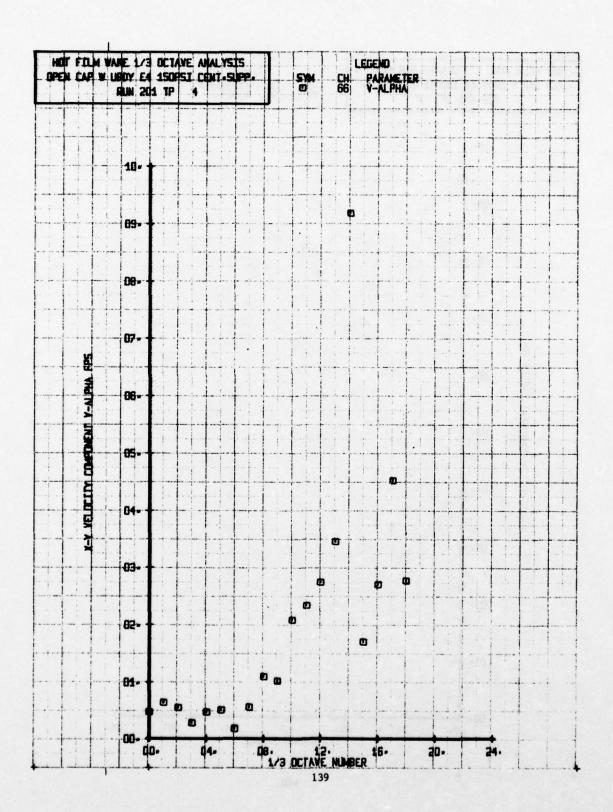


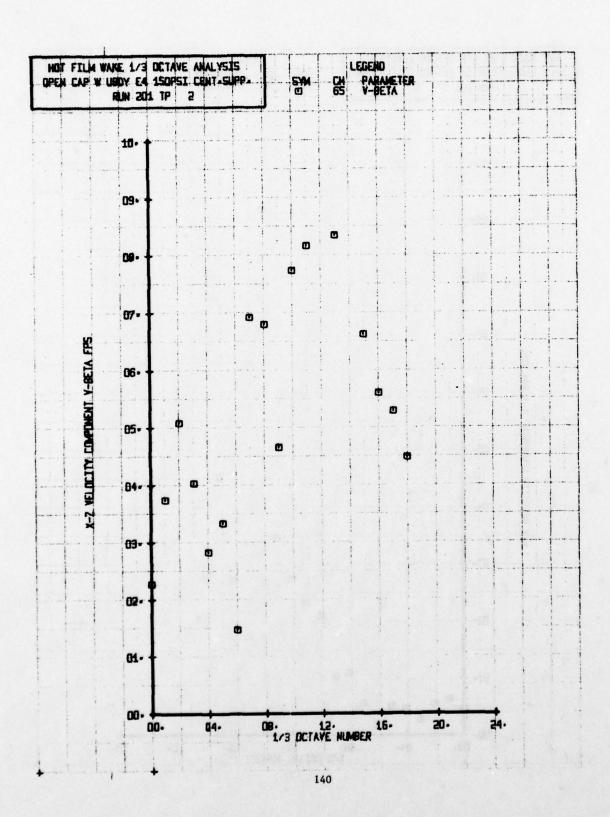


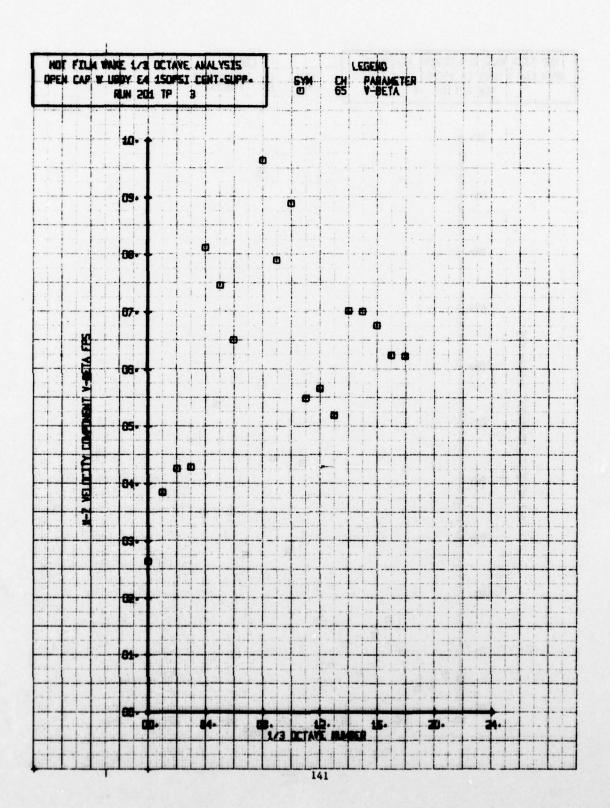


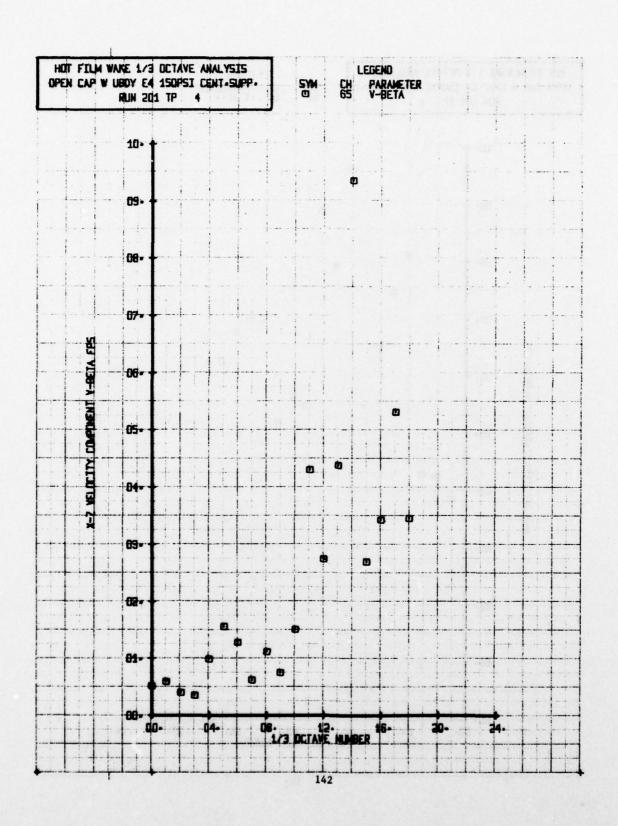


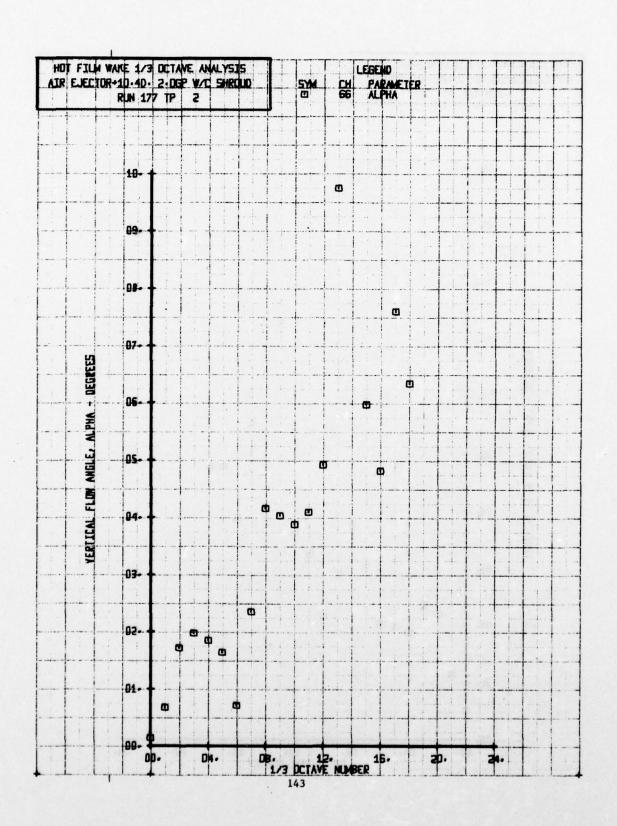


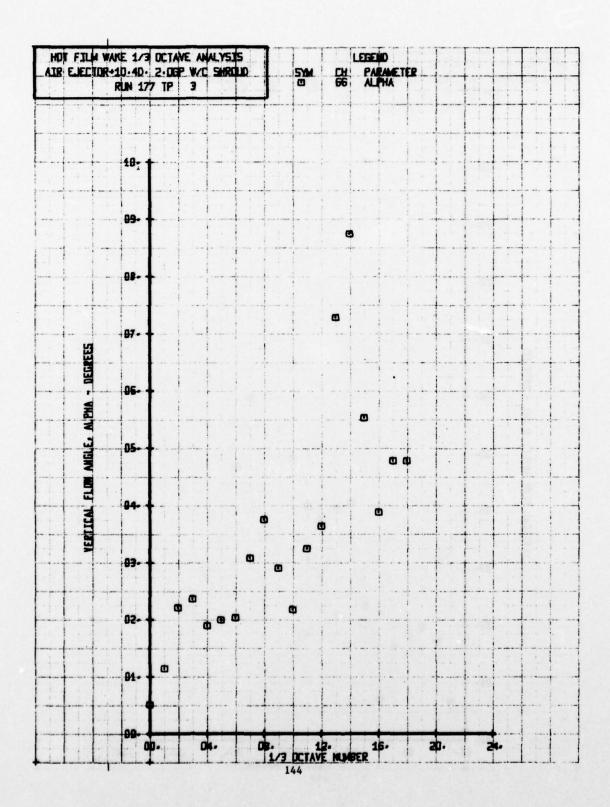


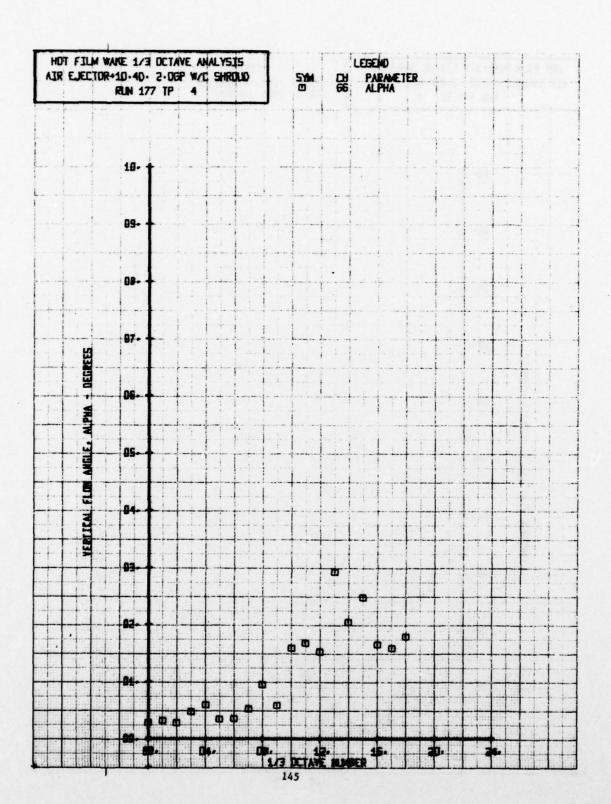


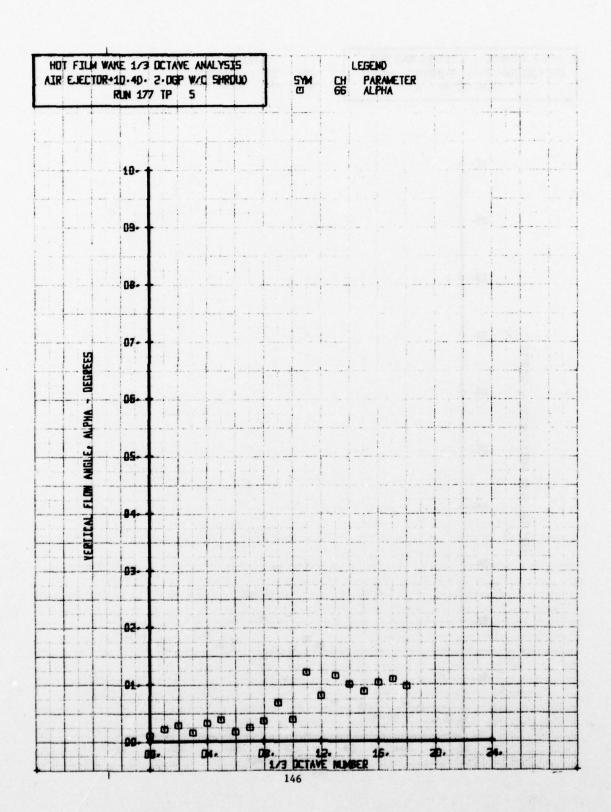


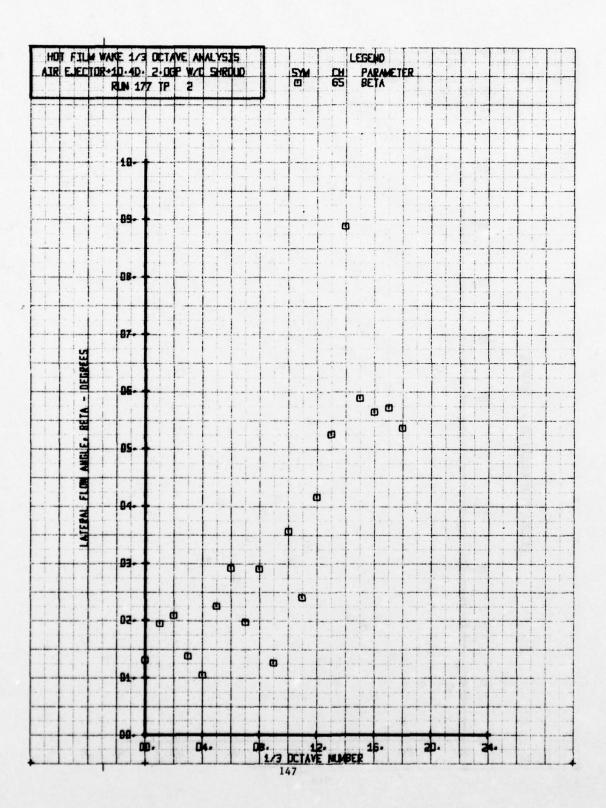


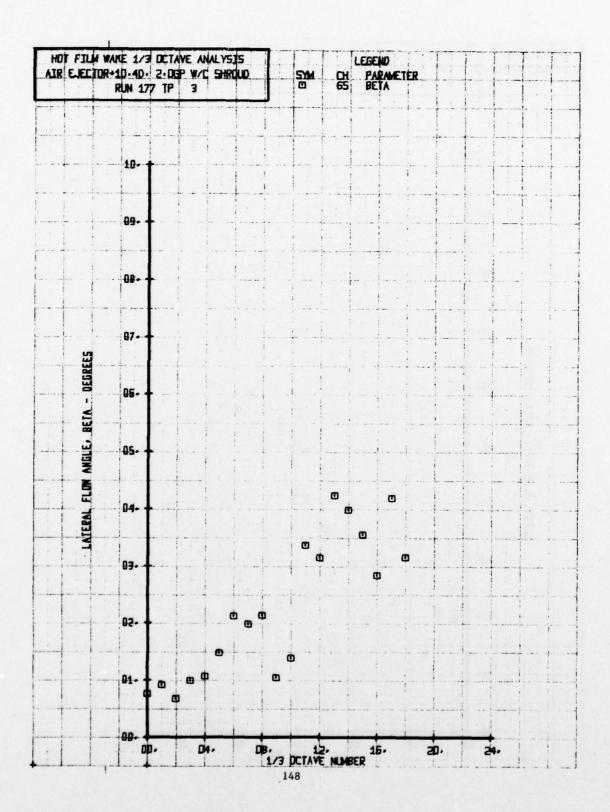


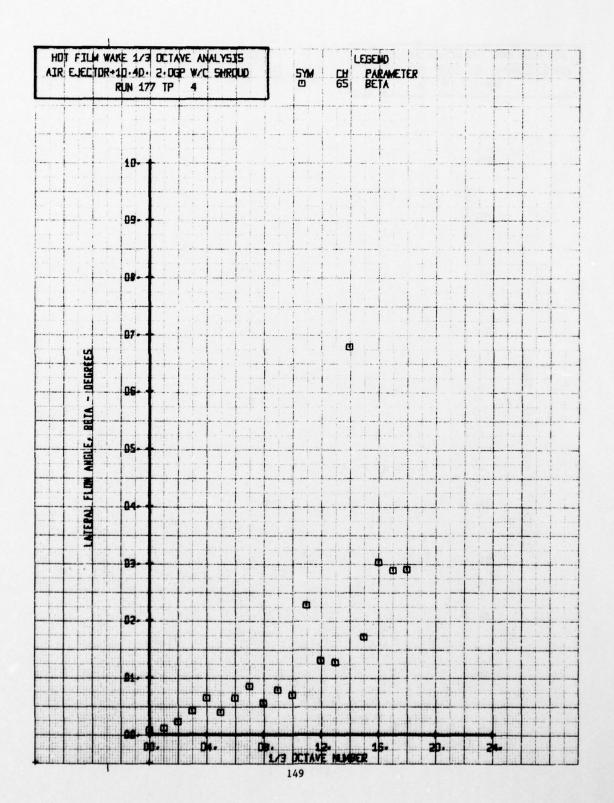


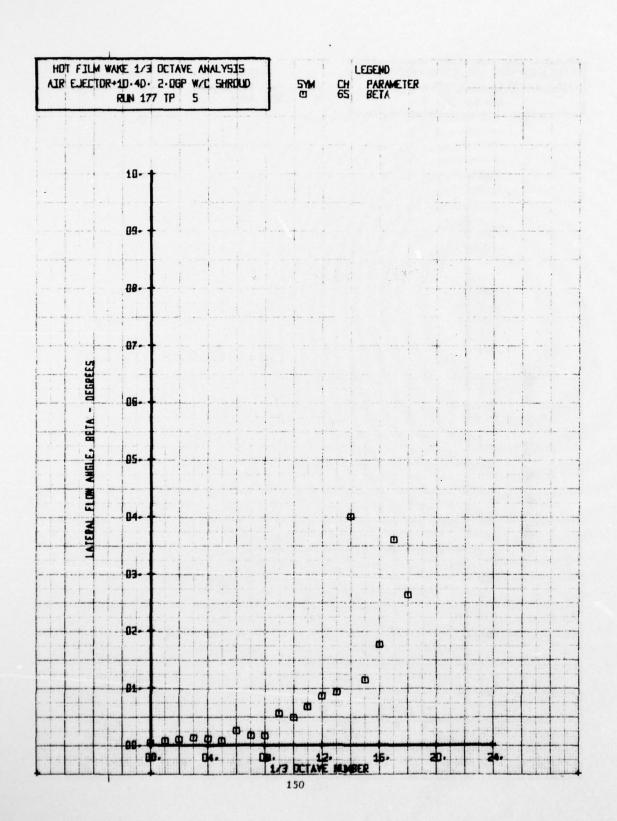


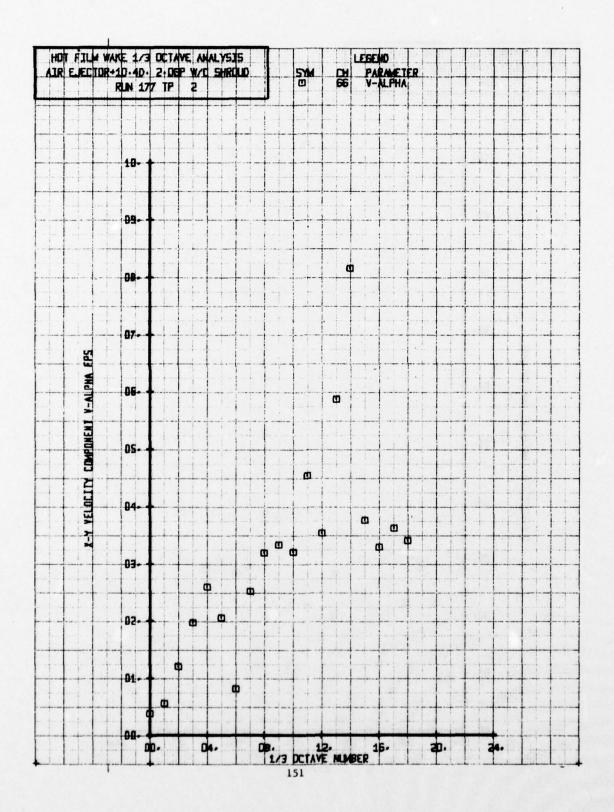


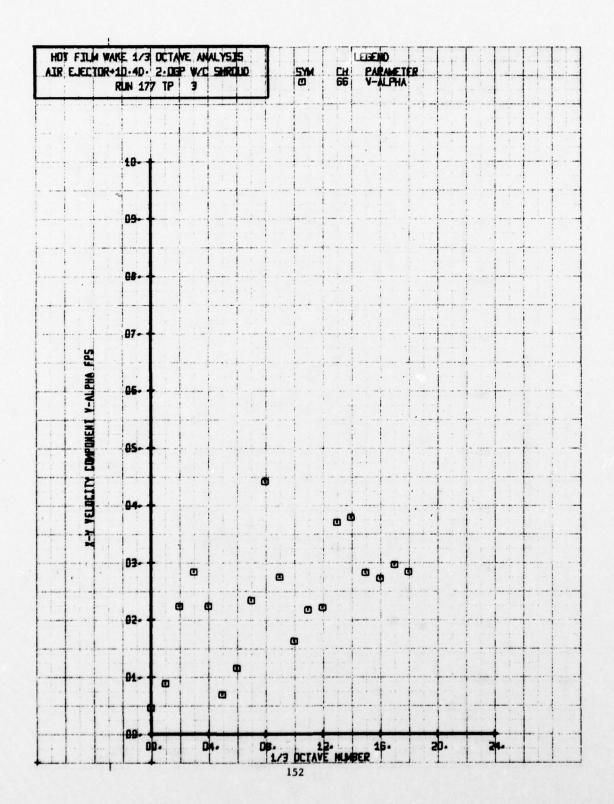


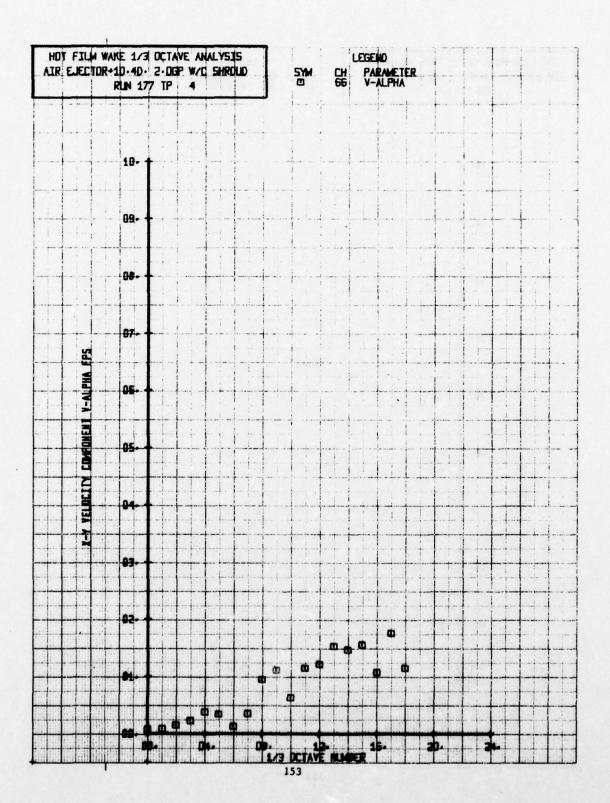


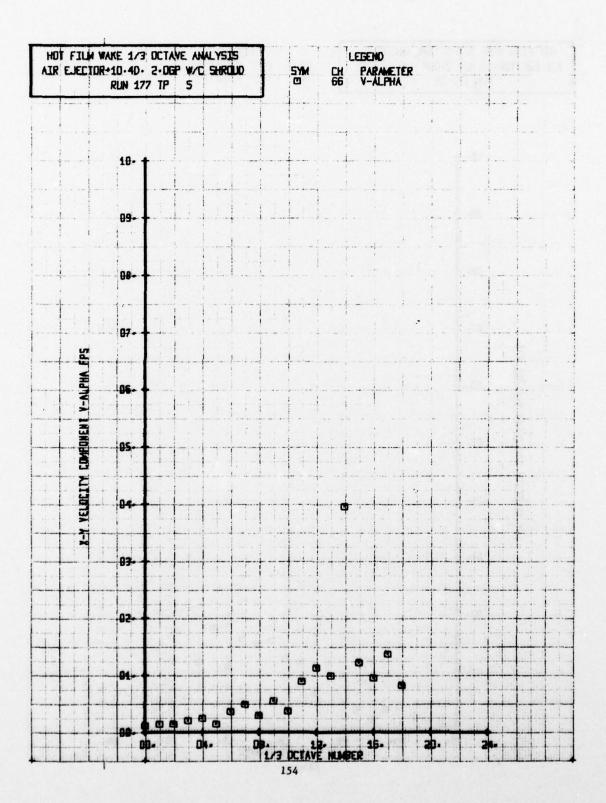


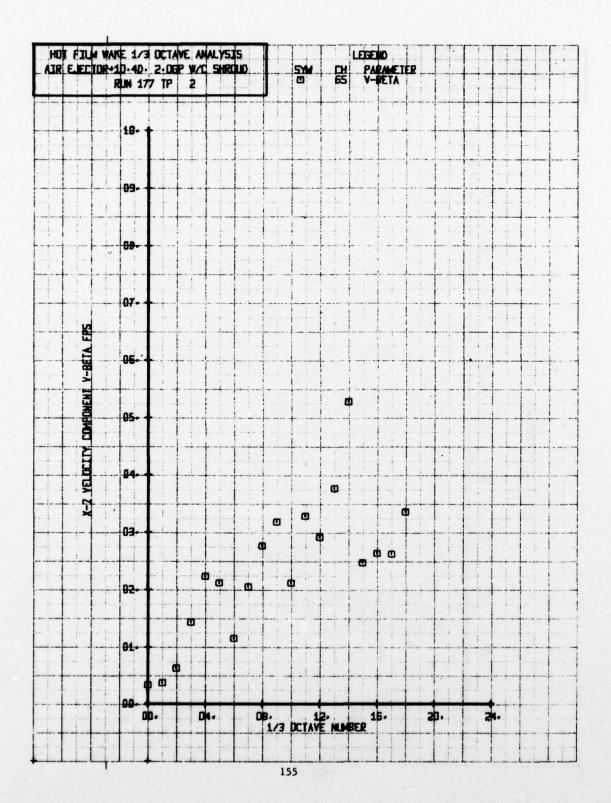


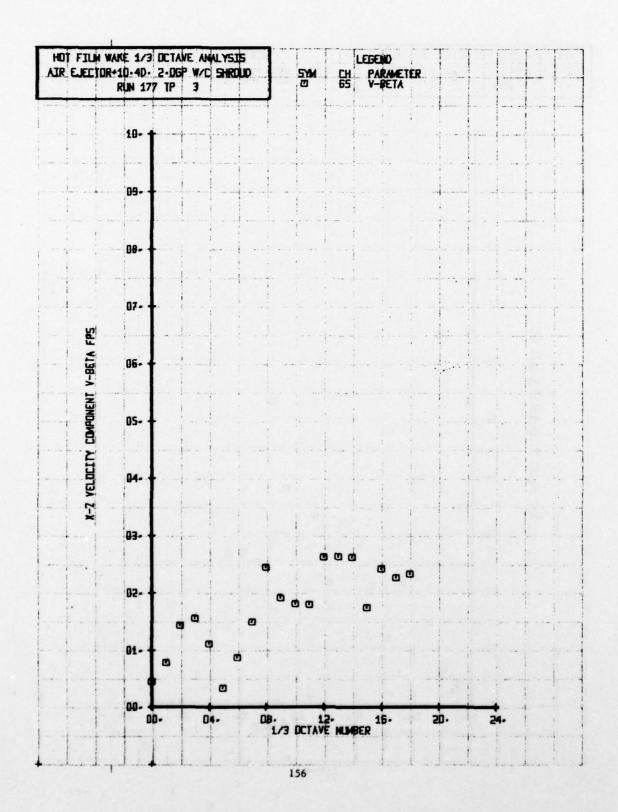


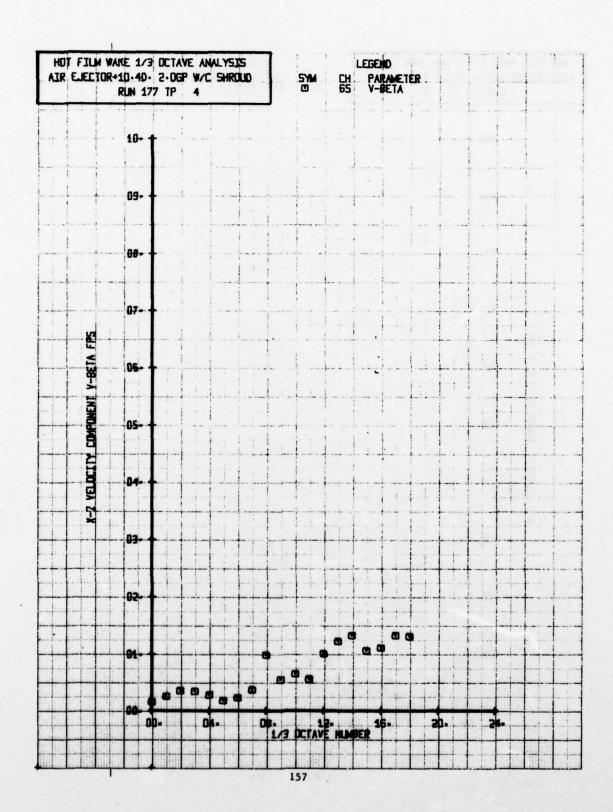


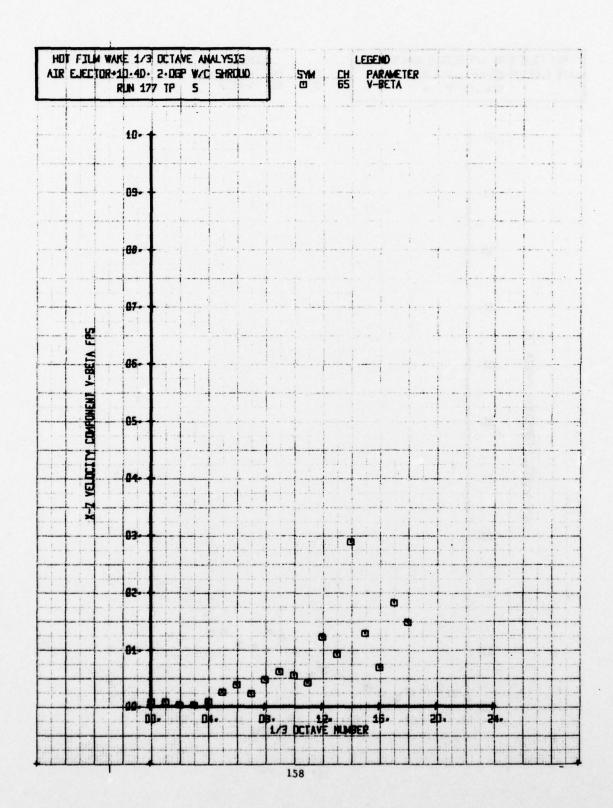


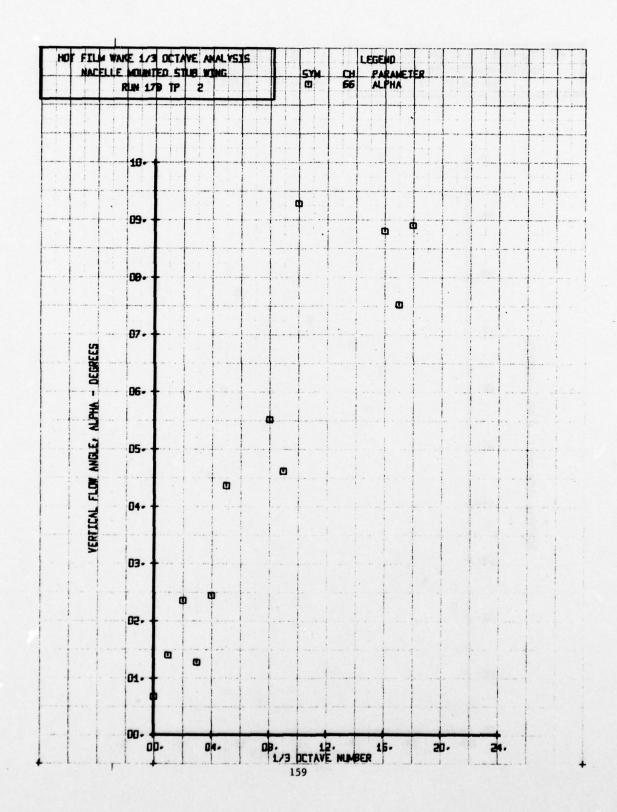


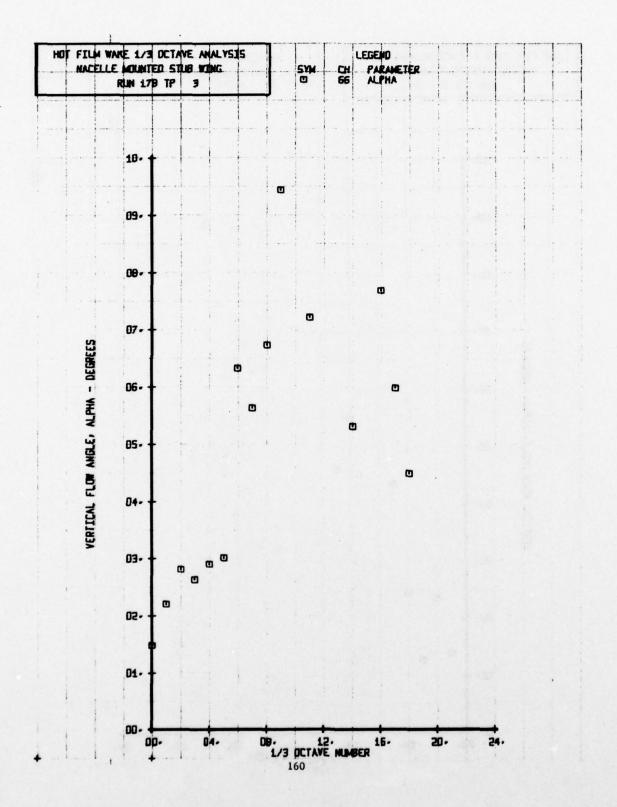


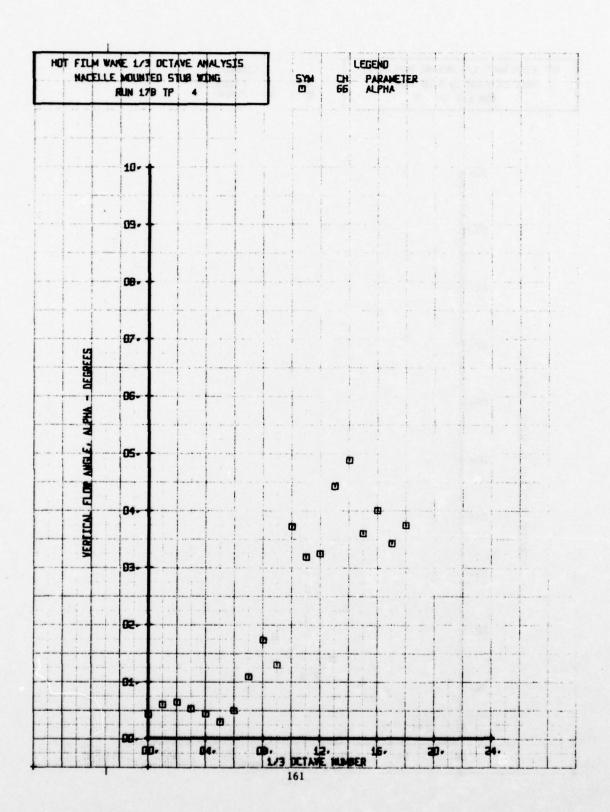


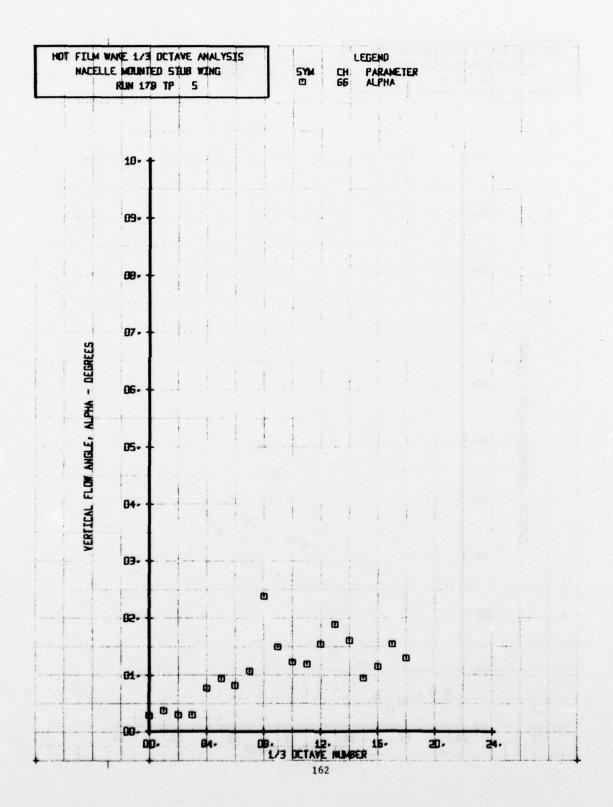


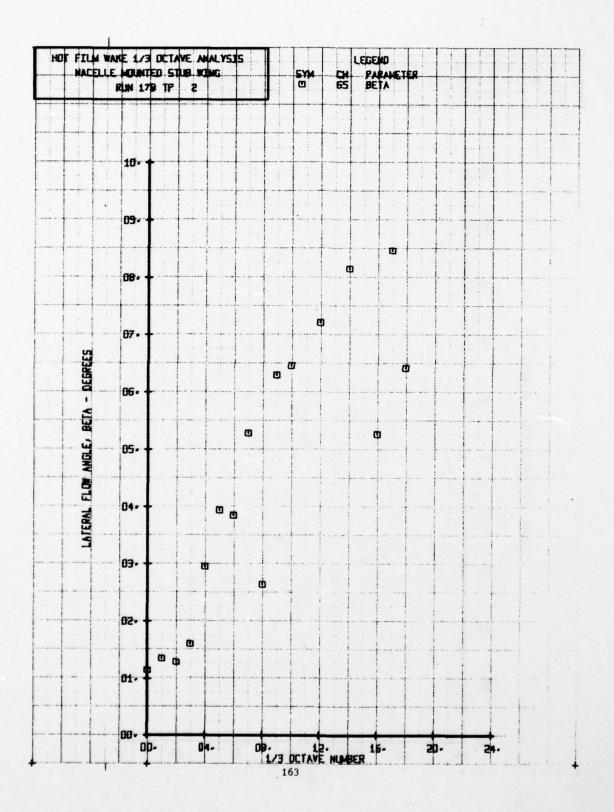


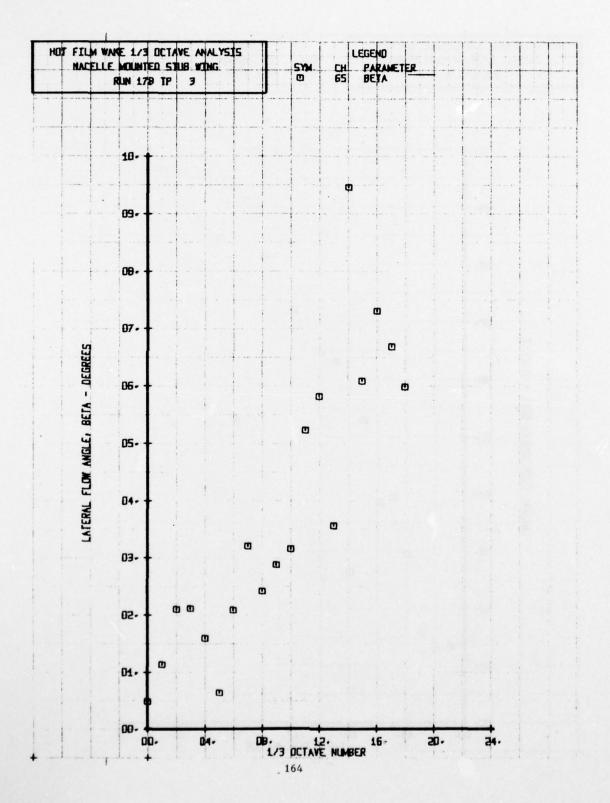


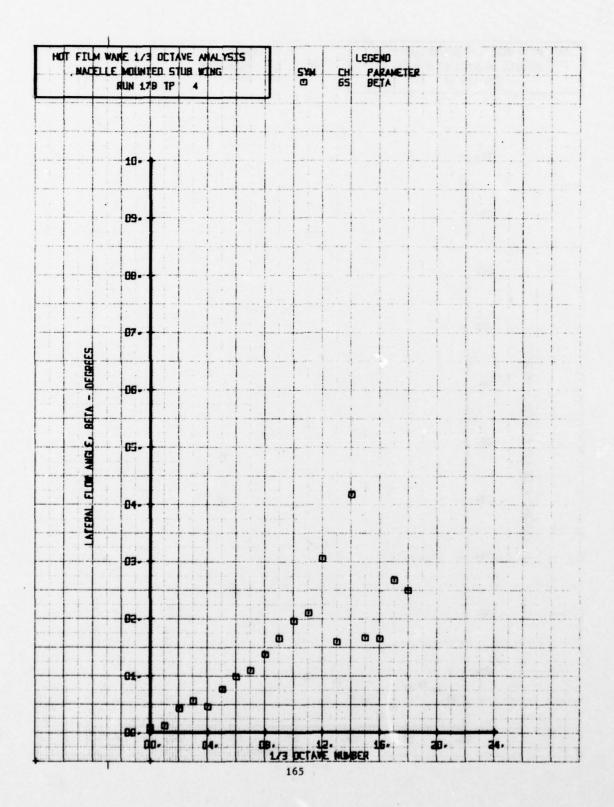




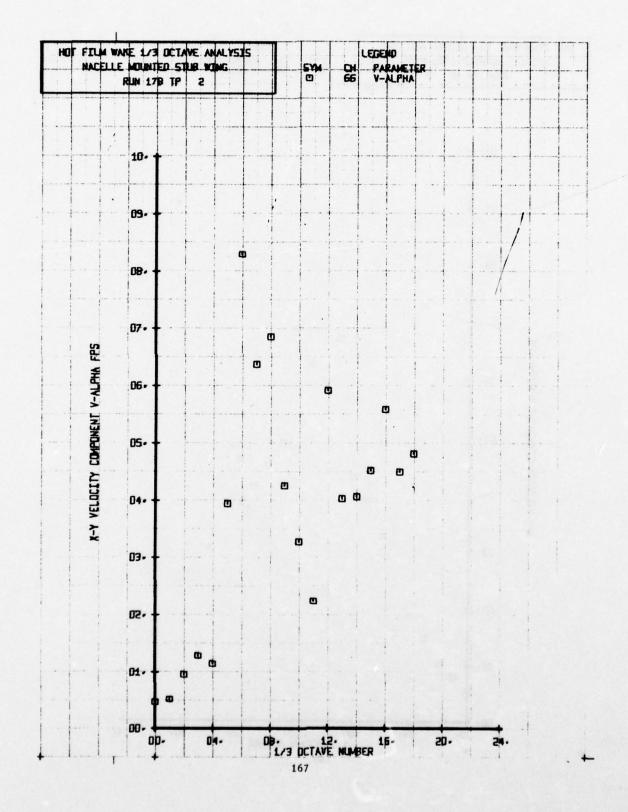


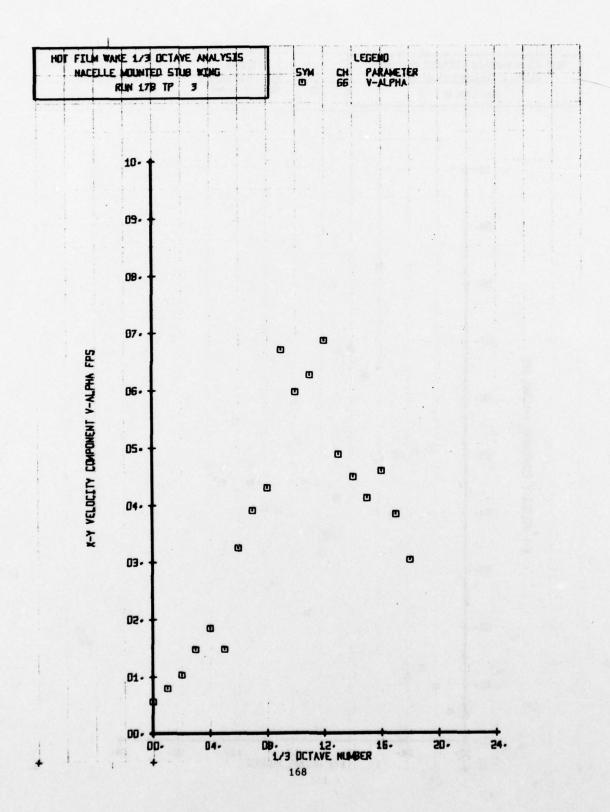


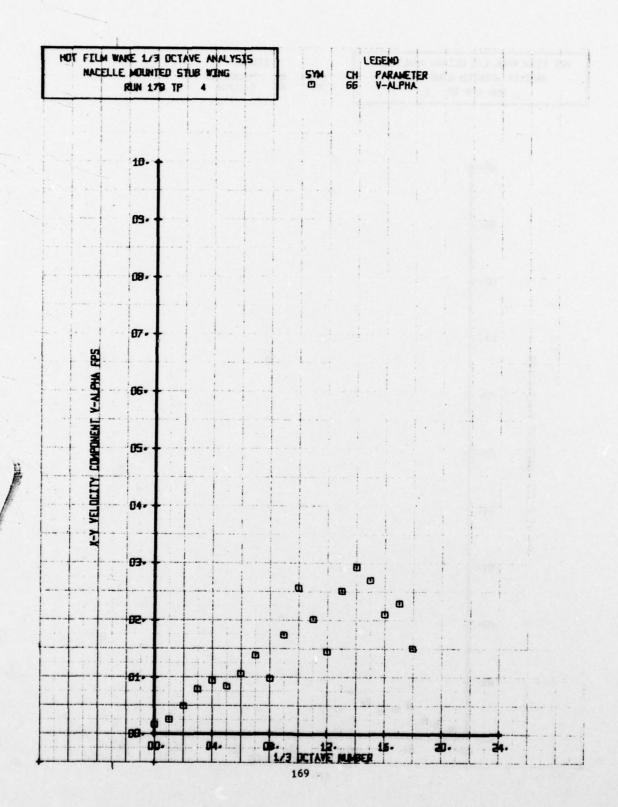


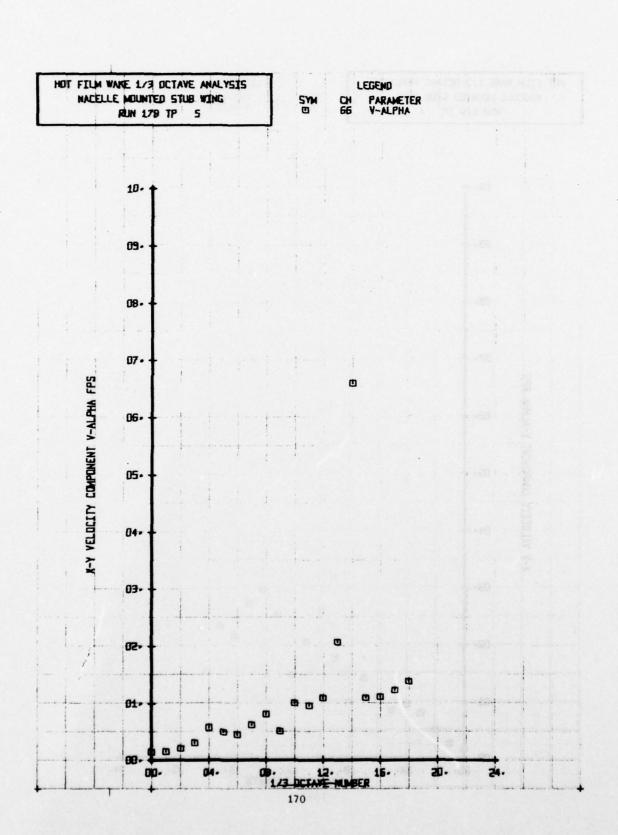


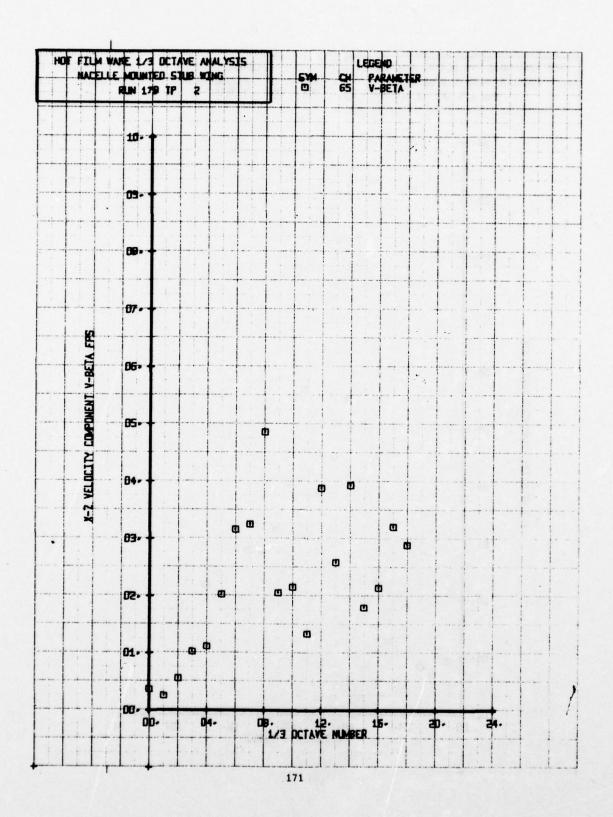
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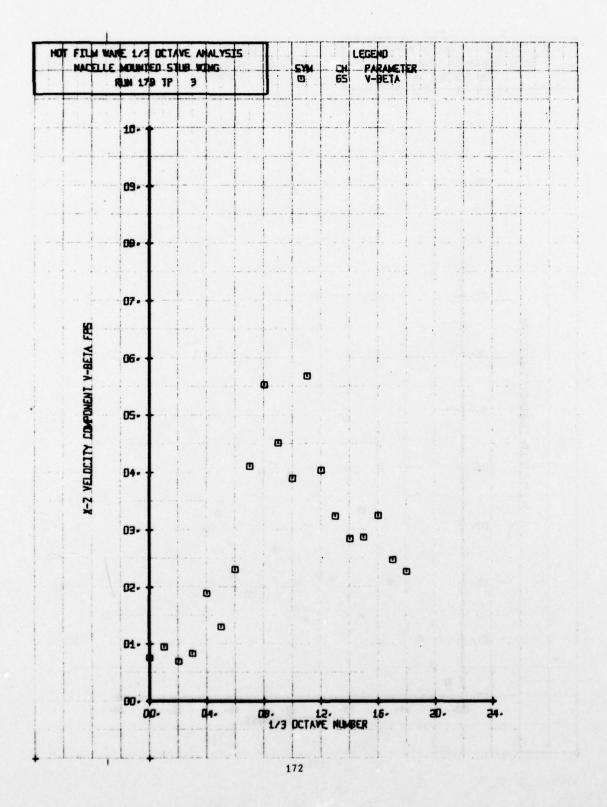


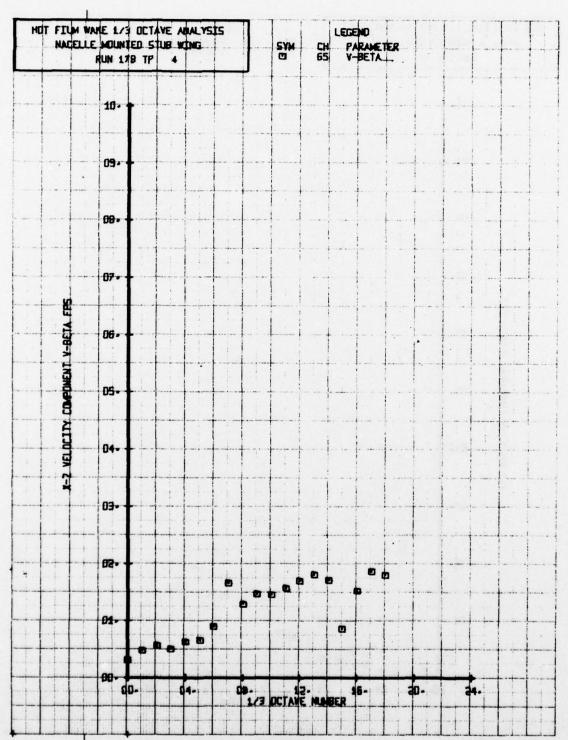


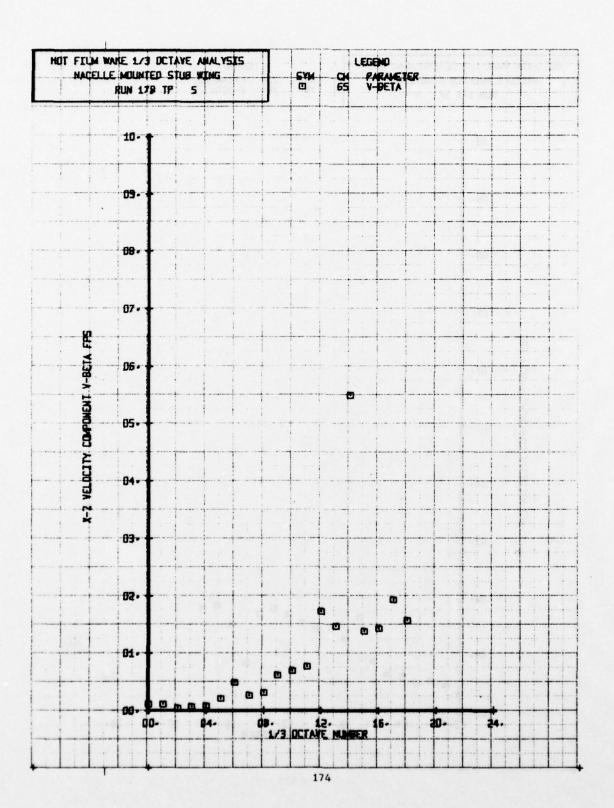


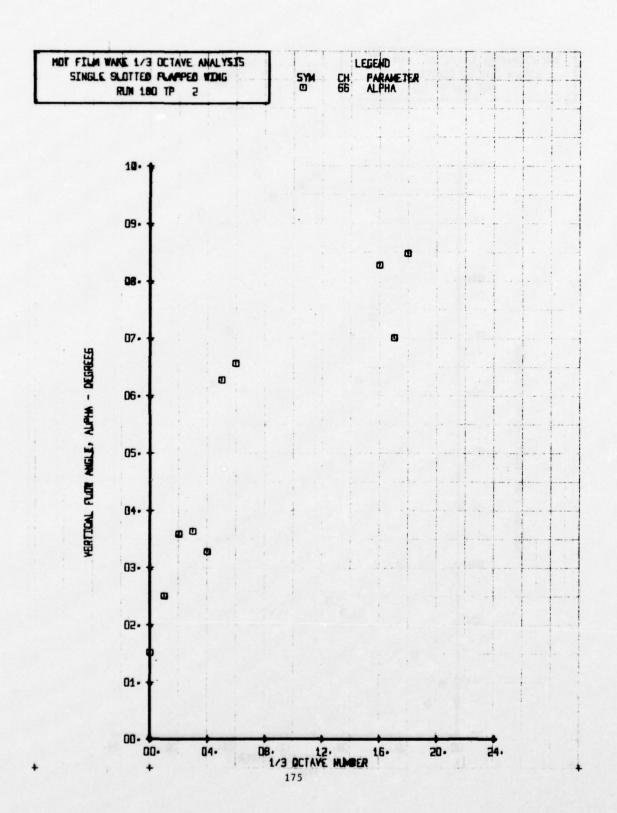


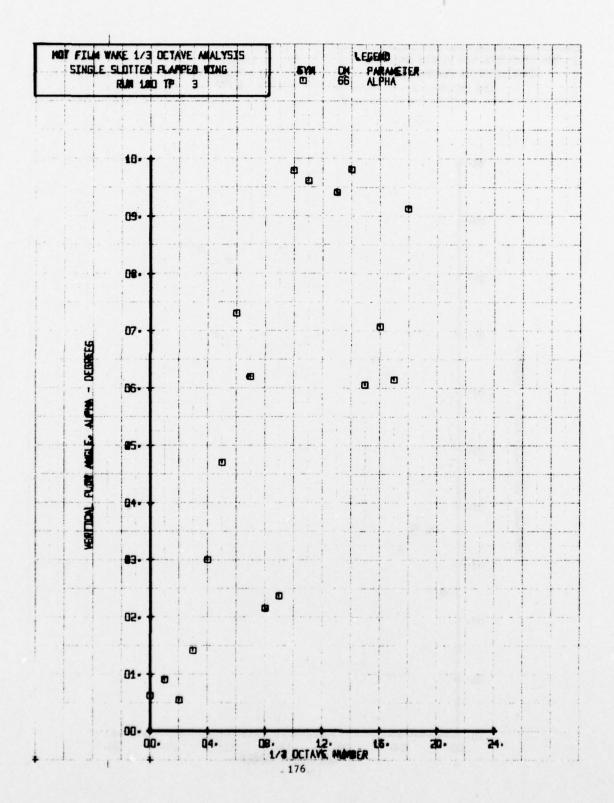


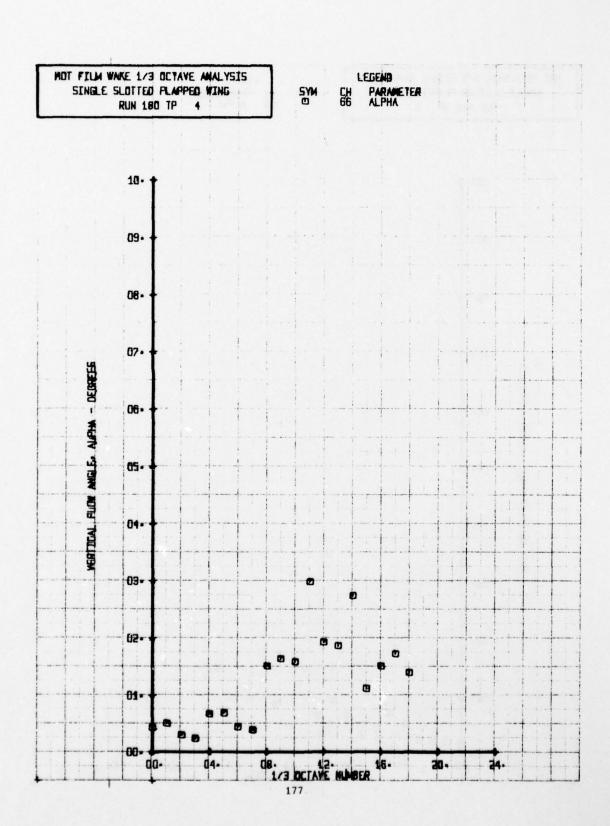


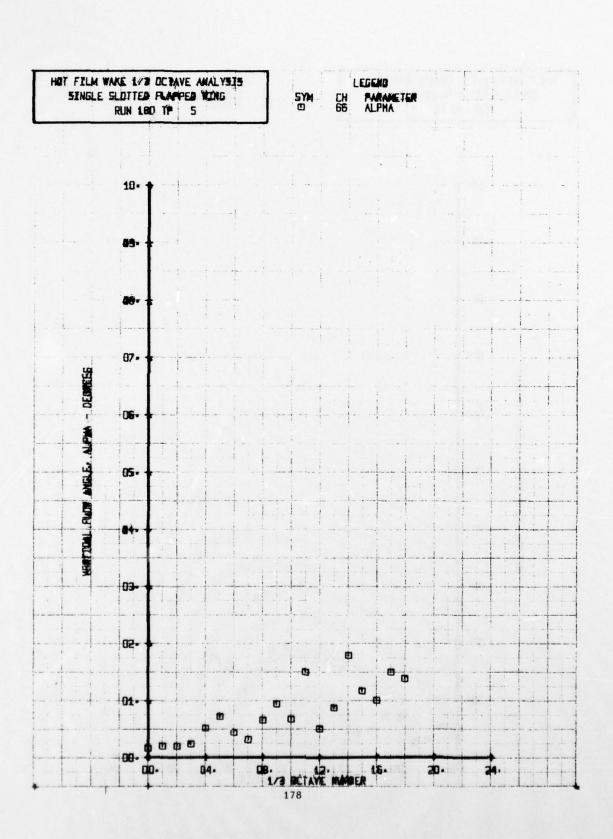


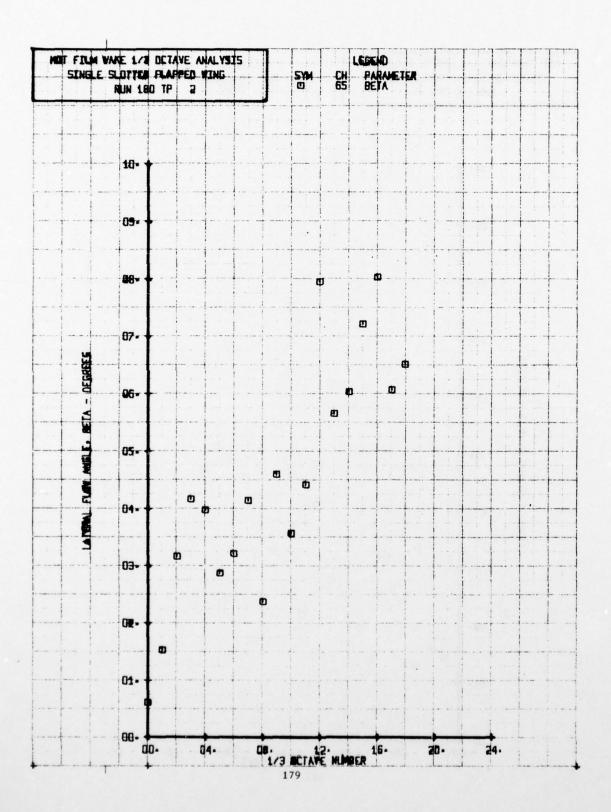


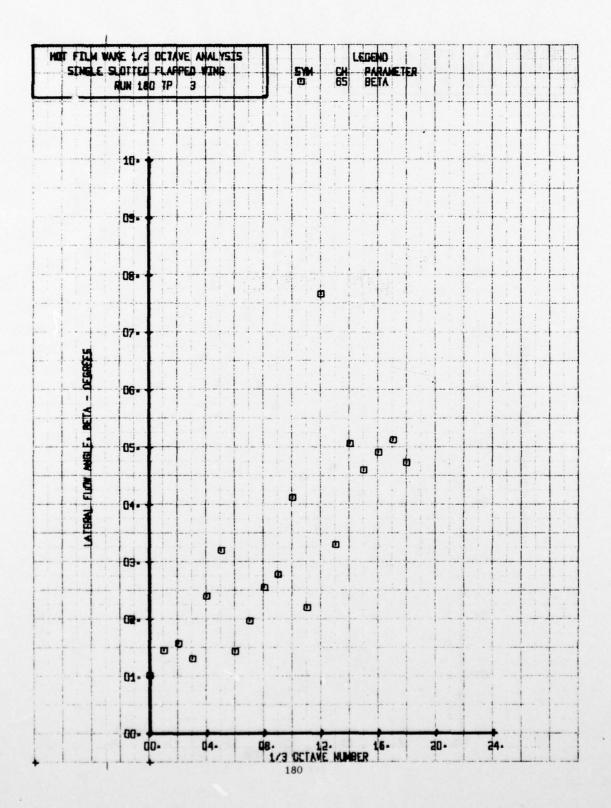


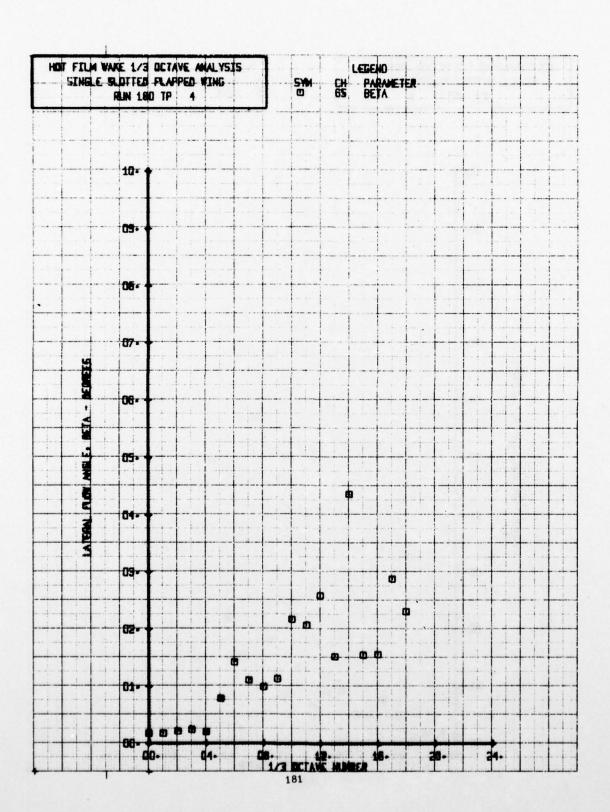


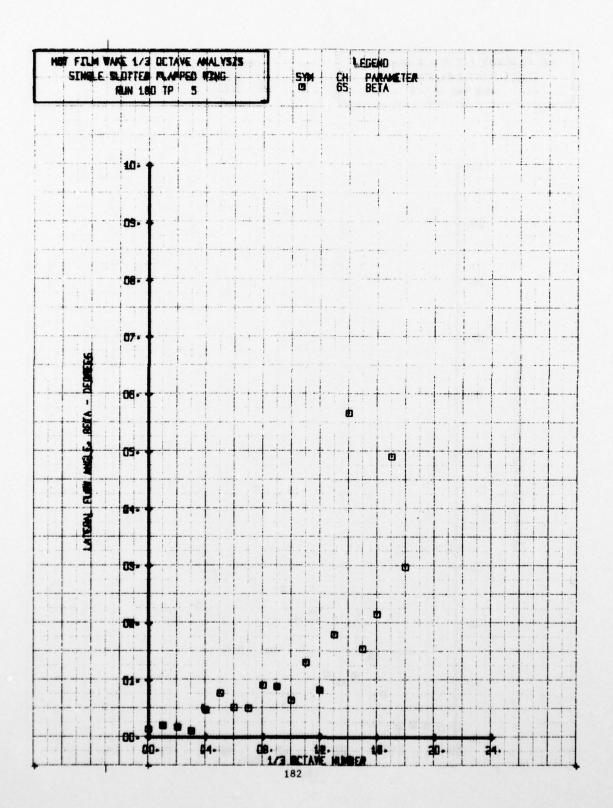


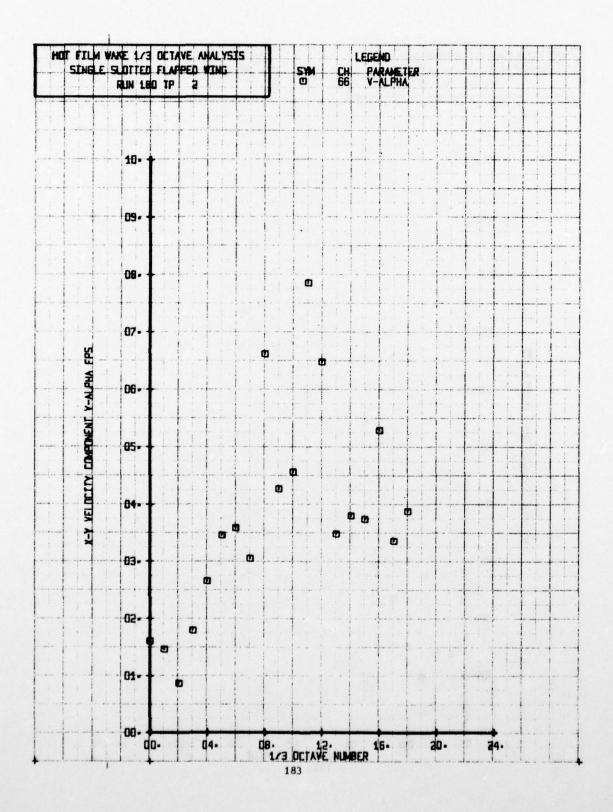


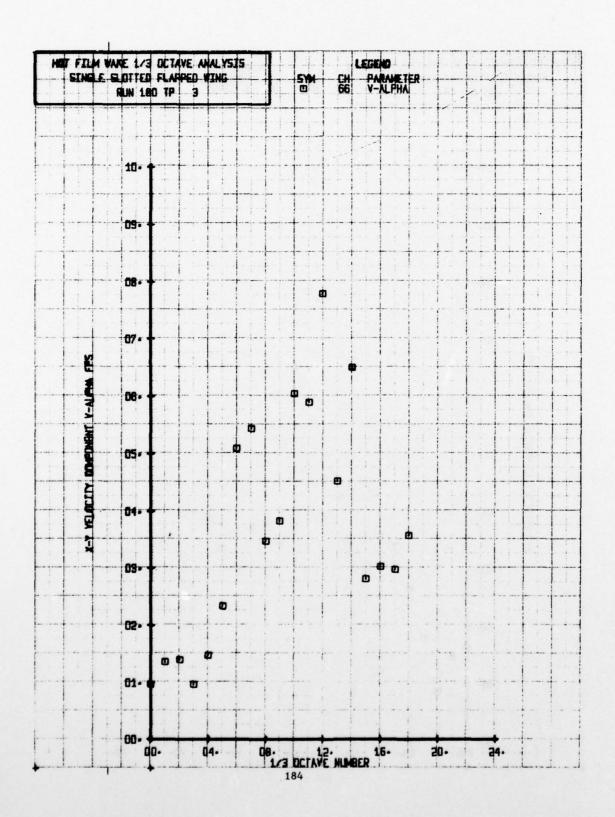


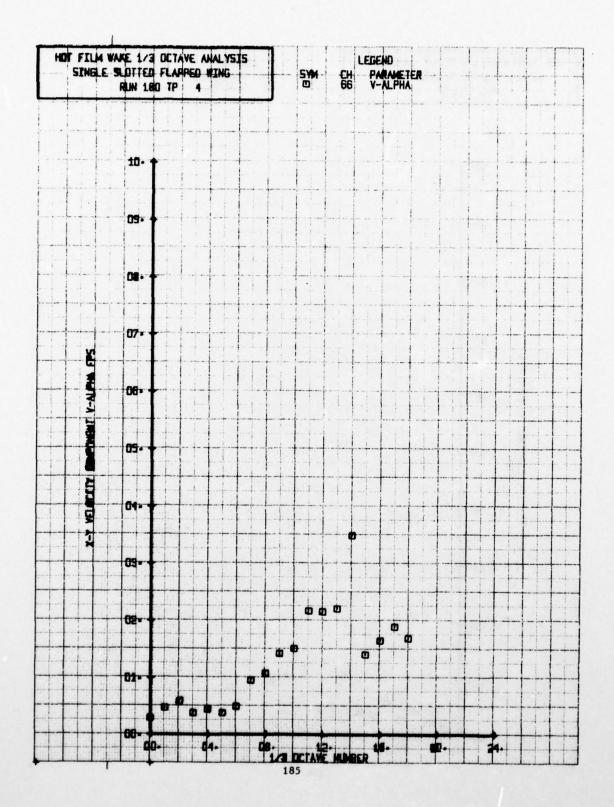


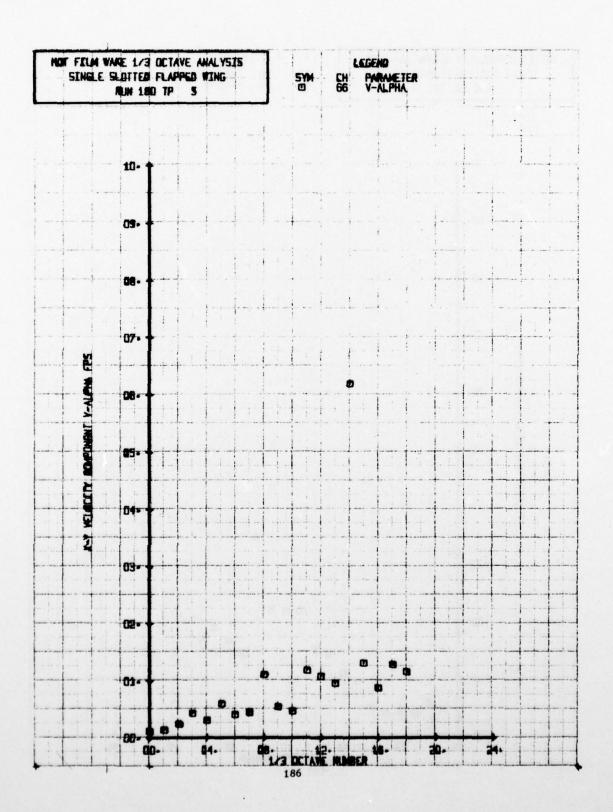


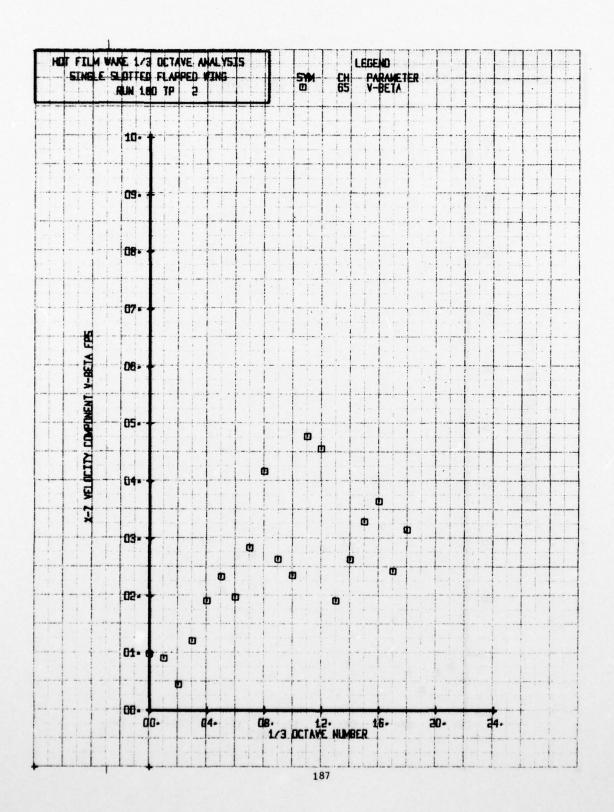


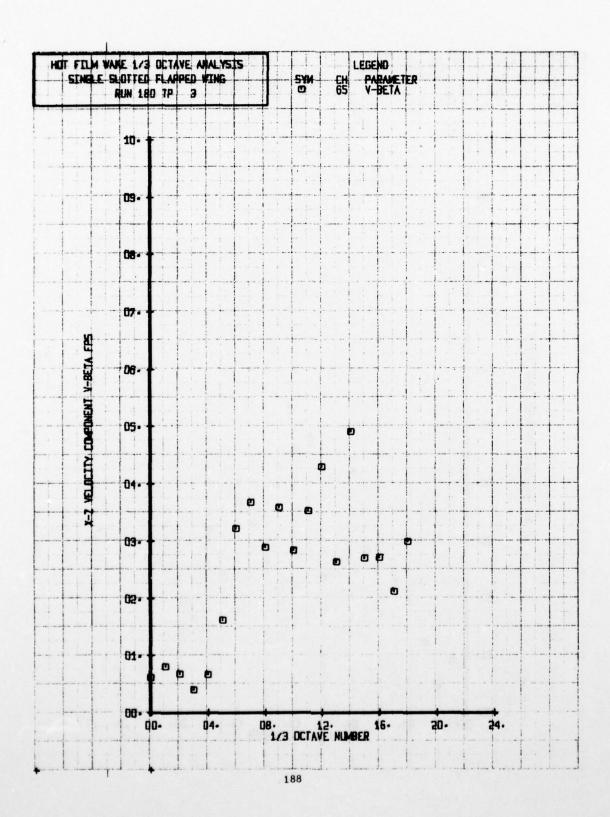


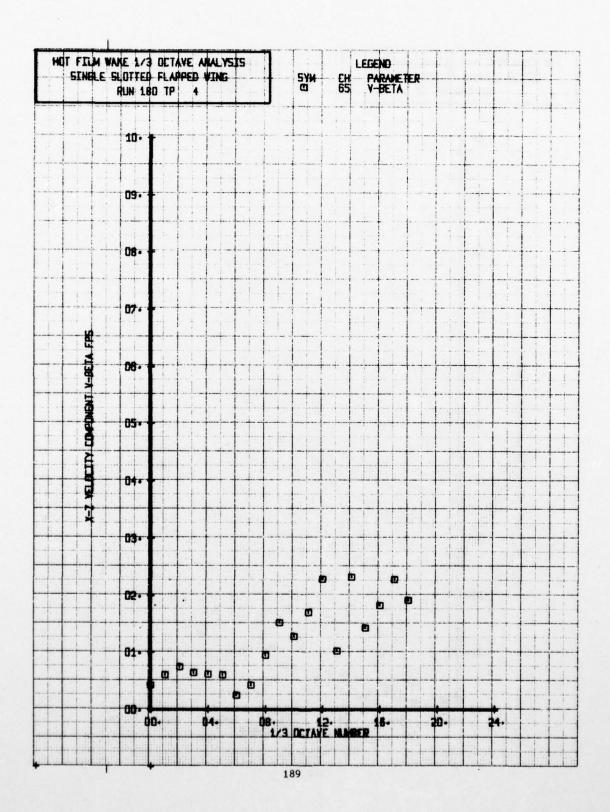


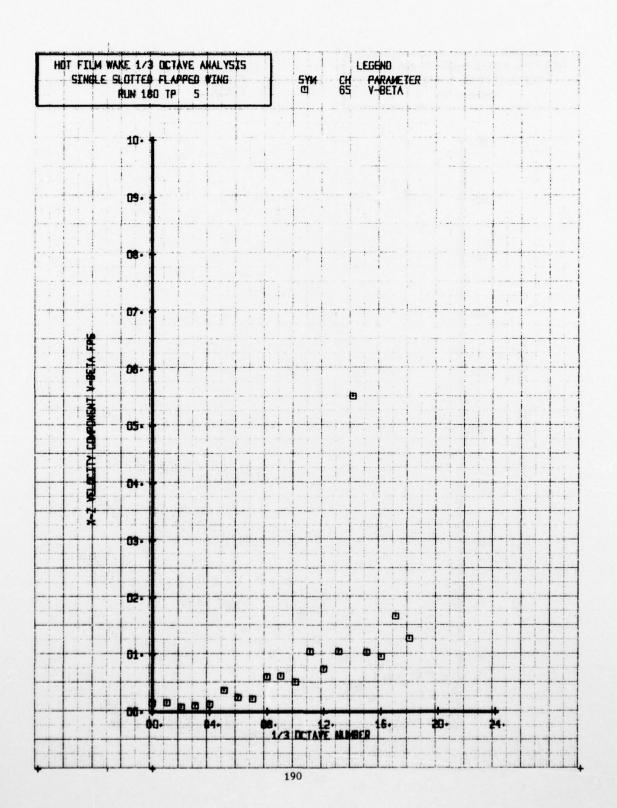


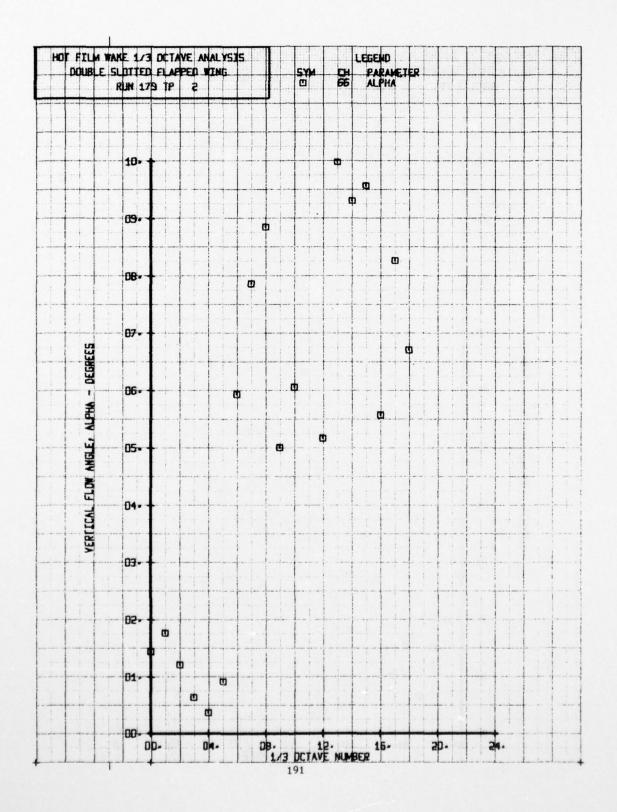


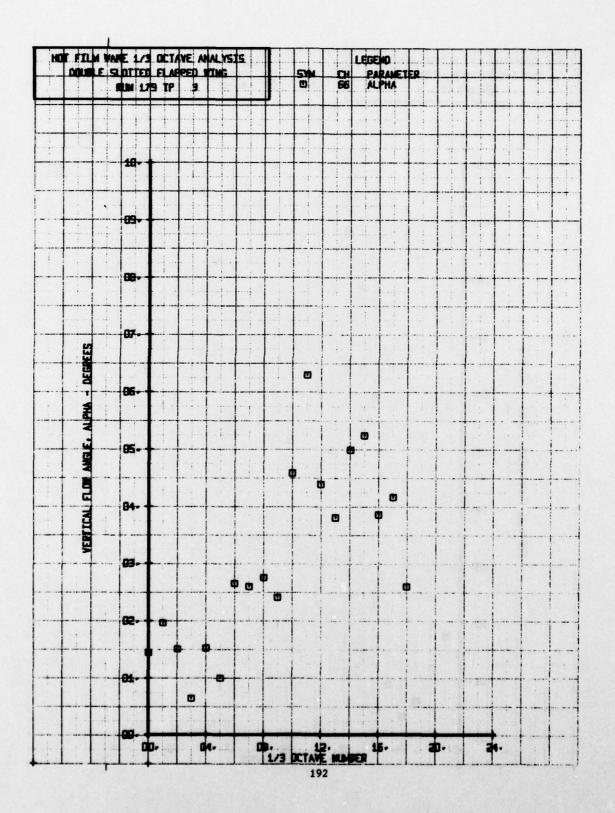






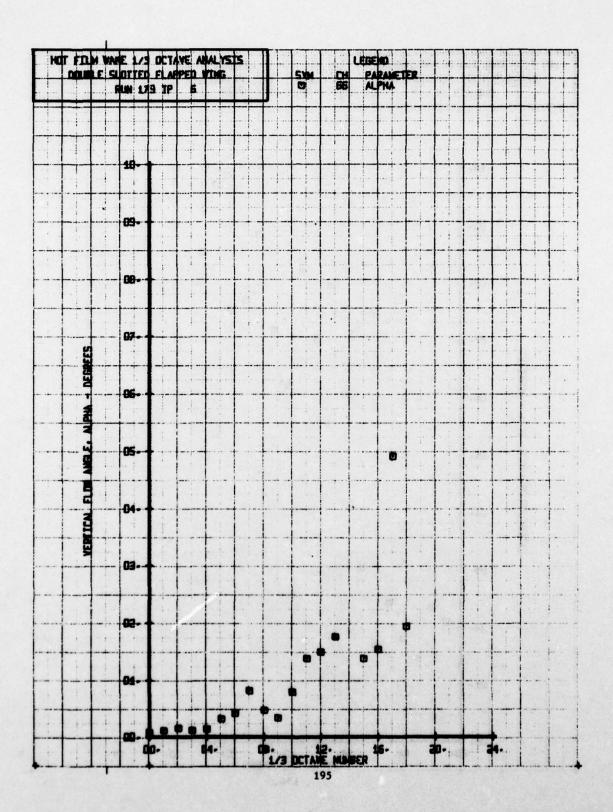


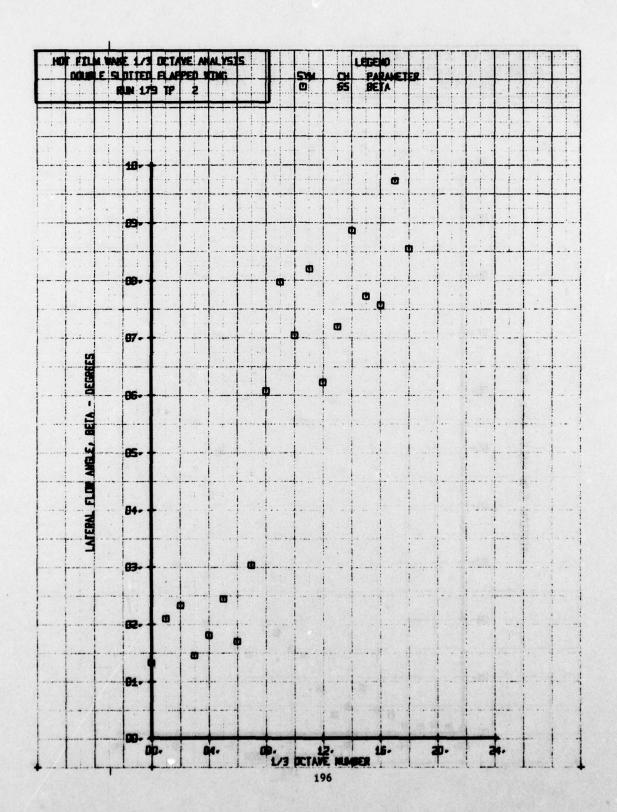




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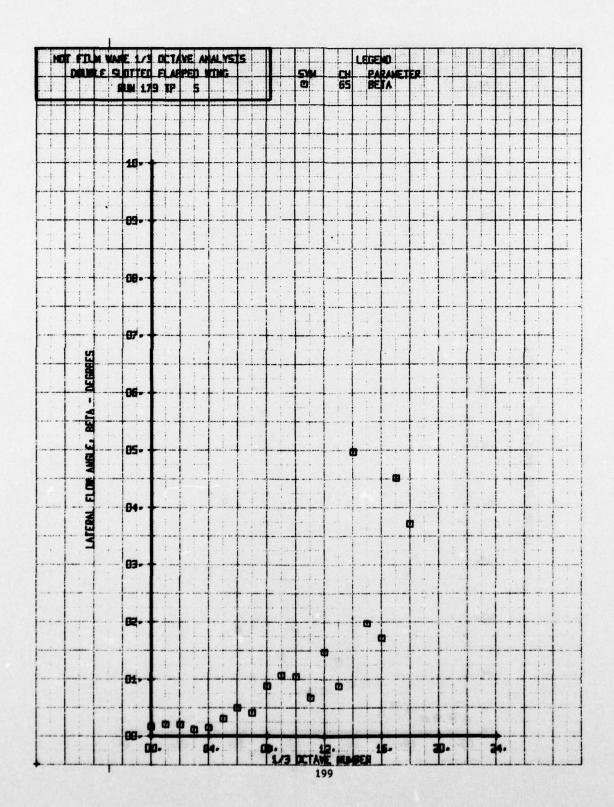
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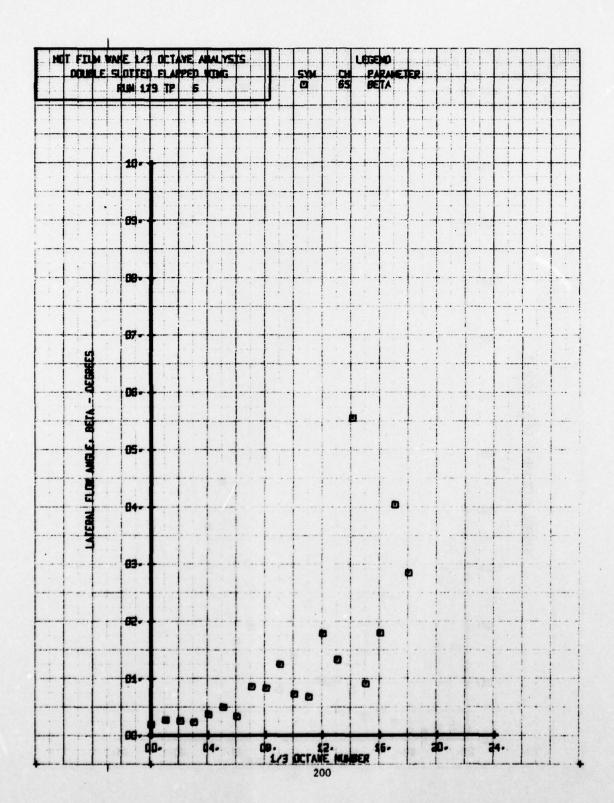




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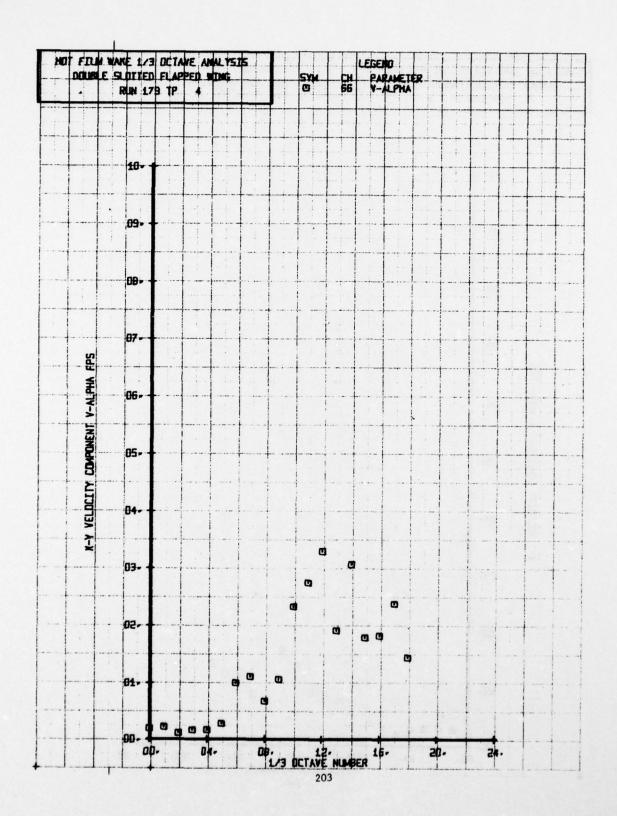
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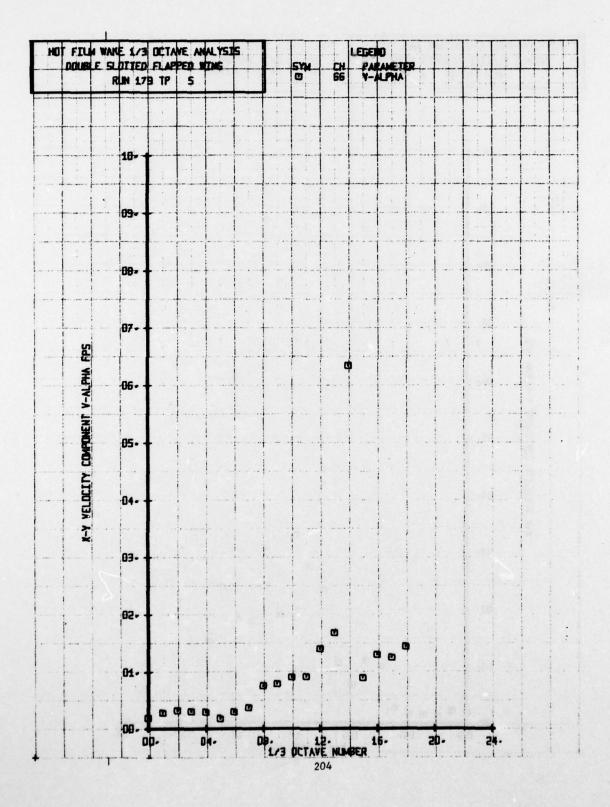




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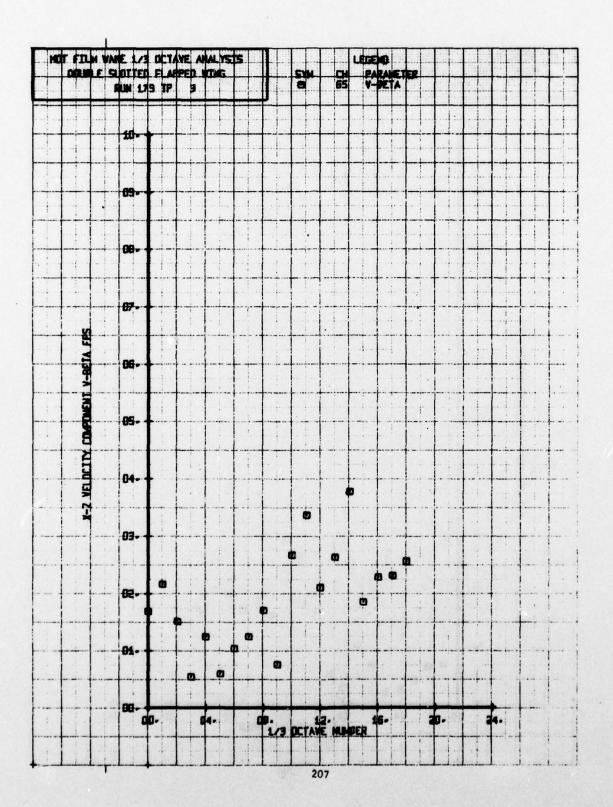
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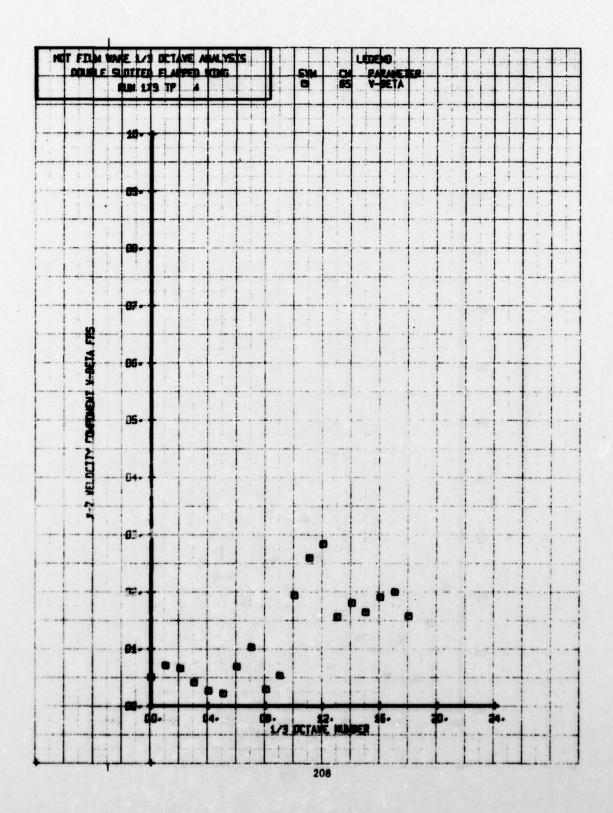




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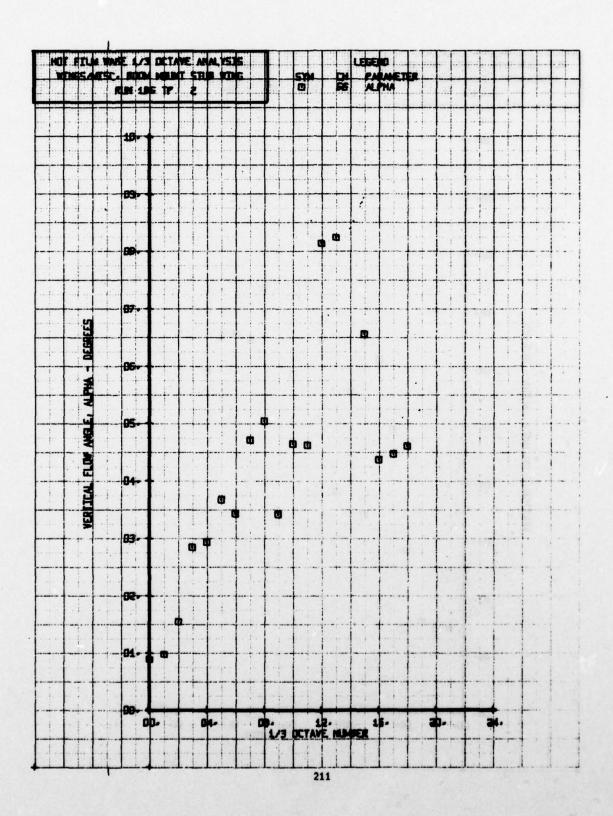
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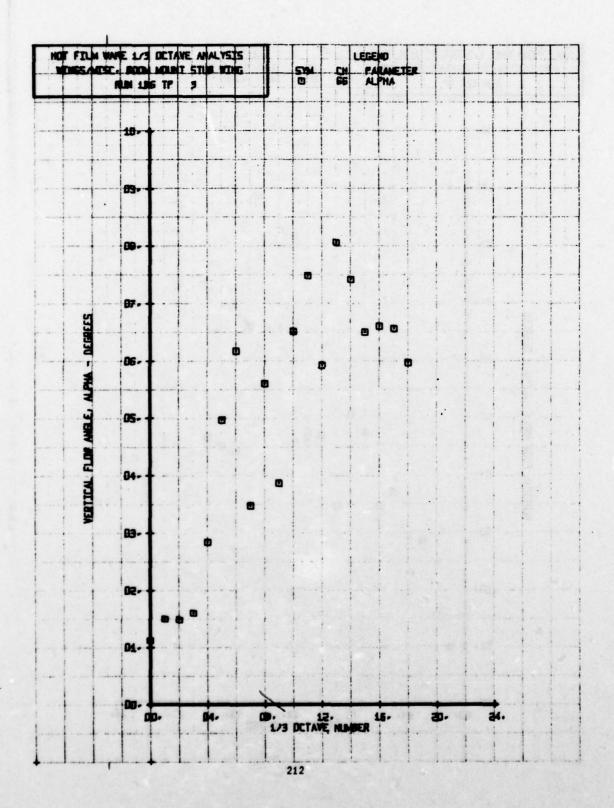


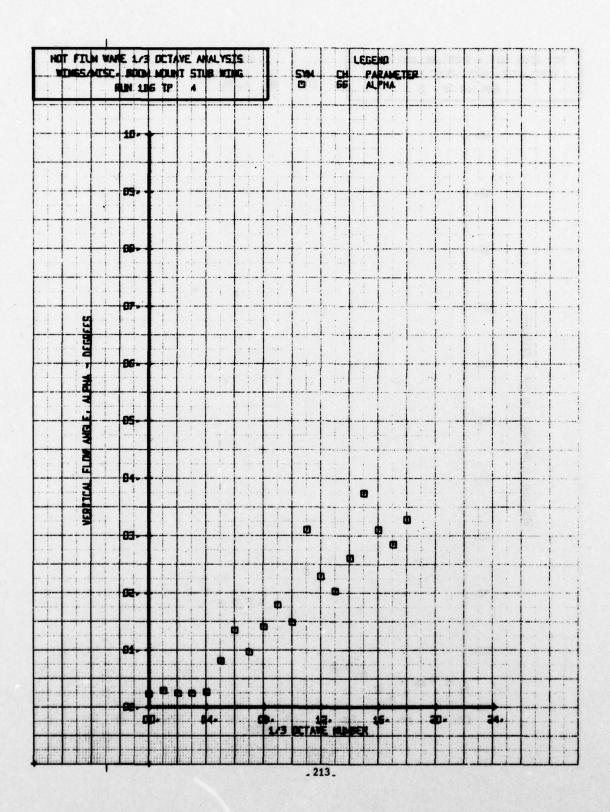


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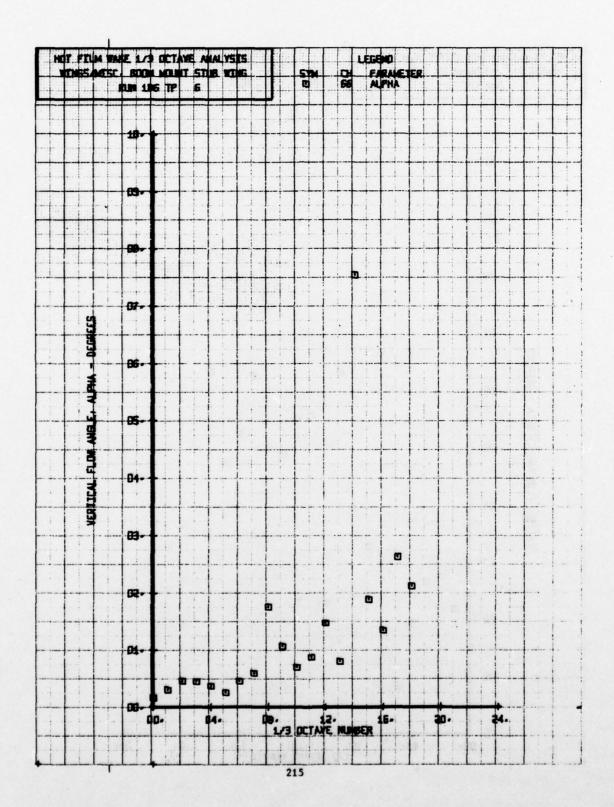
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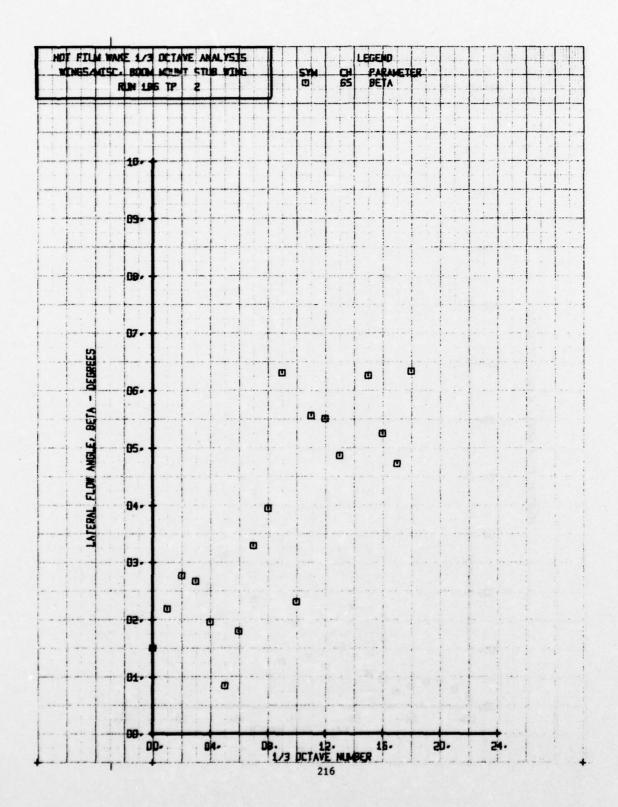


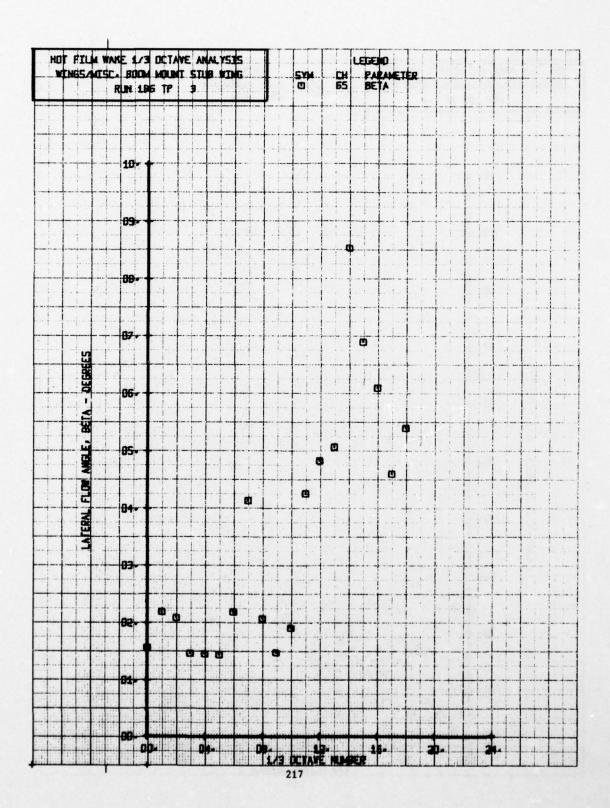


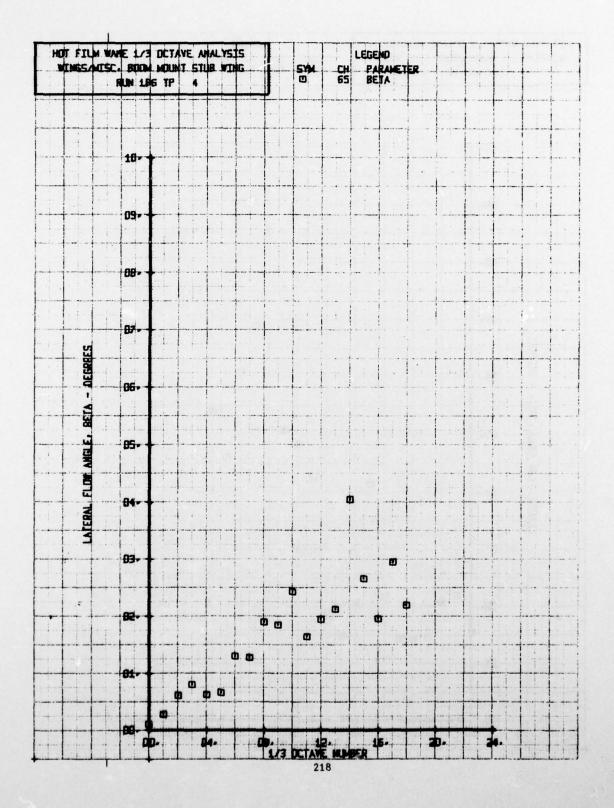


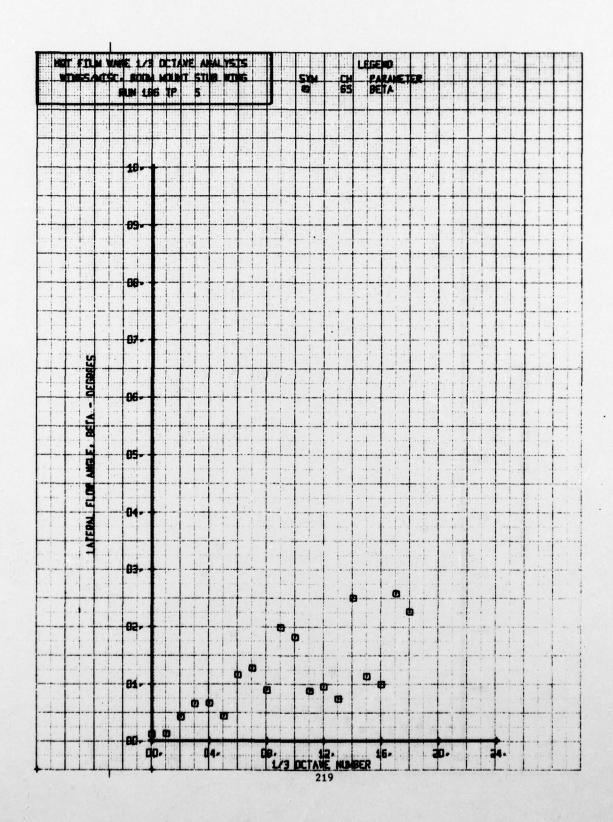
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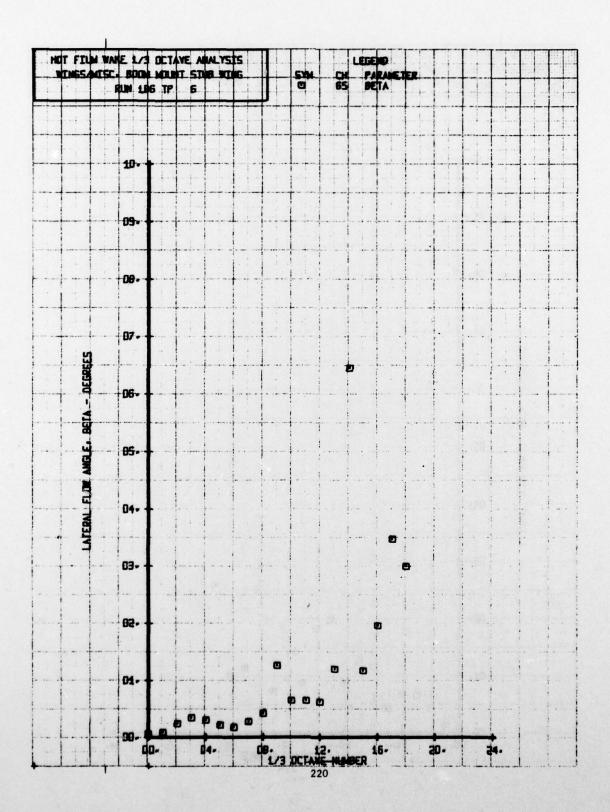


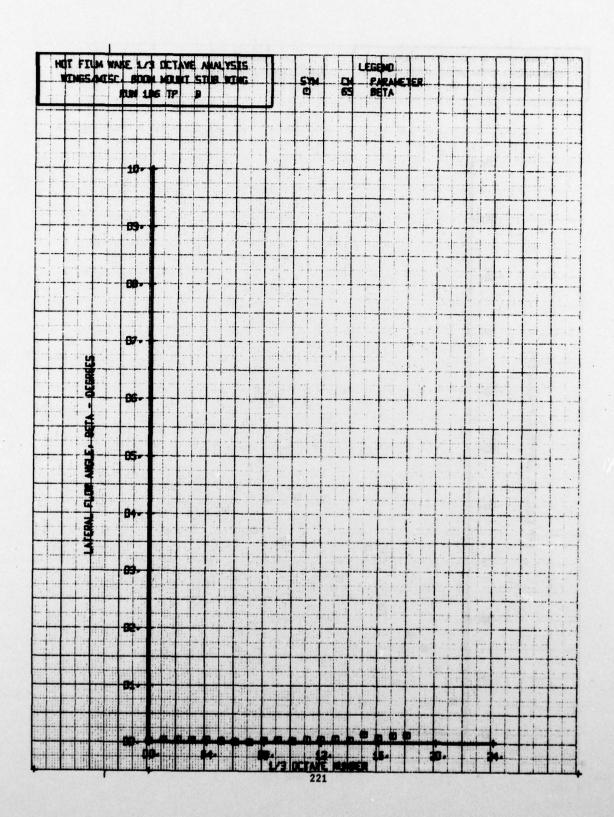




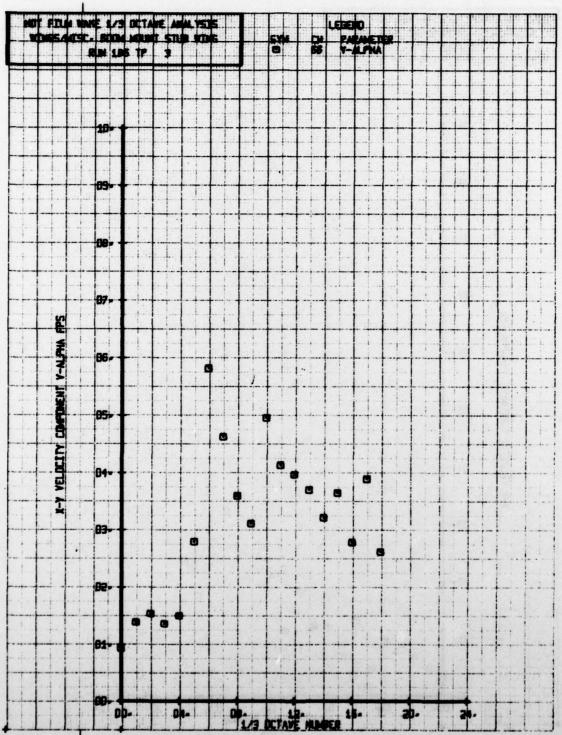


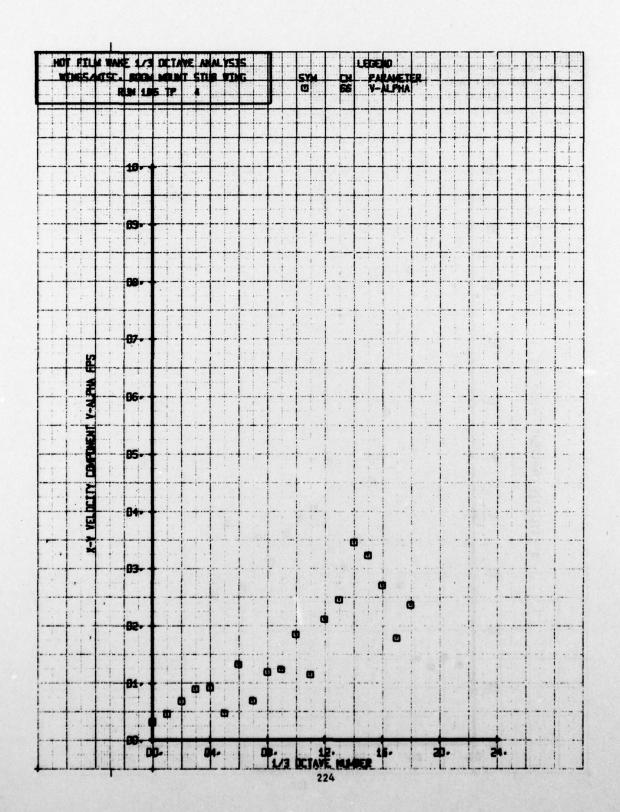


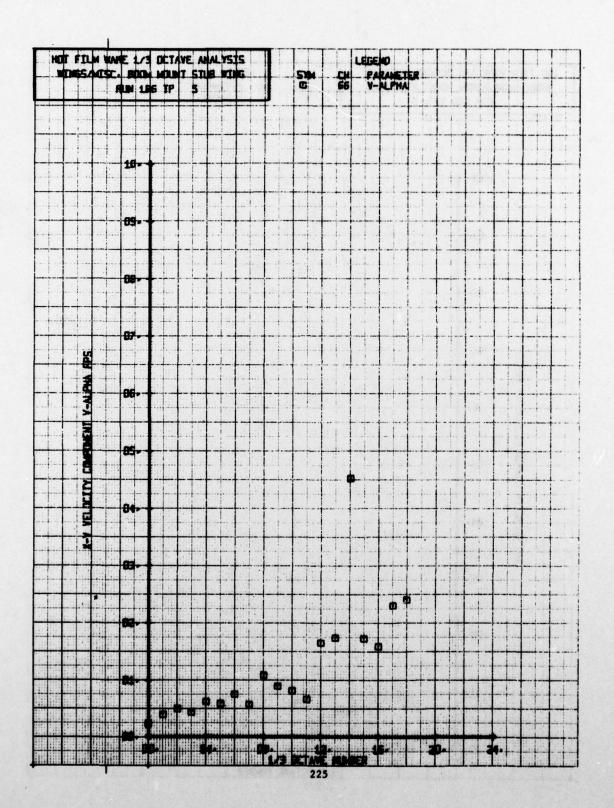


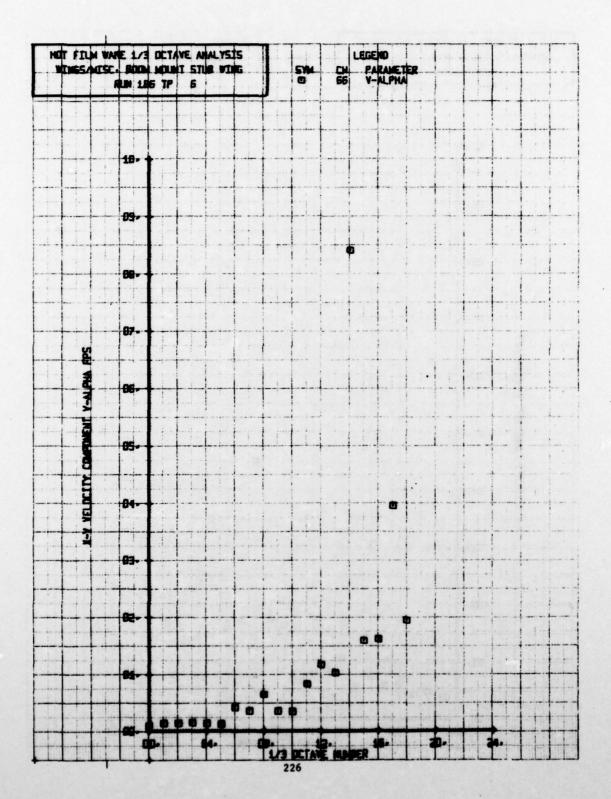


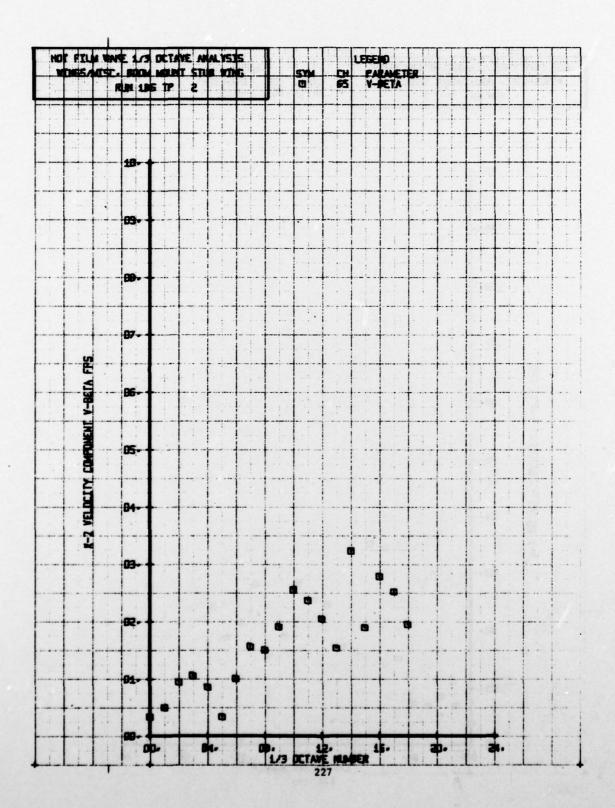
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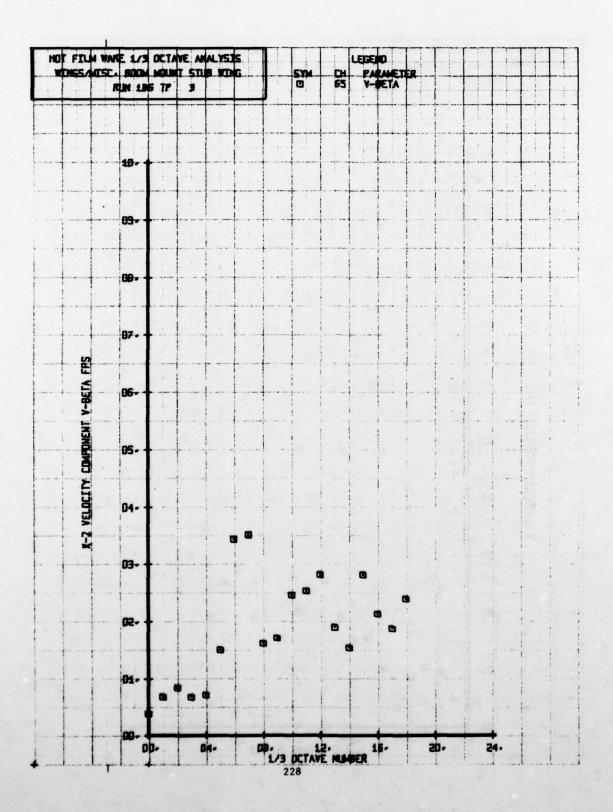












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